



# Contribution to HiLiftPW-3

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Steve Evans, Oisin Tong  
Siemens PLM Software Inc. (003)

# Summary of cases completed:

## STAR-CCM+, committee grids, k- $\omega$ SST ( $a1 = 1$ )

Case	$\alpha = 8$ Fully turbulent grid study	$\alpha = 16$ Fully turbulent grid study
1a (full gap)	YES	YES
1b (full gap w adaptation)	NO	NO
1c (partial seal)	YES	YES
1d (partial seal w adaptation)	NO	NO

Case	Polar Fully turbulent	Polar Specified transition	Polar W transition prediction
2a (no nacelle)	YES	NO	YES
2b (no nacelle w adaptation)	NO	NO	NO
2c (with nacelle)	YES	NO	NO
2d (with nacelle w adaptation)	NO	NO	NO

Case	2D verification study
3 (aerofoil wake)	YES

# Summary of cases completed:

## STAR-CCM+, committee grids, SA

Case	$\alpha = 8$ Fully turbulent grid study	$\alpha = 16$ Fully turbulent grid study
1a (full gap)	YES	YES
1b (full gap w adaptation)	NO	NO
1c (partial seal)	NO	NO
1d (partial seal w adaptation)	NO	NO

Case	Polar Fully turbulent	Polar Specified transition	Polar W transition prediction
2a (no nacelle)	NO	NO	NO
2b (no nacelle w adaptation)	NO	NO	NO
2c (with nacelle)	NO	NO	NO
2d (with nacelle w adaptation)	NO	NO	NO

Case	2D verification study
3 (aerofoil wake)	YES

# Summary of cases completed:

## STAR-CCM+, committee grids, k-ε Lag Elliptic Blending

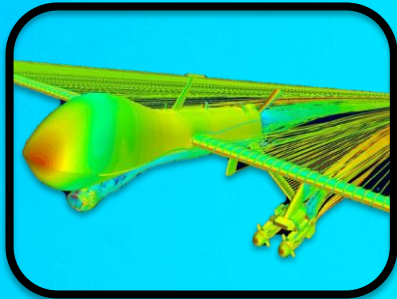
Case	$\alpha = 8$ Fully turbulent grid study	$\alpha = 16$ Fully turbulent grid study
1a (full gap)	YES	YES
1b (full gap w adaptation)	NO	NO
1c (partial seal)	NO	NO
1d (partial seal w adaptation)	NO	NO

Case	Polar Fully turbulent	Polar Specified transition	Polar W transition prediction
2a (no nacelle)	NO	NO	YES
2b (no nacelle w adaptation)	NO	NO	NO
2c (with nacelle)	NO	NO	NO
2d (with nacelle w adaptation)	NO	NO	NO

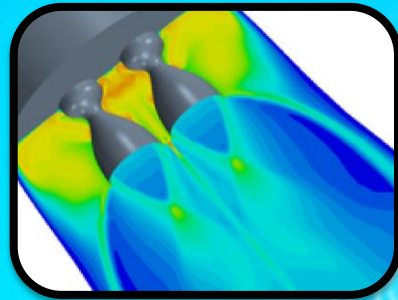
Case	2D verification study
3 (aerofoil wake)	YES

# Aerospace and Defense applications of STAR-CCM+

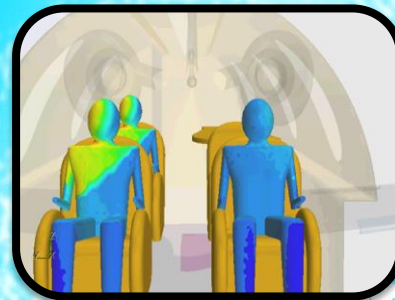
**SIEMENS**  
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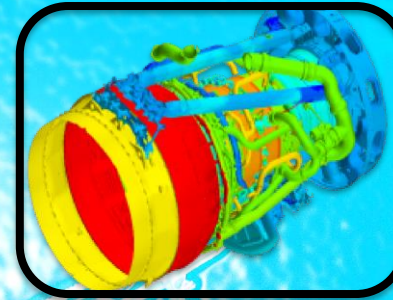
External  
Aerodynamics



Rocket  
Propulsion



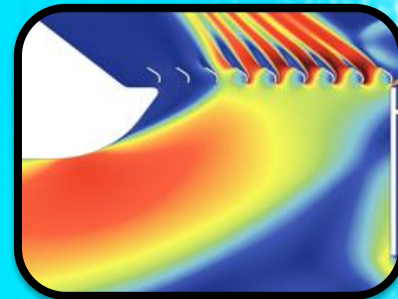
Environmental  
Control Systems



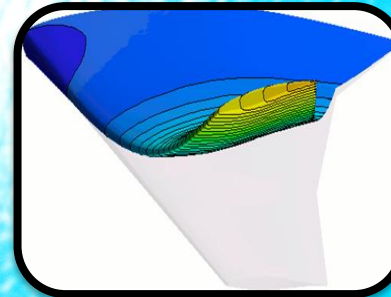
Thermal  
Management



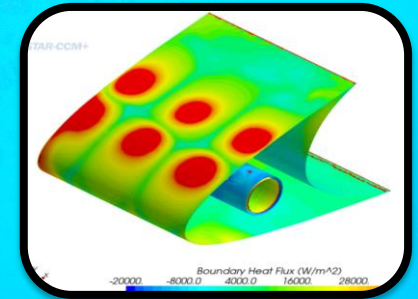
Aeroacoustics



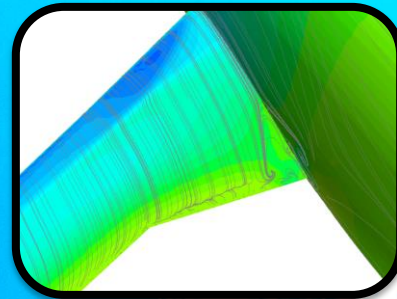
Design Exploration



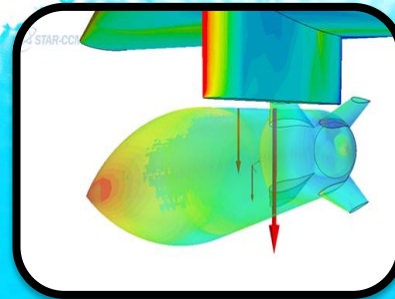
Fluid Structure  
interactions



Ice Accretion and  
Protection



Validation



Store Separation

# STAR-CCM+

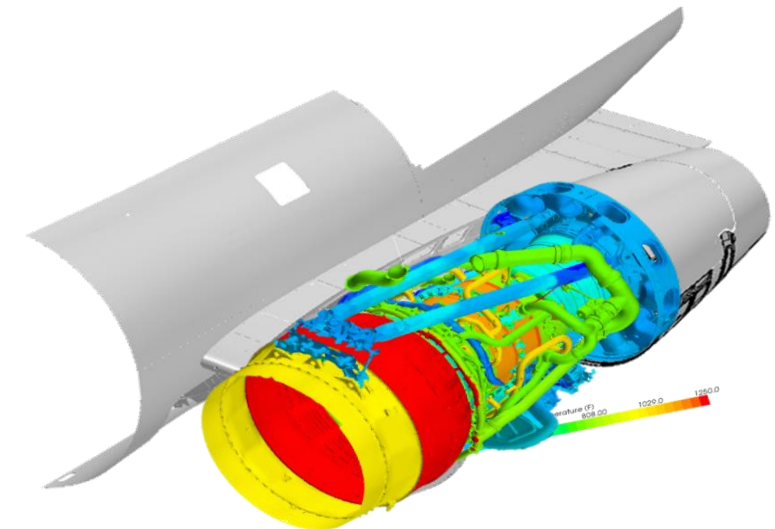
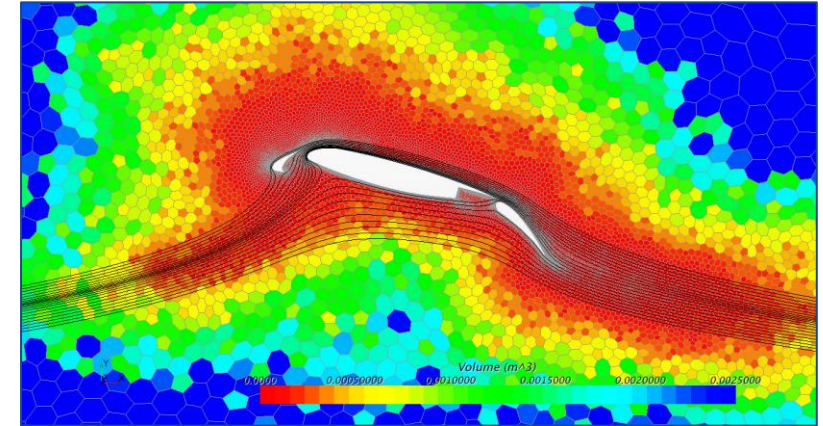
## Summary of the code and numerics used

### Modelling

- Density-based coupled-flow solver, with an automated convergence control tool (expert driver) to accelerate convergence rate.
- Various turbulence models: SA, SST k- $\omega$ , Lag EB k- $\epsilon$
- Additional transition modelling for SST k- $\omega$ :  $\gamma$ -model

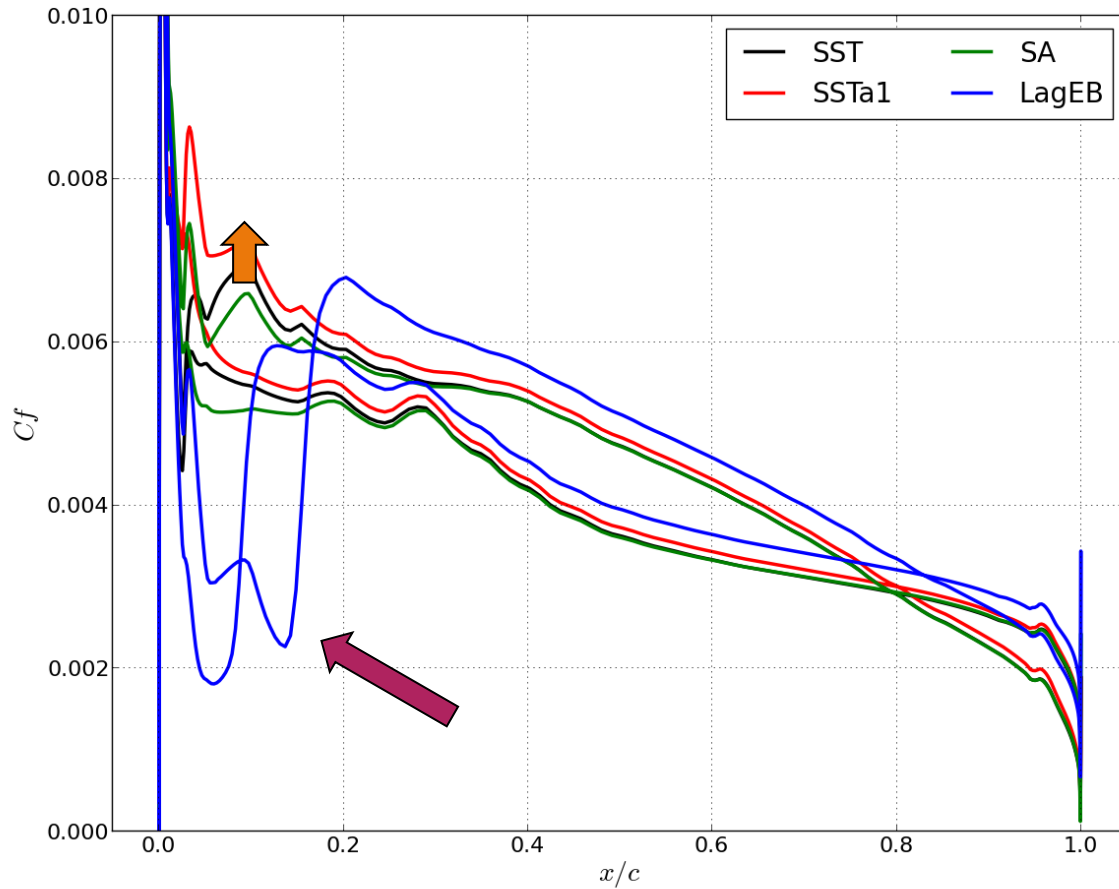
### References

- Evans, S., and Lardeau, S., "Validation of a turbulence methodology using the SST k- $\omega$  model for adjoint calculation," *54th AIAA Aerospace Sciences Meeting*, 2016.
- Lardeau, S., and Billard F., "Development of an elliptic-blending lag model for industrial applications," *54th AIAA Aerospace Sciences Meeting*, 2016.
- Menter, F., et al., "A one-equation local correlation-based transition model," *Flow, Turbulence and Combustion* 95.4 (2015): 583-619.
- Weiss, J.M., and Smith, W.A.,. "Preconditioning applied to variable and constant density flows," *AIAA Journal*, 33(11), pp. 2050-2057.1995

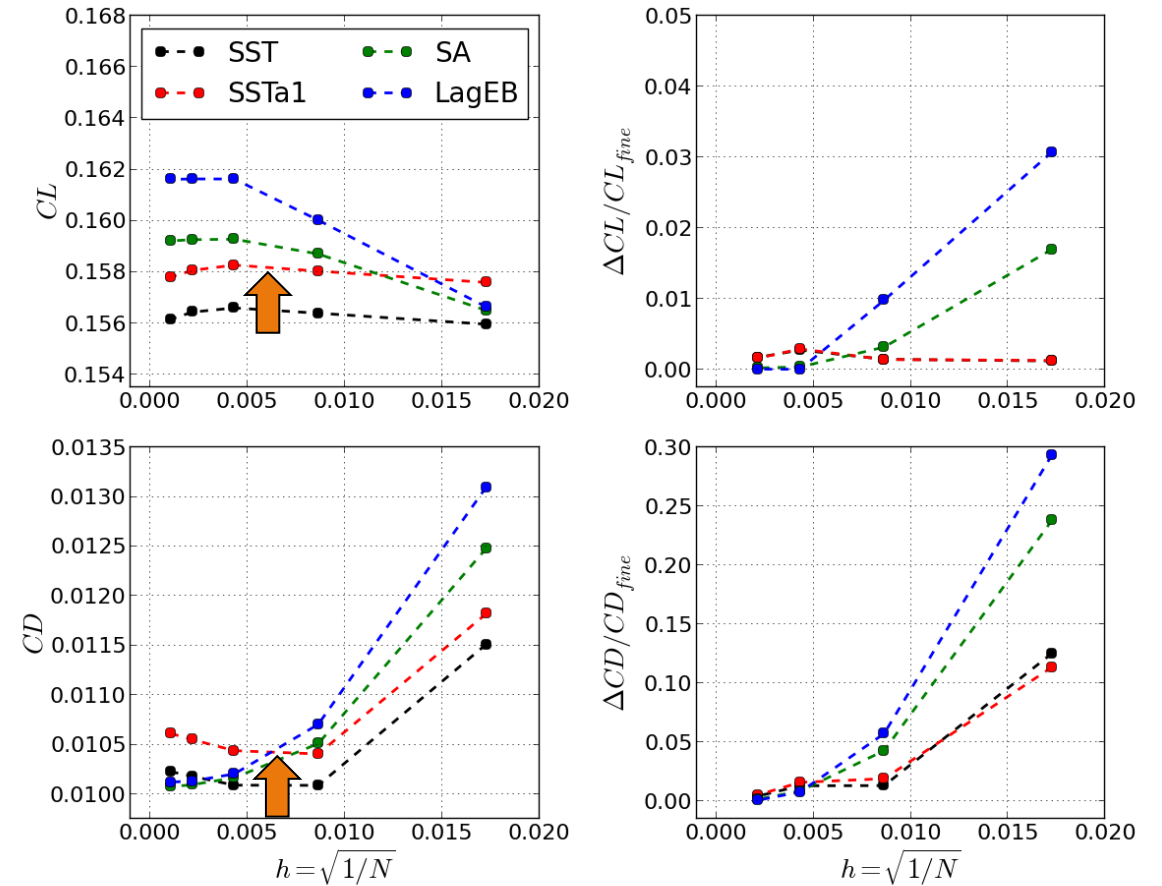


# Verification study results

## 2D Airfoil Near-Wake

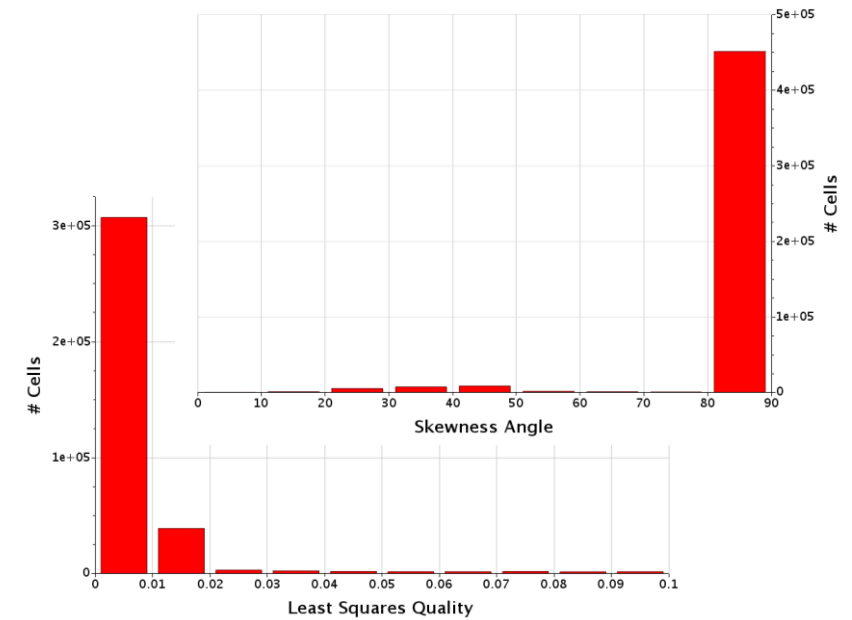
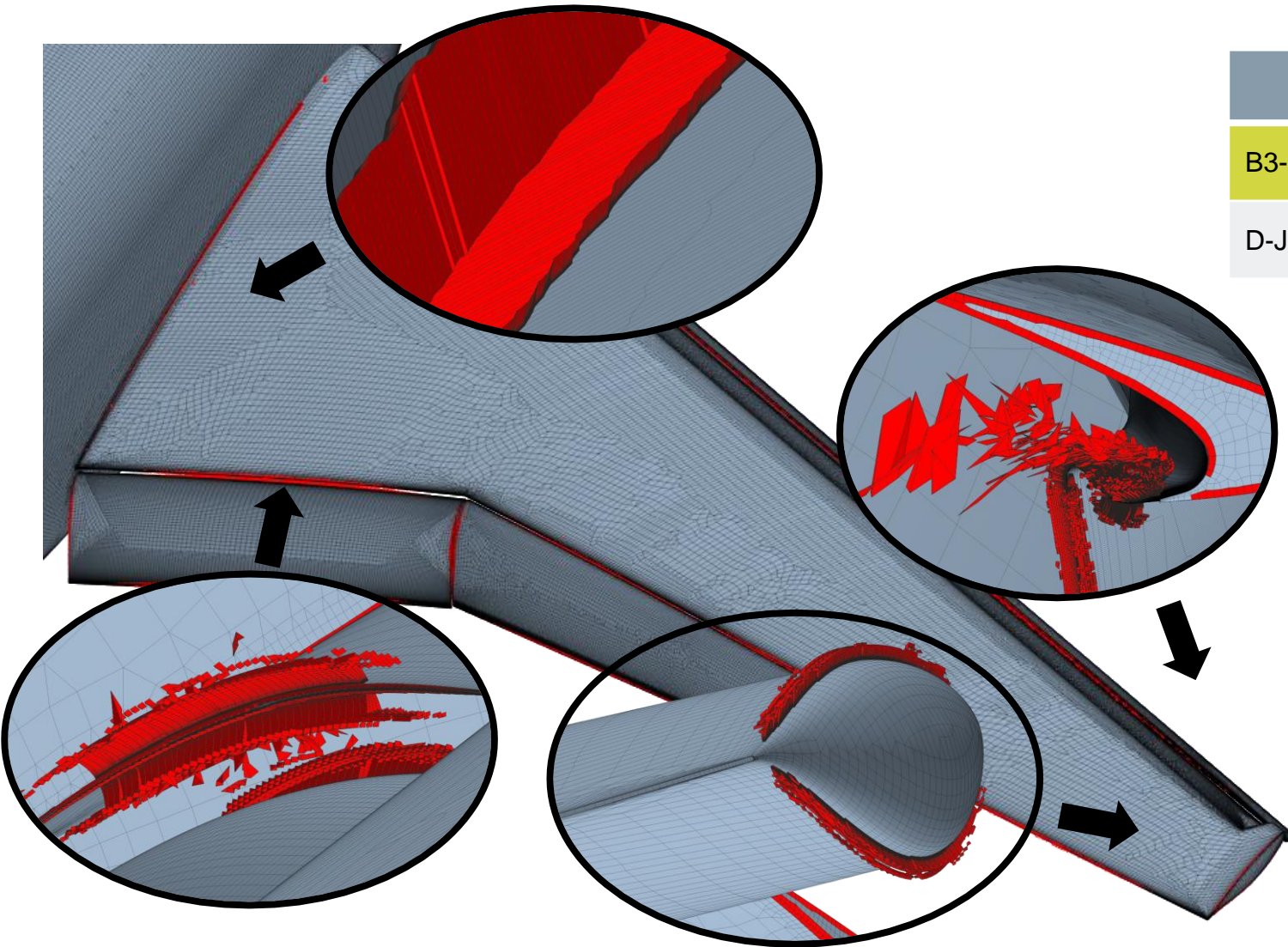


### Case 3 - Grid Convergence



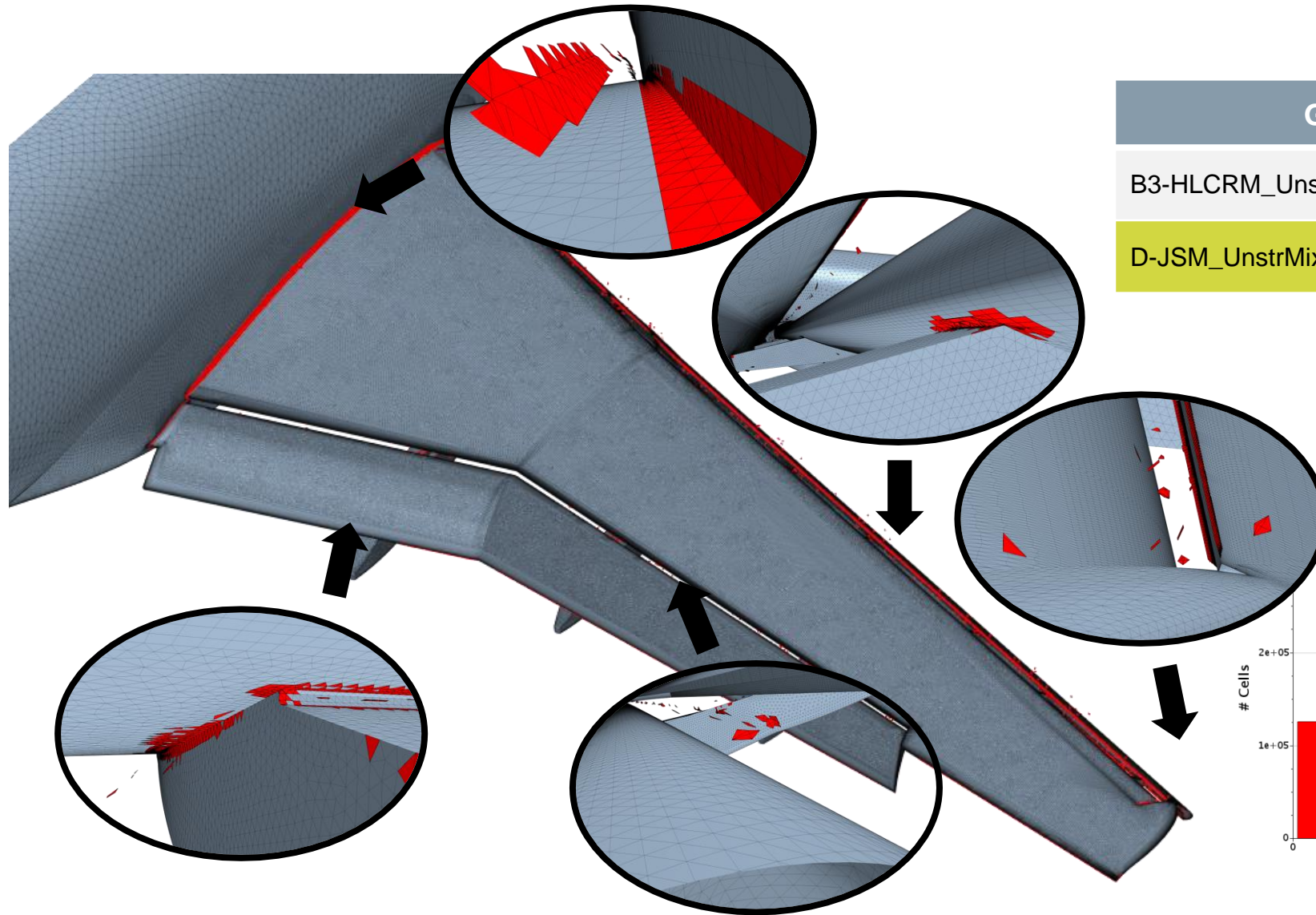
# Overview of grid systems – HL-CRM

Grid System	Cases
B3-HLCRM_UnstrHexPrismPyrTet_PW	1a, 1c
D-JSM_UnstrMixed_JAXA	2a, 2c

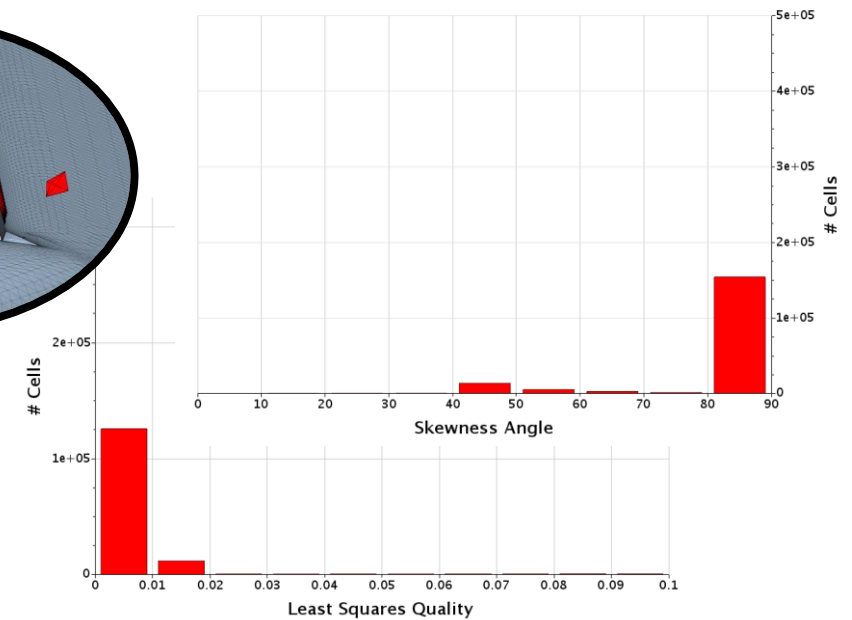




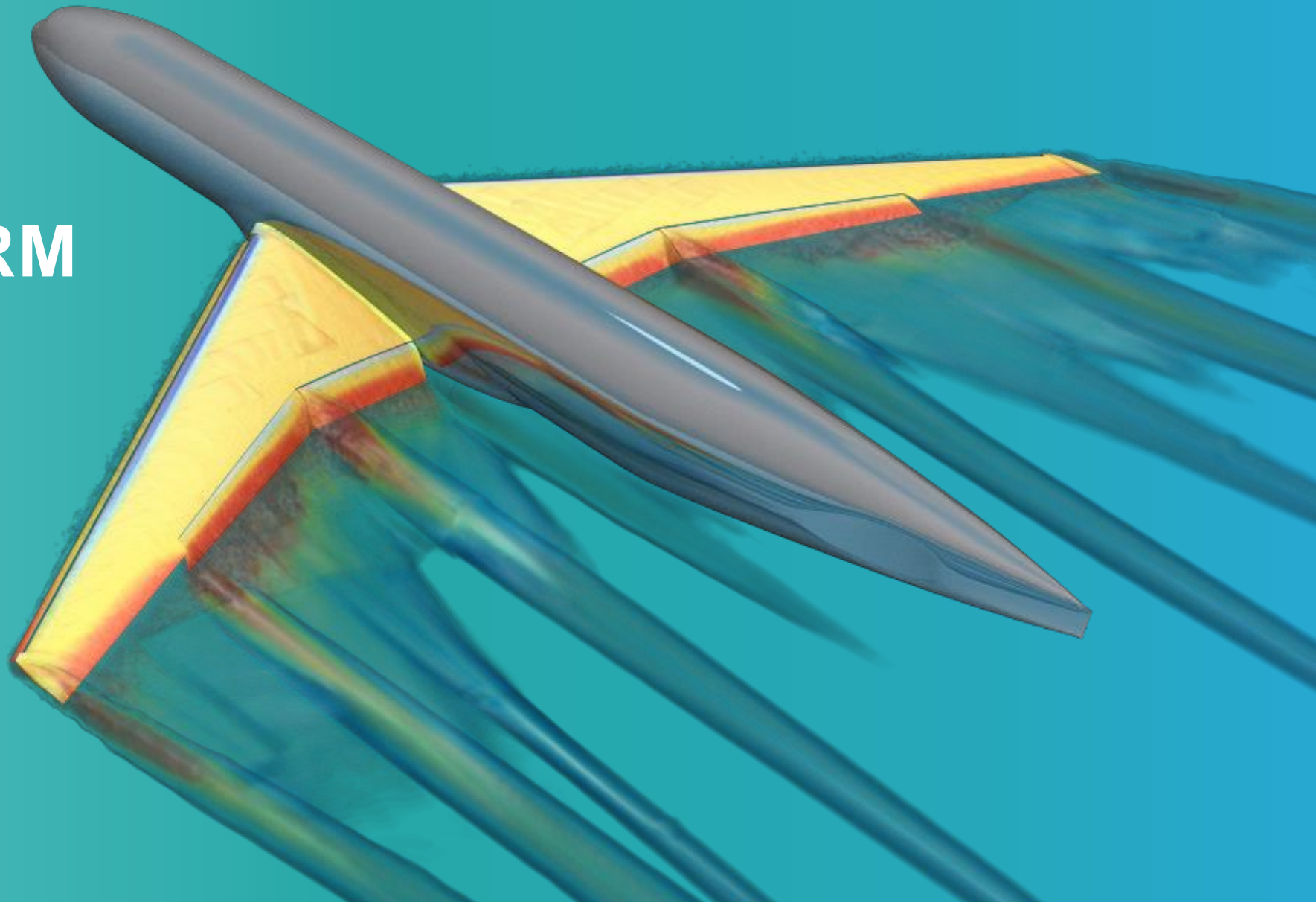
# Overview of grid systems - JSM



Grid System	Cases
B3-HLCRM_UnstrHexPrismPyrTet_PW	1a, 1c
D-JSM_UnstrMixed_JAXA	2a, 2c

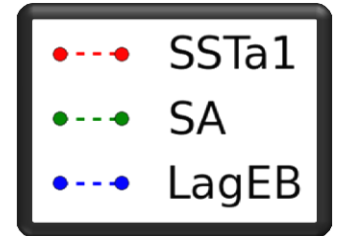
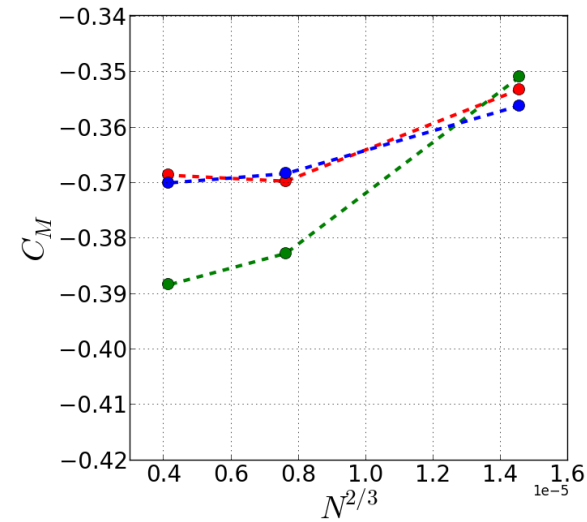
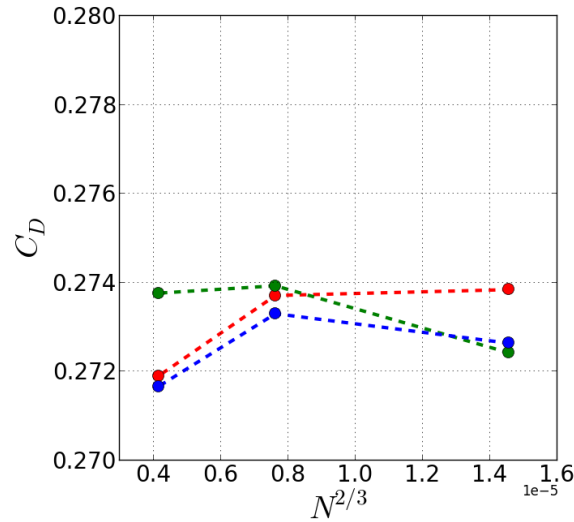
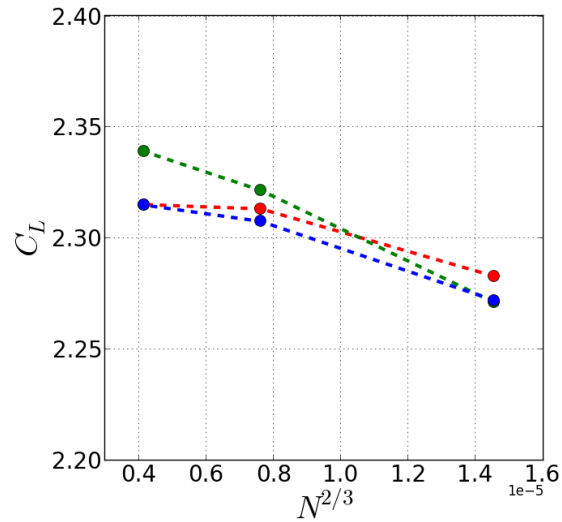


# Case 1: HL - CRM

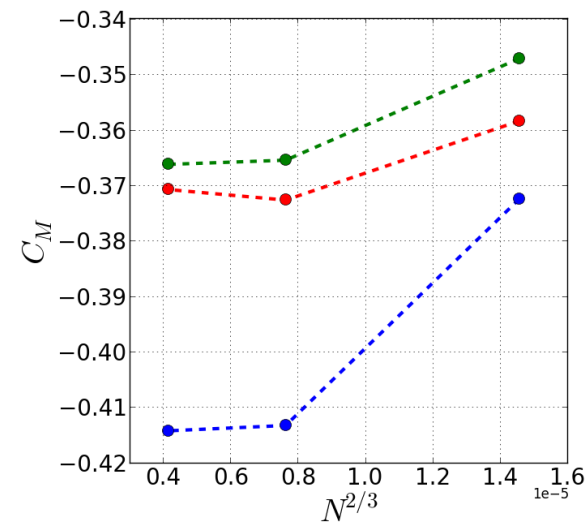
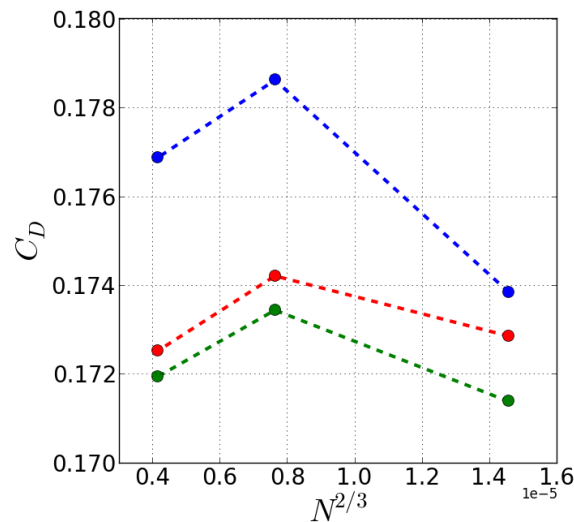
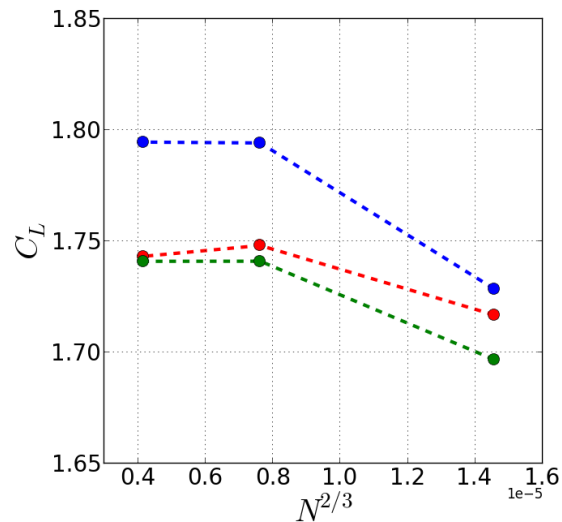


# Overview of HL-CRM results

16°



8°



# HL-CRM – Comparison of turbulence models

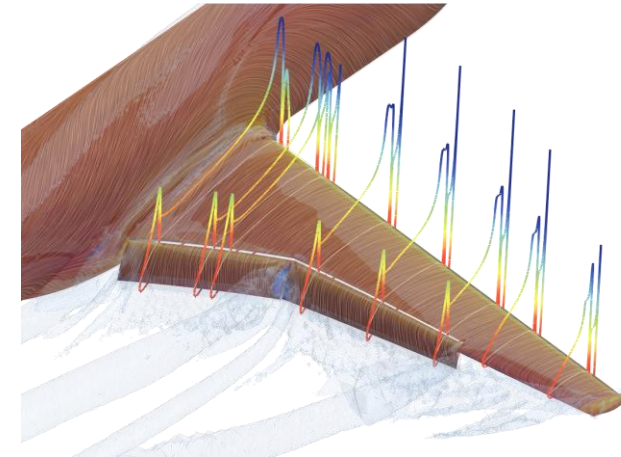
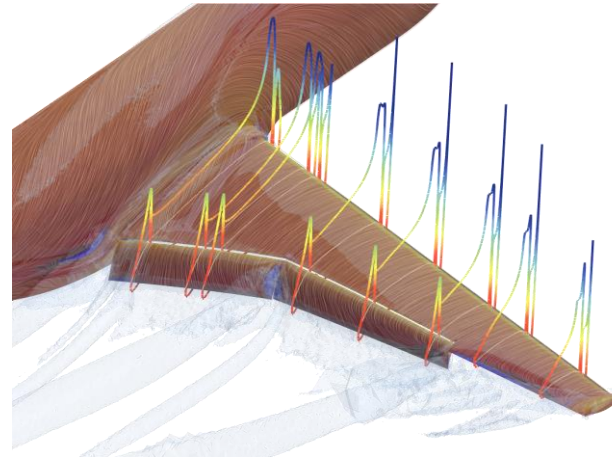
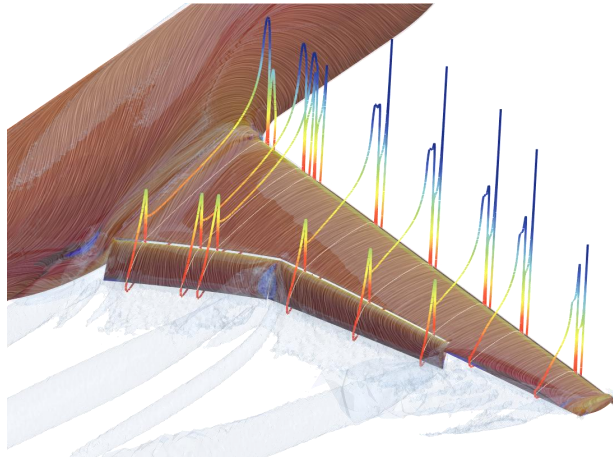
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**SSTa1**

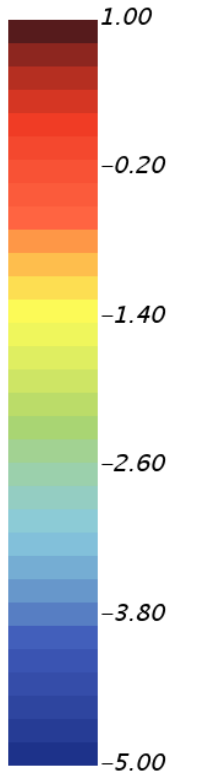
**SA**

**LagEB**

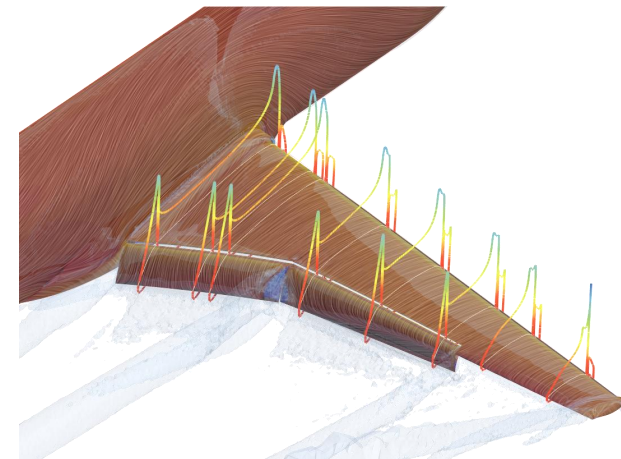
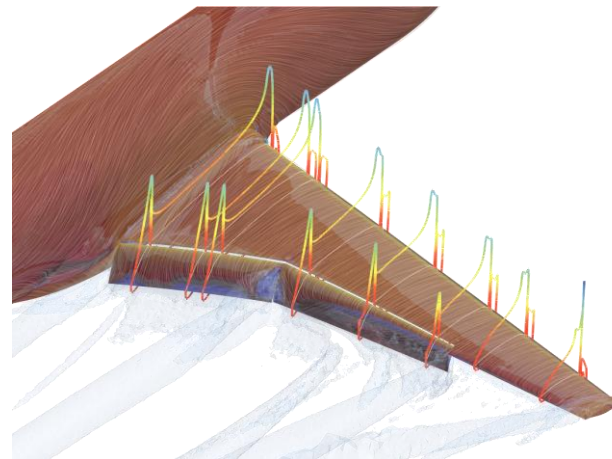
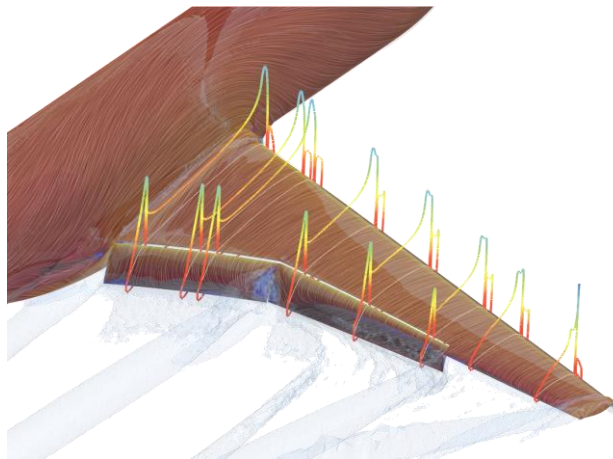
**16°**



*Pressure  
Coefficient*



**8°**



*Skin Friction Coefficient*



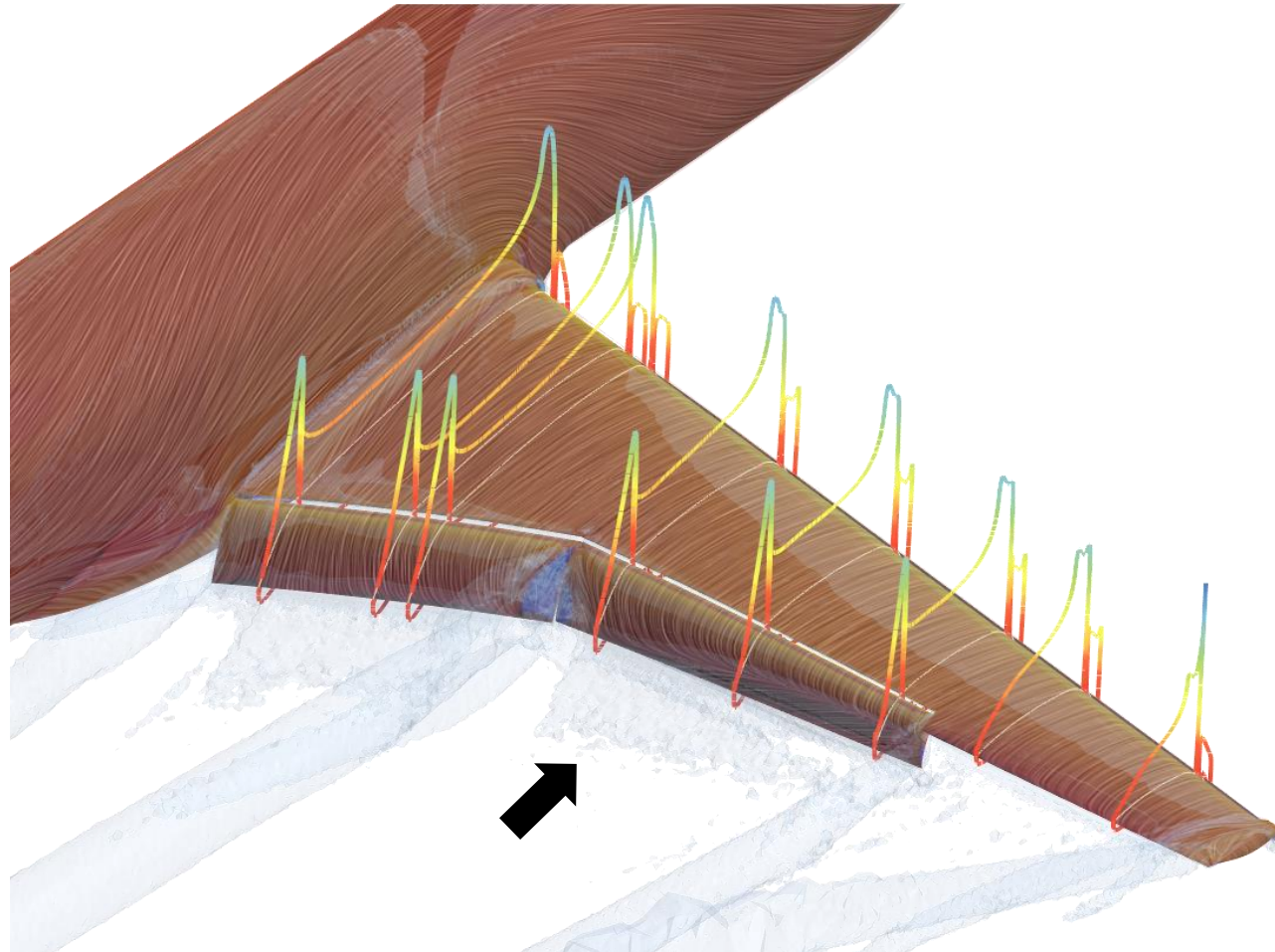
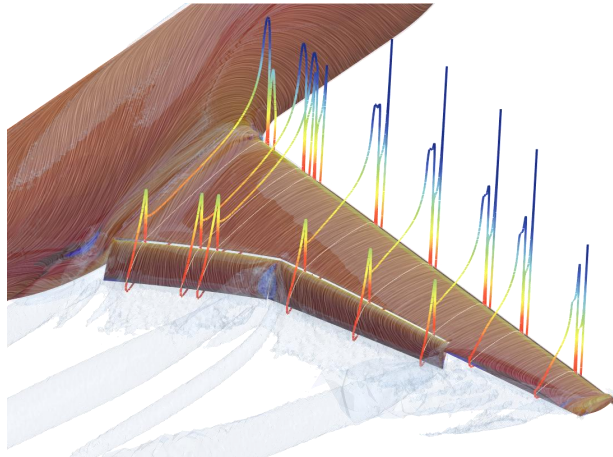
# HL-CRM – Comparison of turbulence models

SSTa1

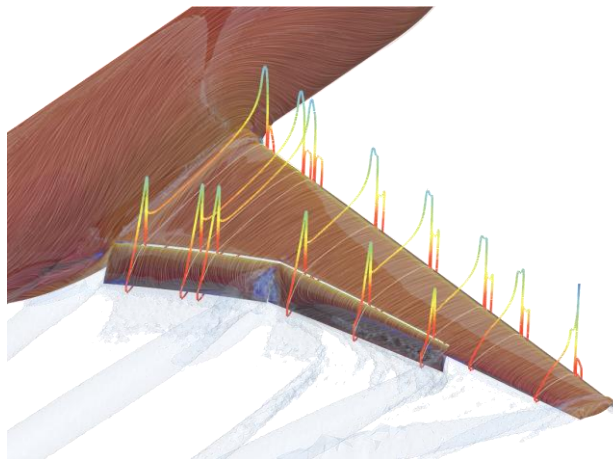
SA

LagEB

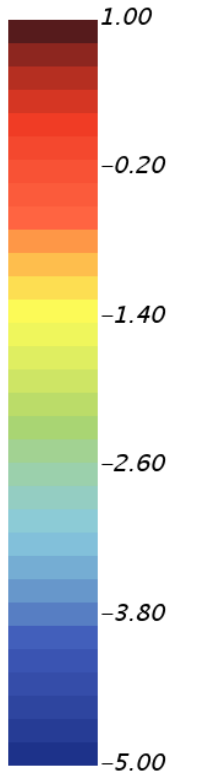
16°



8°



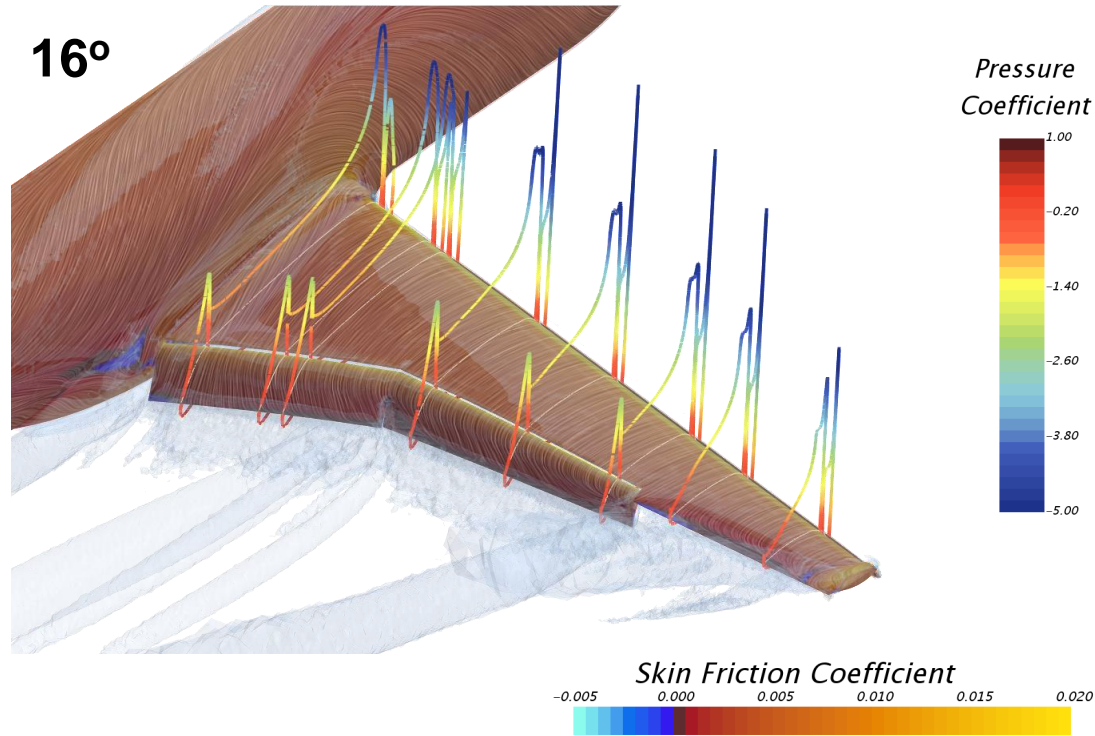
Pressure  
Efficient



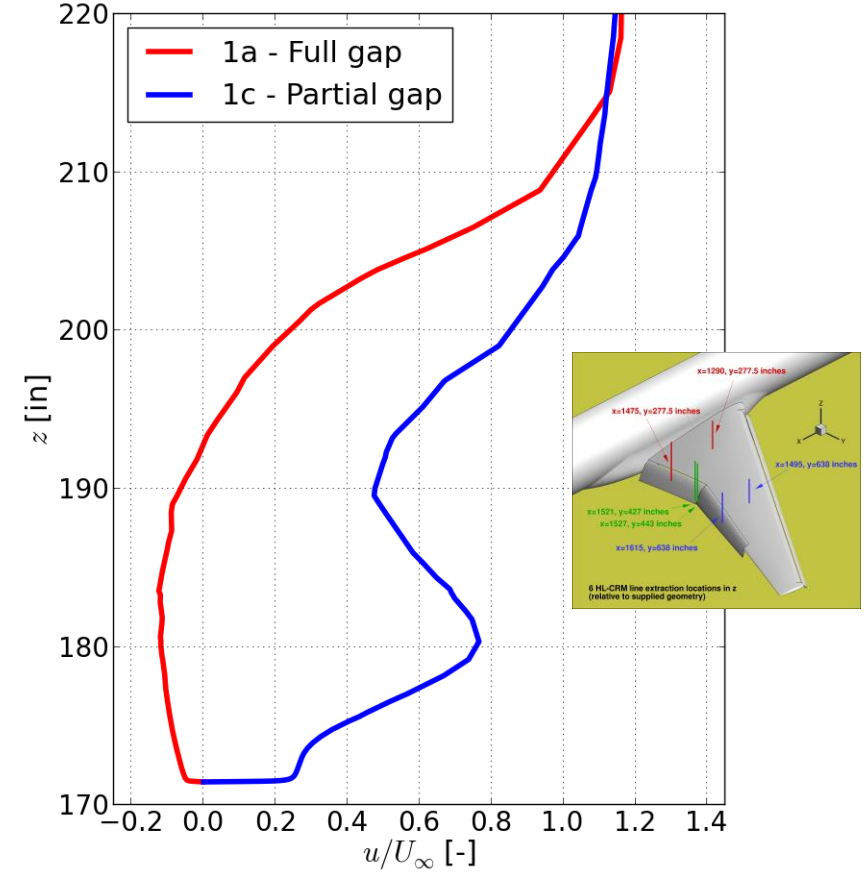
Skin Friction Coefficient



# HL-CRM – Impact of the flap gap (SSTa1)

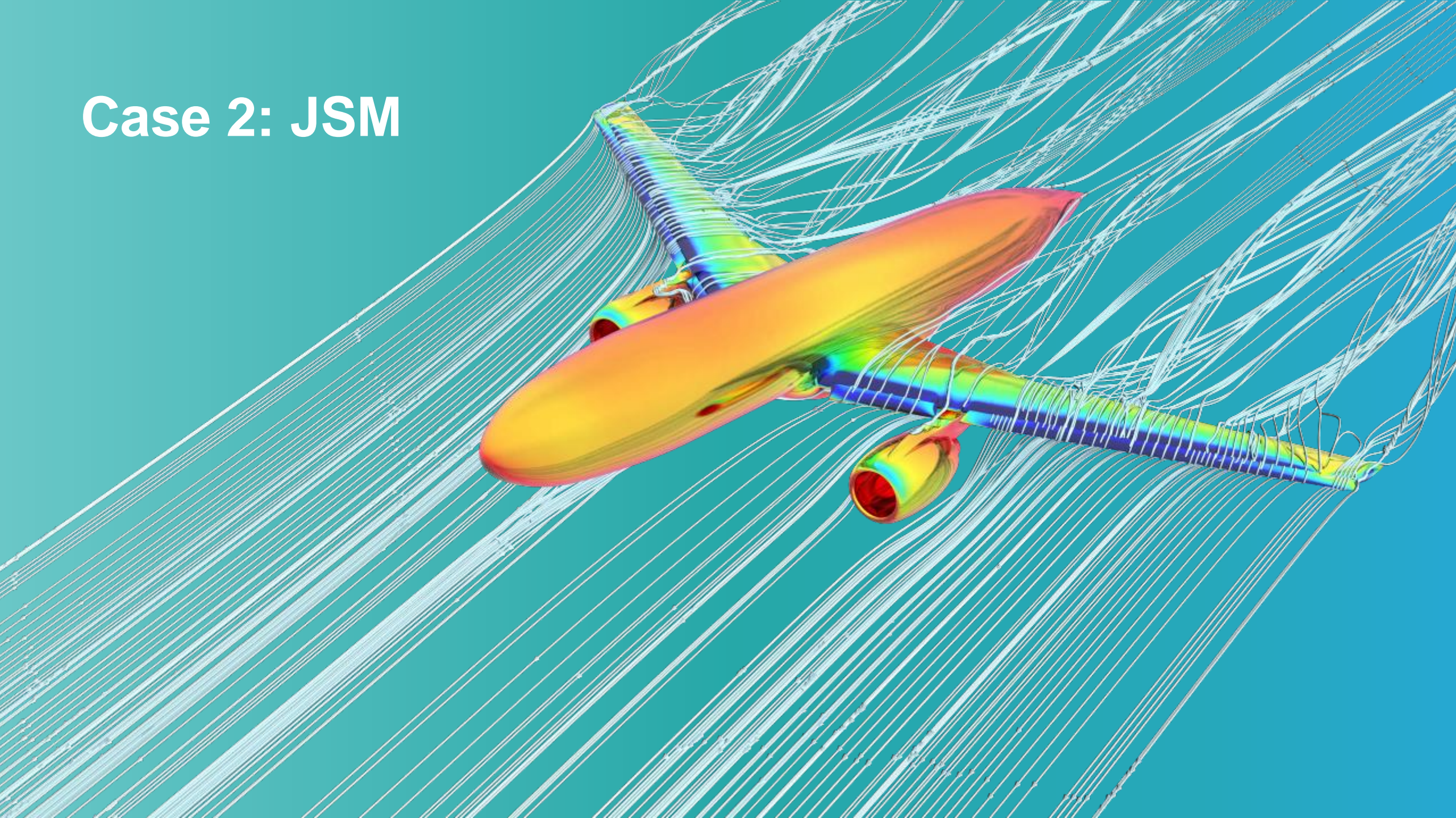


Velocity profile x=1527 y=443 (16°)



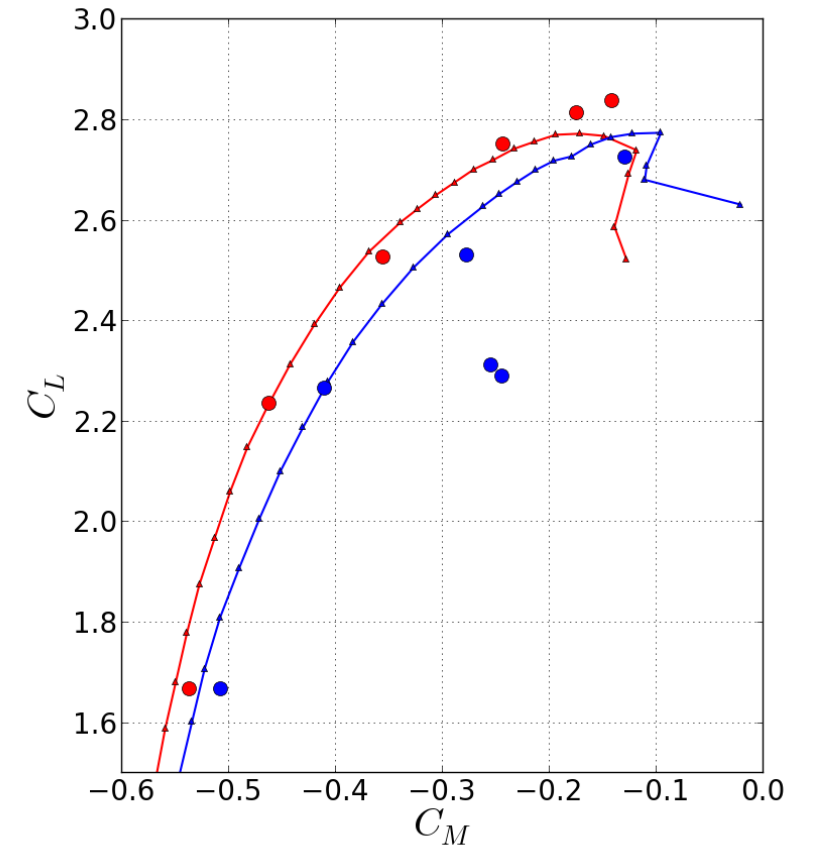
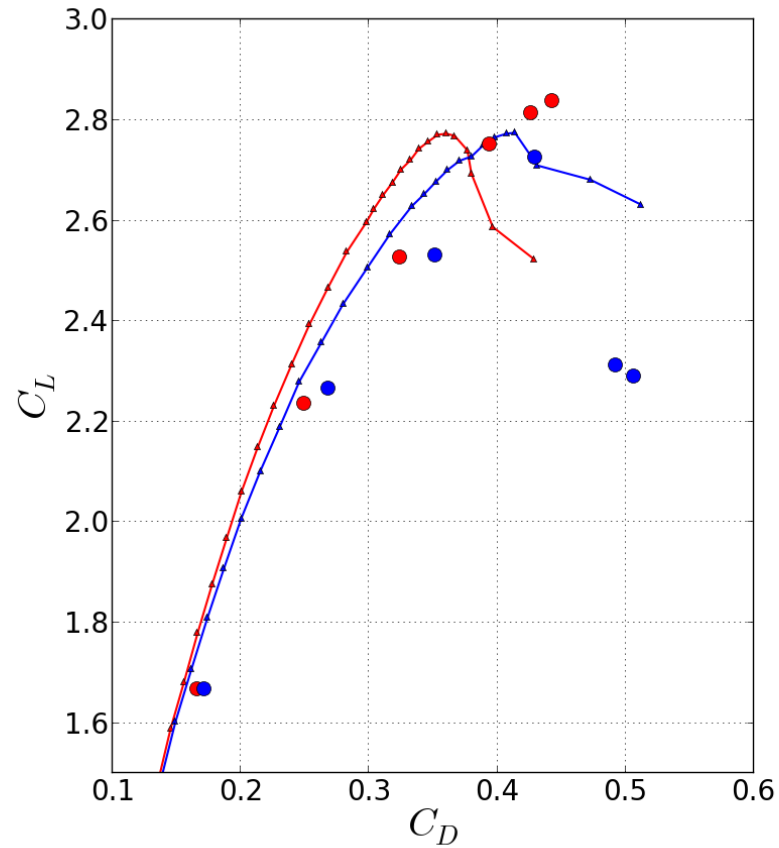
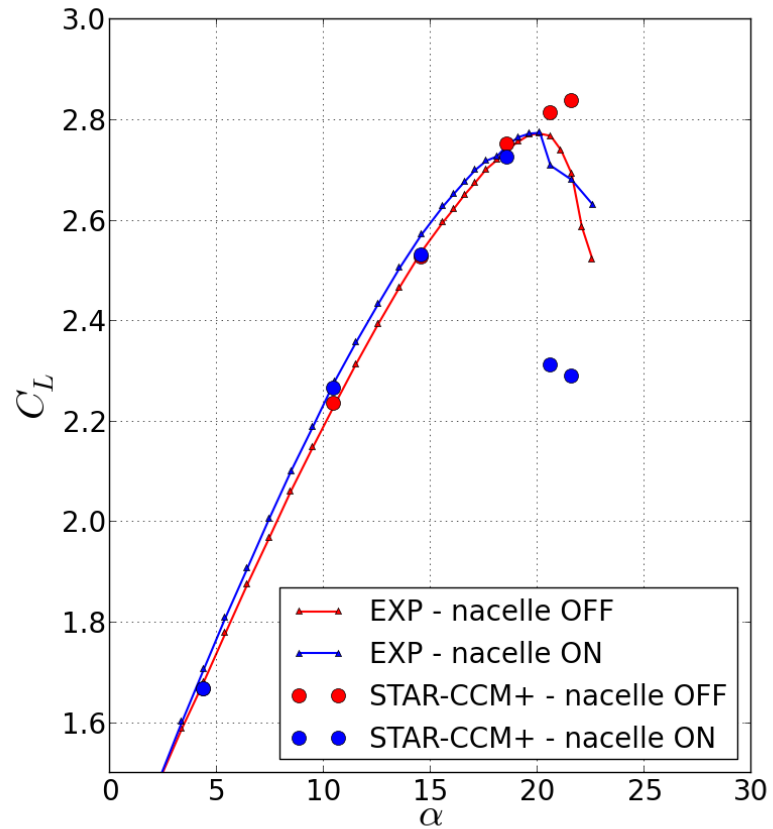
Alpha	$\Delta CL$	$\Delta CD$	$\Delta CM$
8	+2.26%	+0.36%	-0.98%
16	+2.16%	+1.57%	-4.19%

## Case 2: JSM



# Overview of JSM results

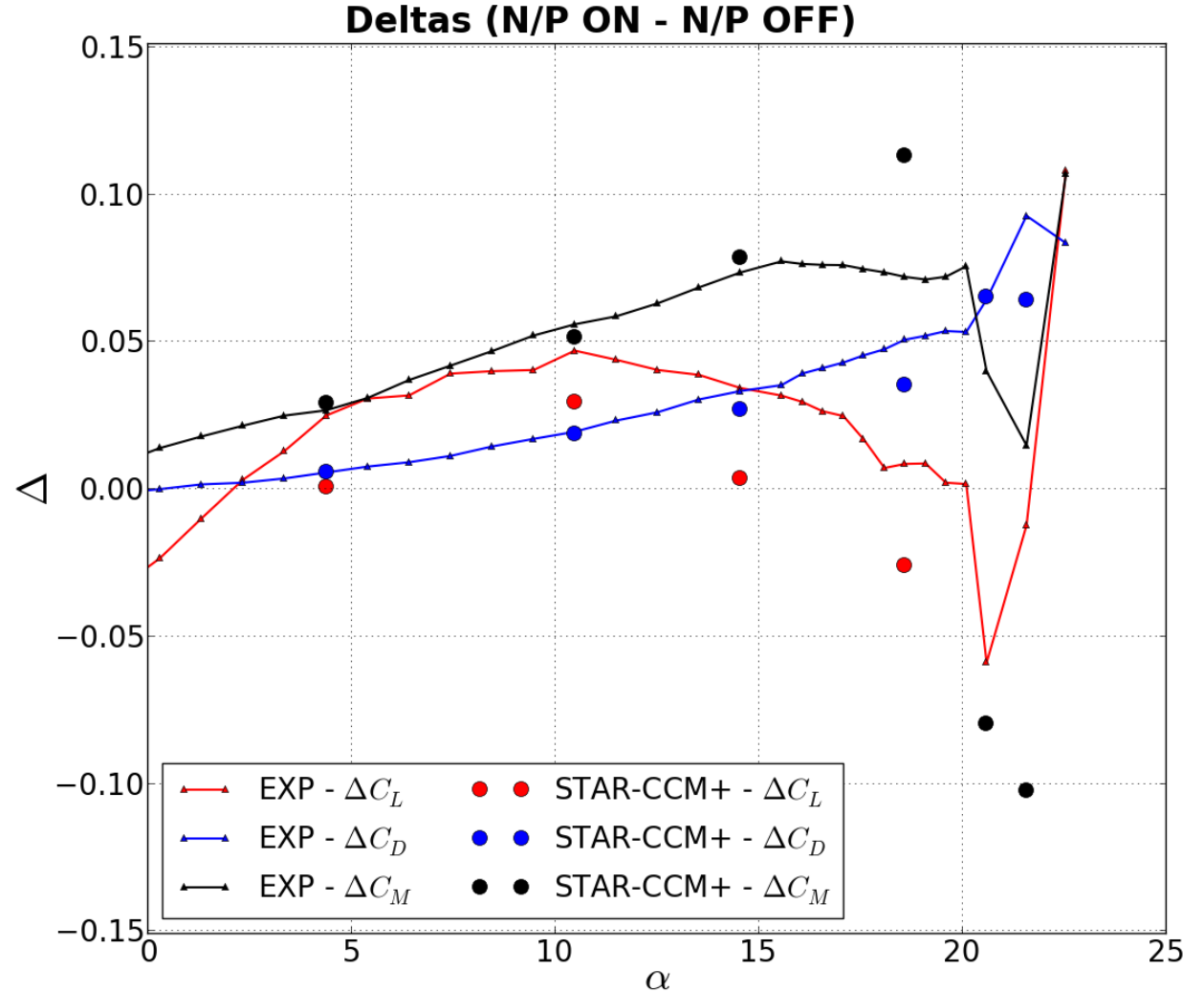
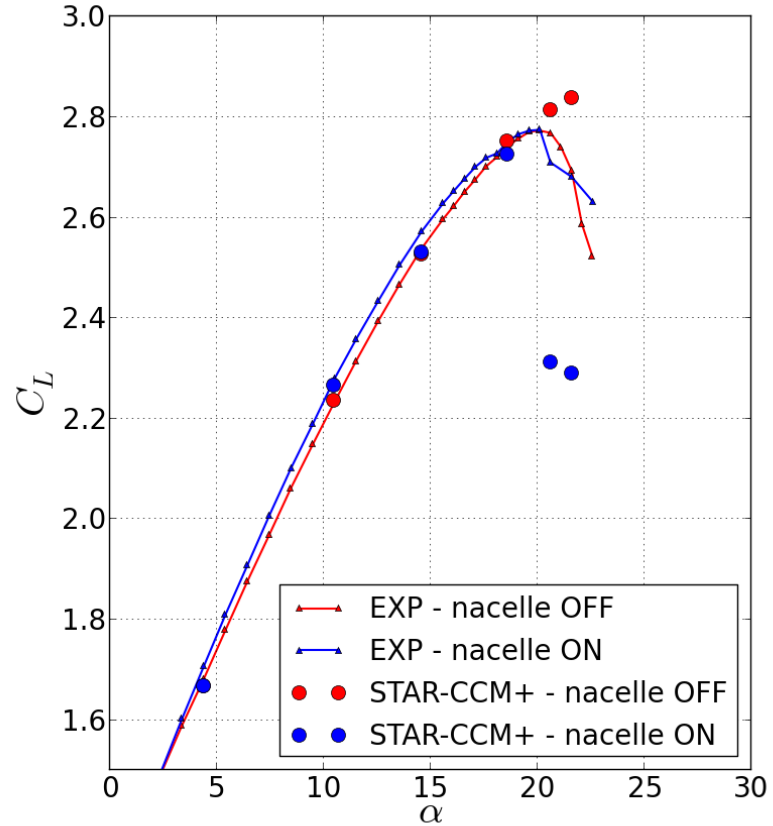
## Nacelle installation study (SSTa1)



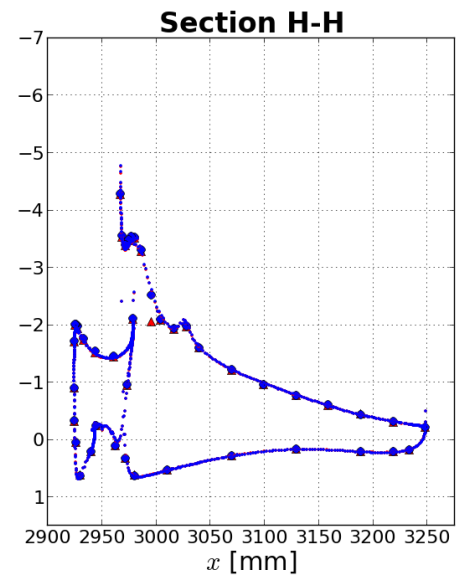
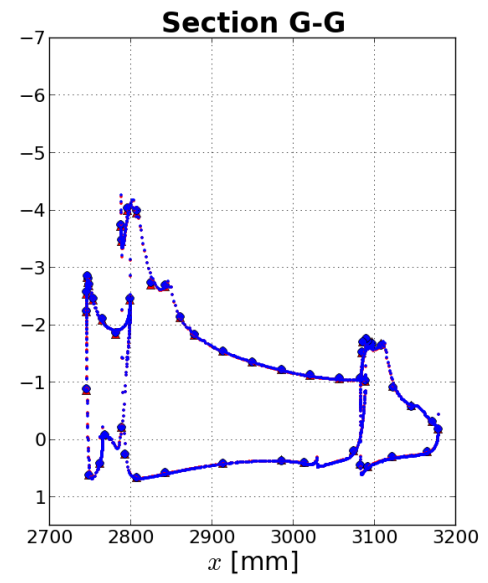
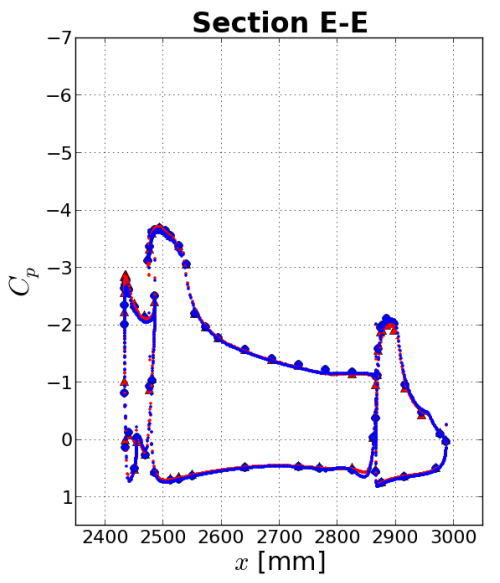
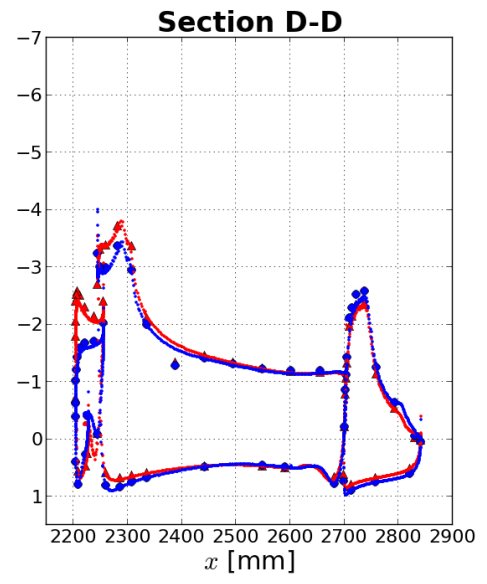
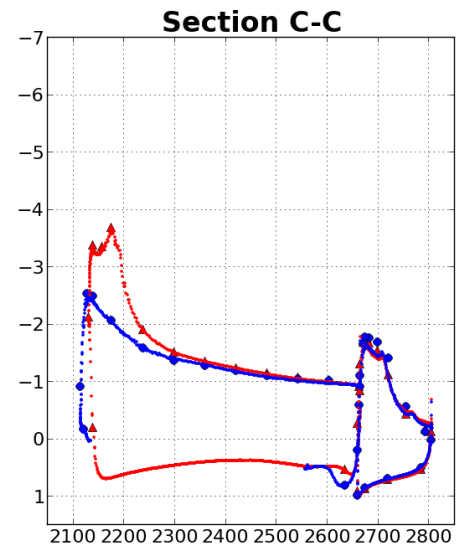
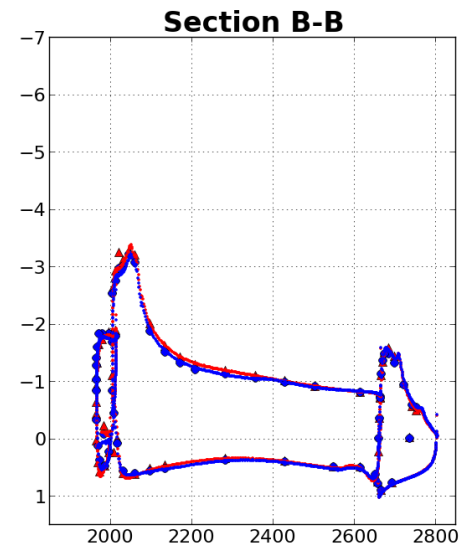
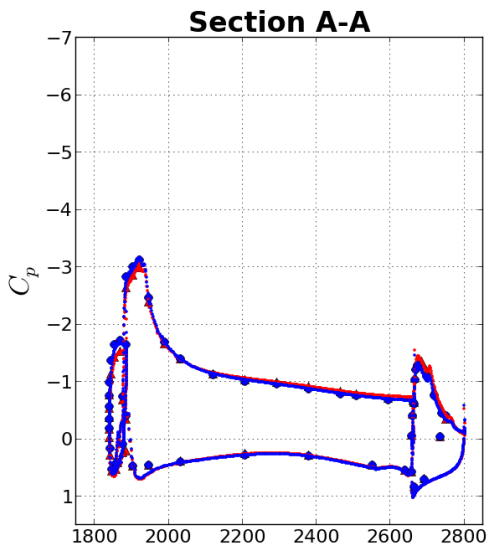


# Overview of JSM results

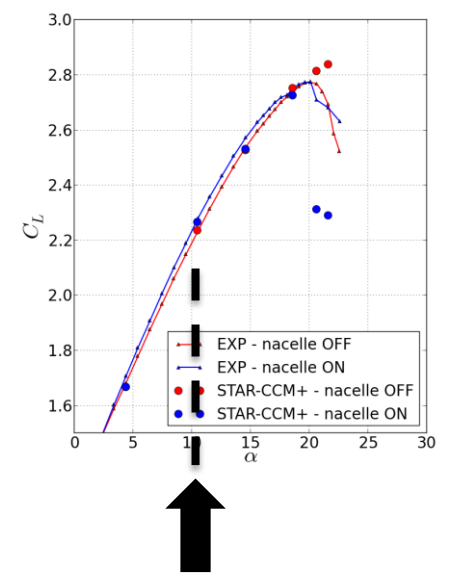
## Nacelle installation study (SSTa1)



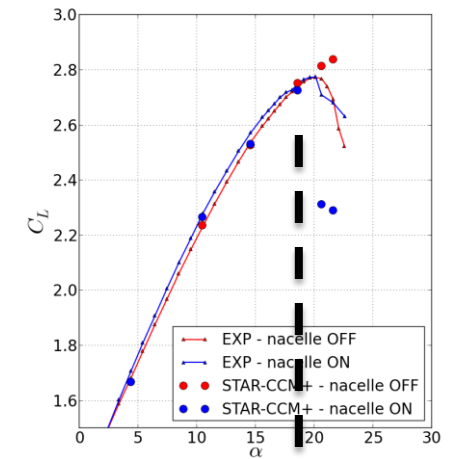
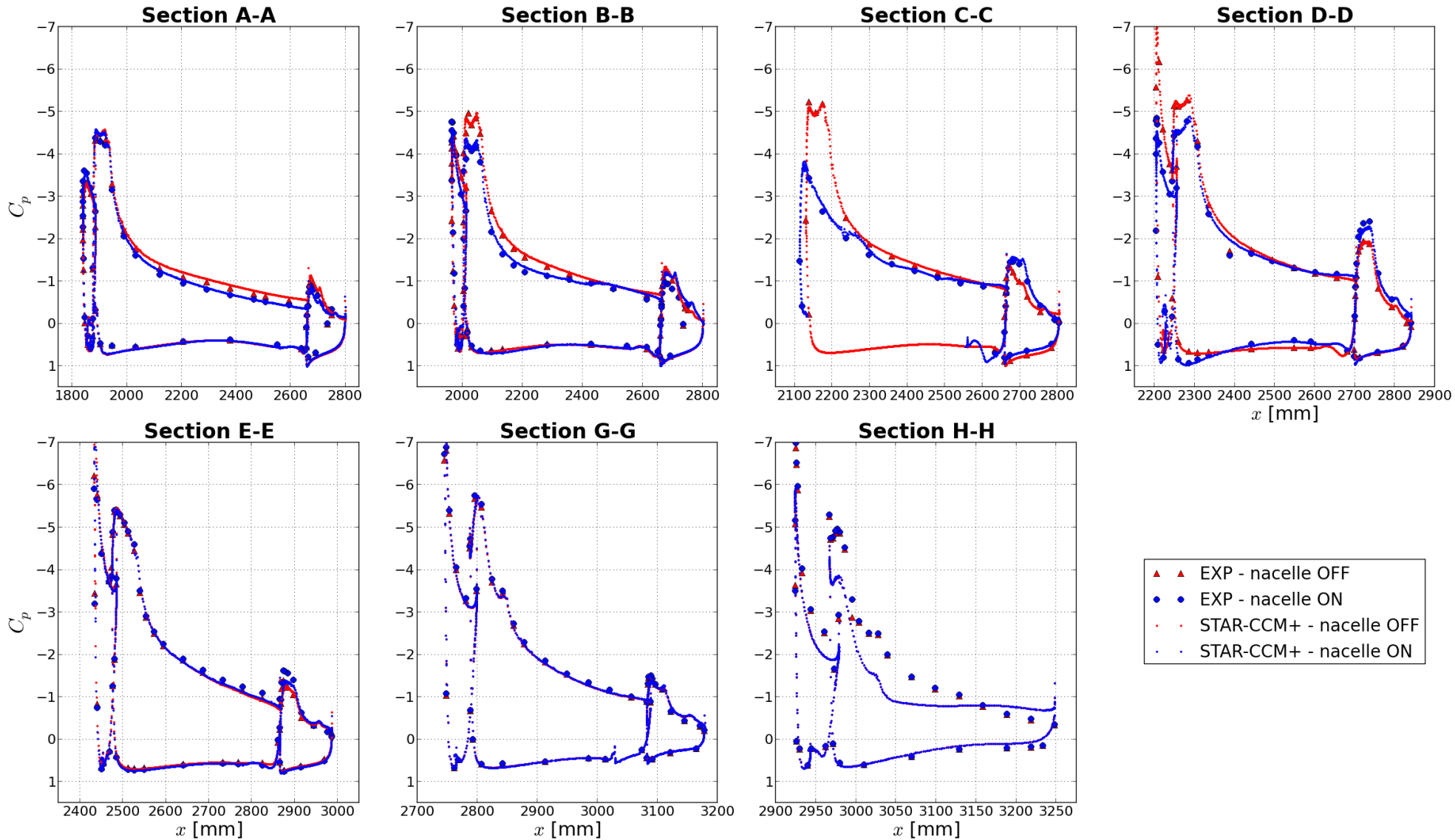
# Pressure Coefficient - $\alpha = 10.47$ deg



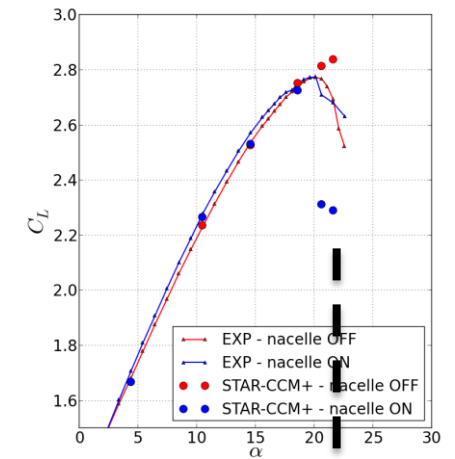
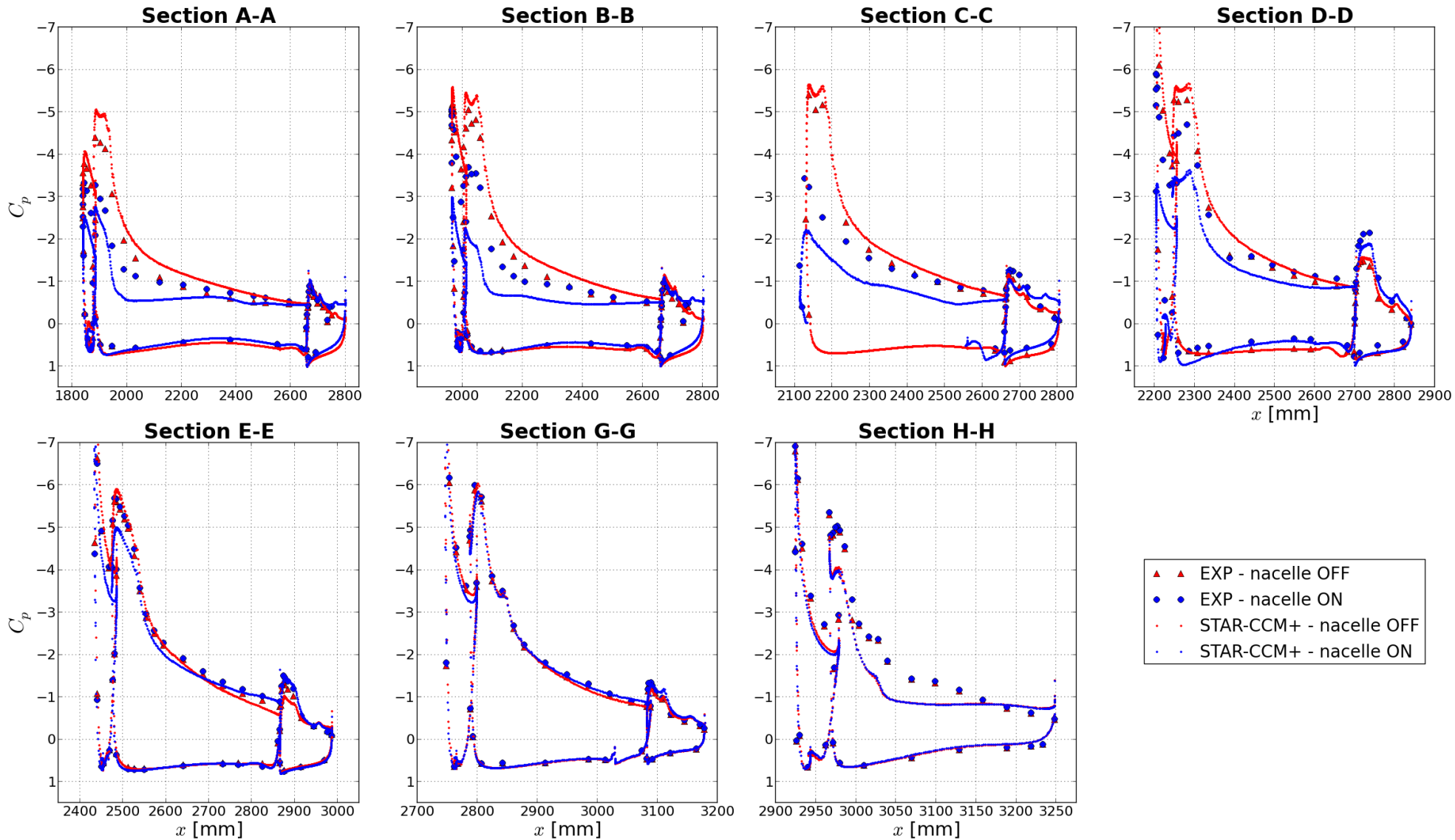
- ▲ EXP - nacelle OFF
- EXP - nacelle ON
- STAR-CCM+ - nacelle OFF
- STAR-CCM+ - nacelle ON



# Pressure Coefficient - $\alpha = 18.58$ deg



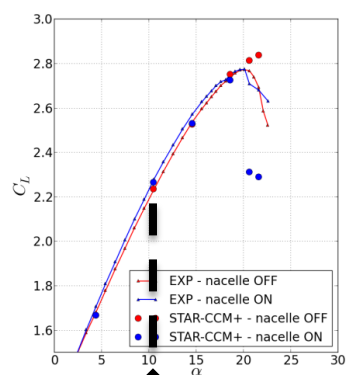
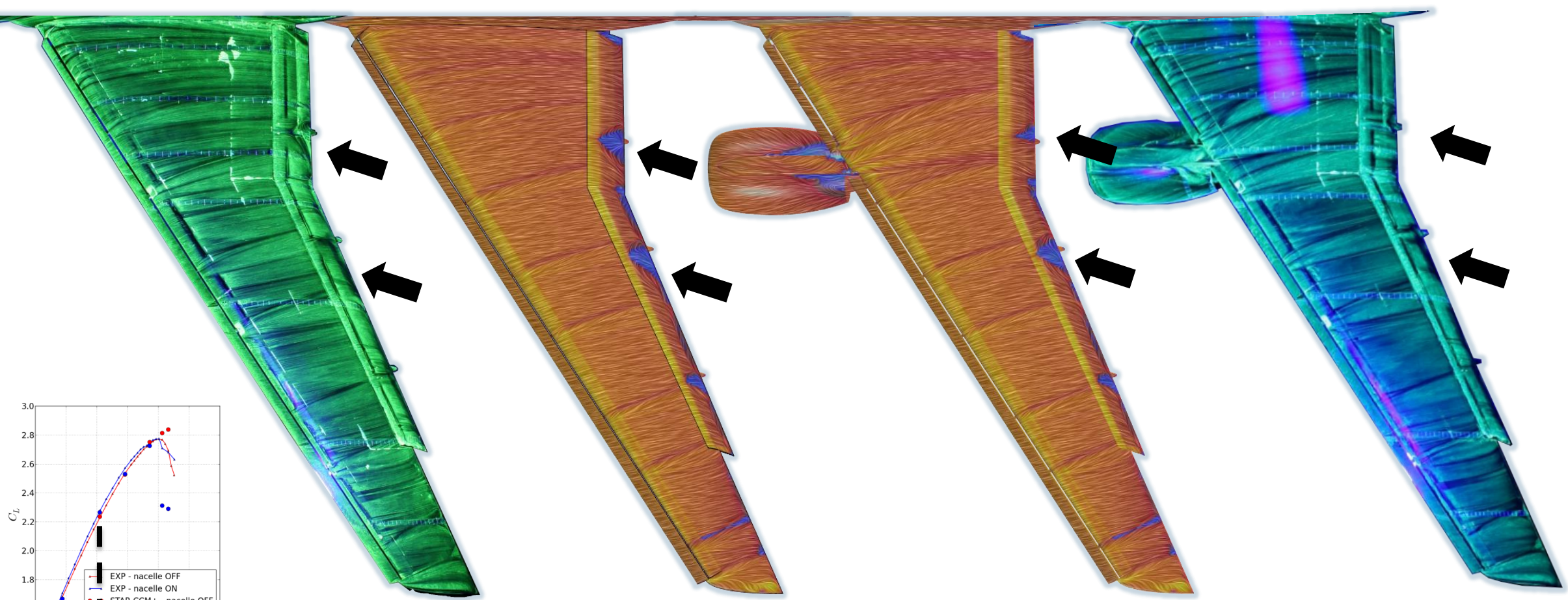
# Pressure Coefficient - $\alpha = 21.57$ deg



# Flow visualisation on the upper wing surface

$\alpha = 10.47$  deg

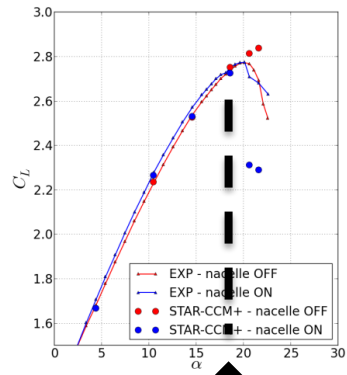
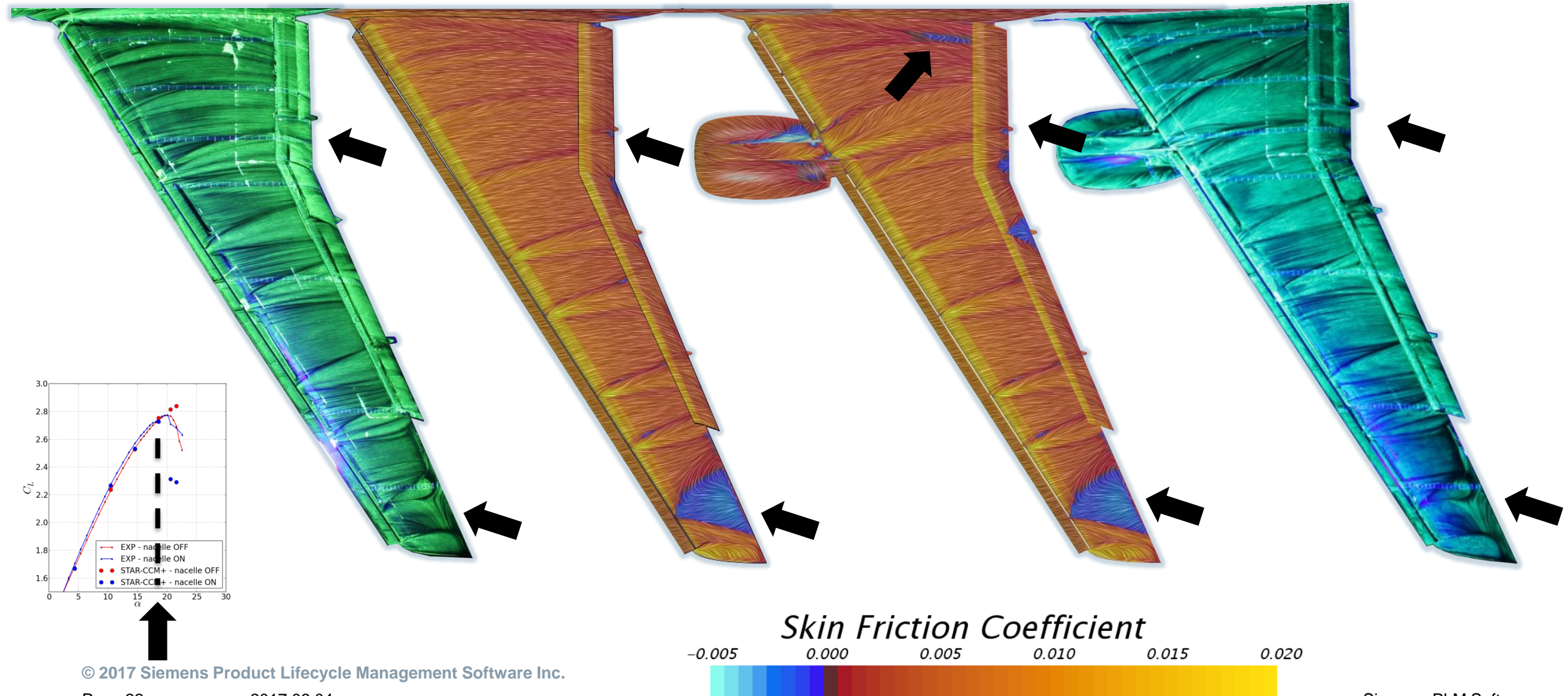
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# Flow visualisation on the upper wing surface

$\alpha = 18.58$  deg

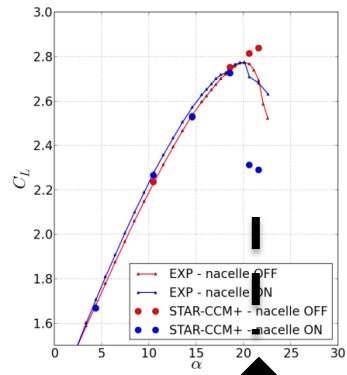
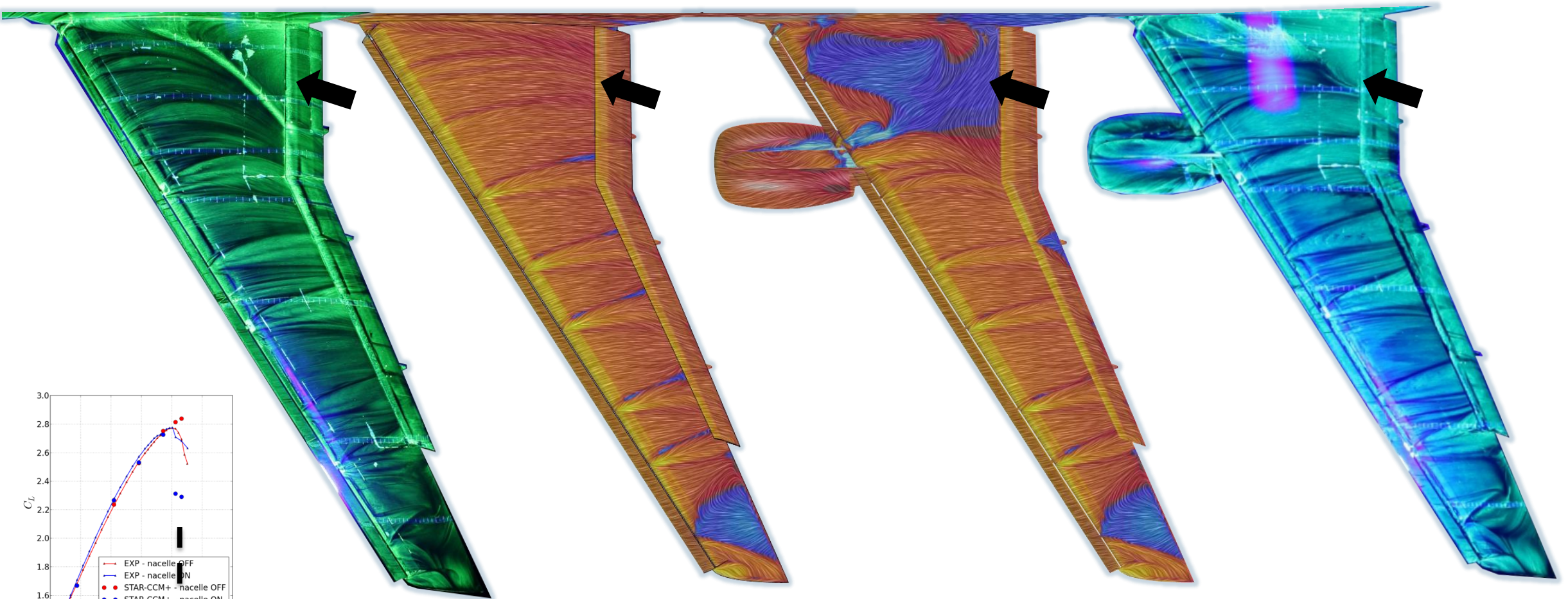
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# Flow visualisation on the upper wing surface

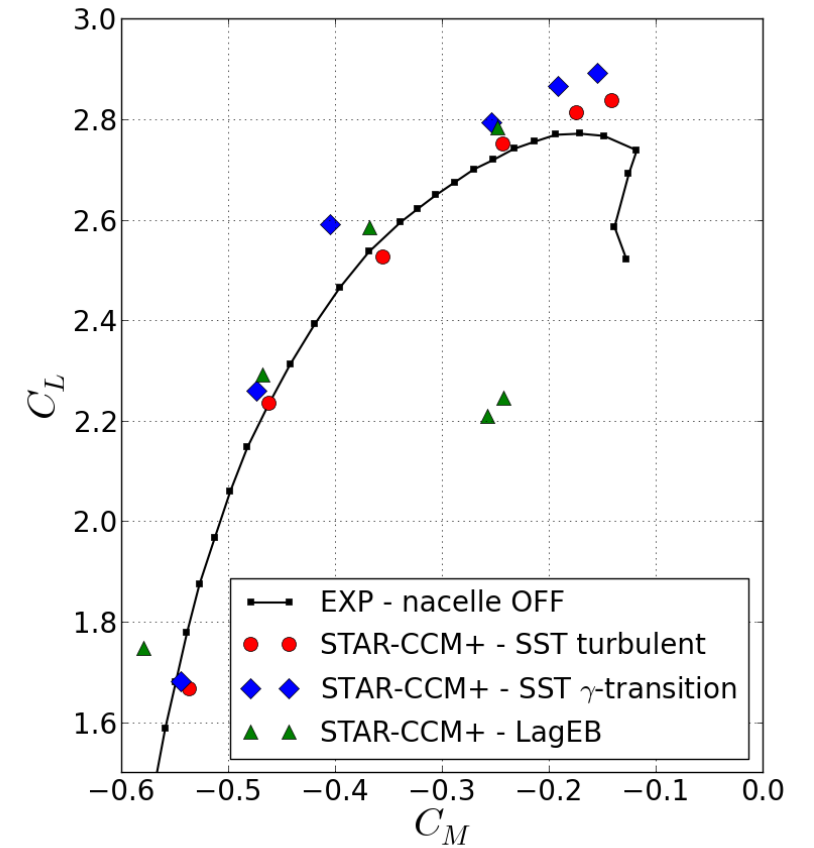
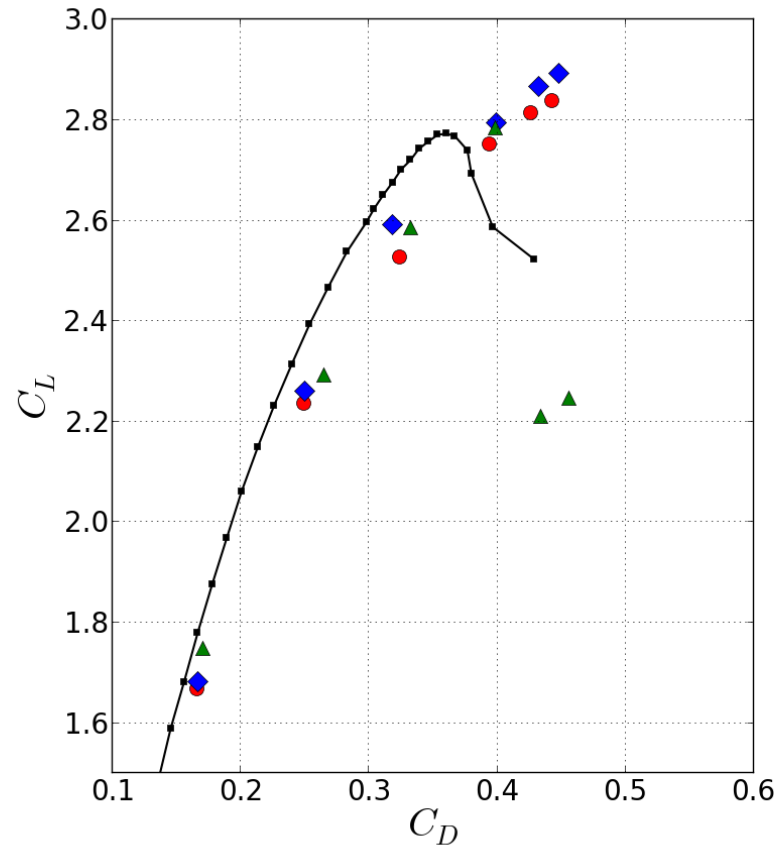
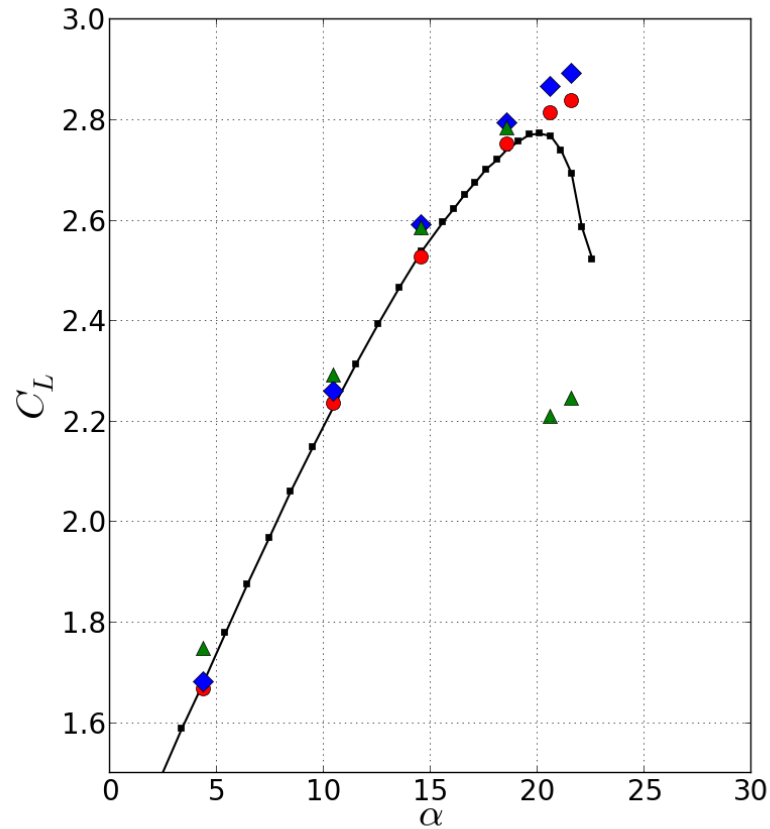
$\alpha = 21.57$  deg

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# Overview of JSM results

## Transition model study





# Flow visualisation on the upper wing surface

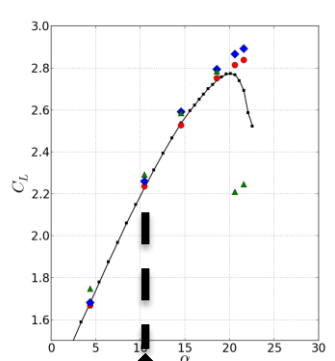
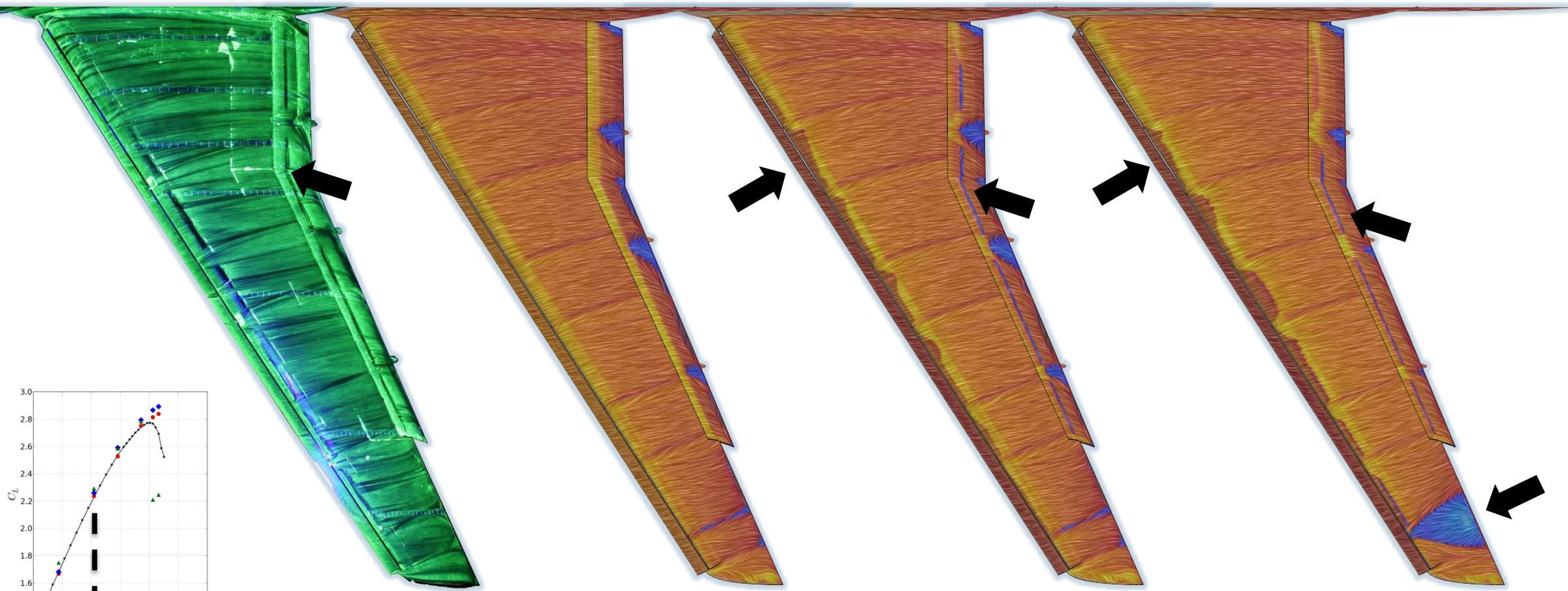
$\alpha = 10.47$  deg

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turbulent

$\gamma$ -transition

LagEB



*Skin Friction Coefficient*



# Flow visualisation on the upper wing surface

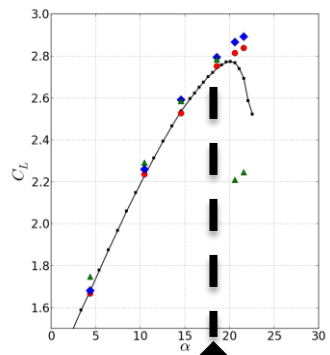
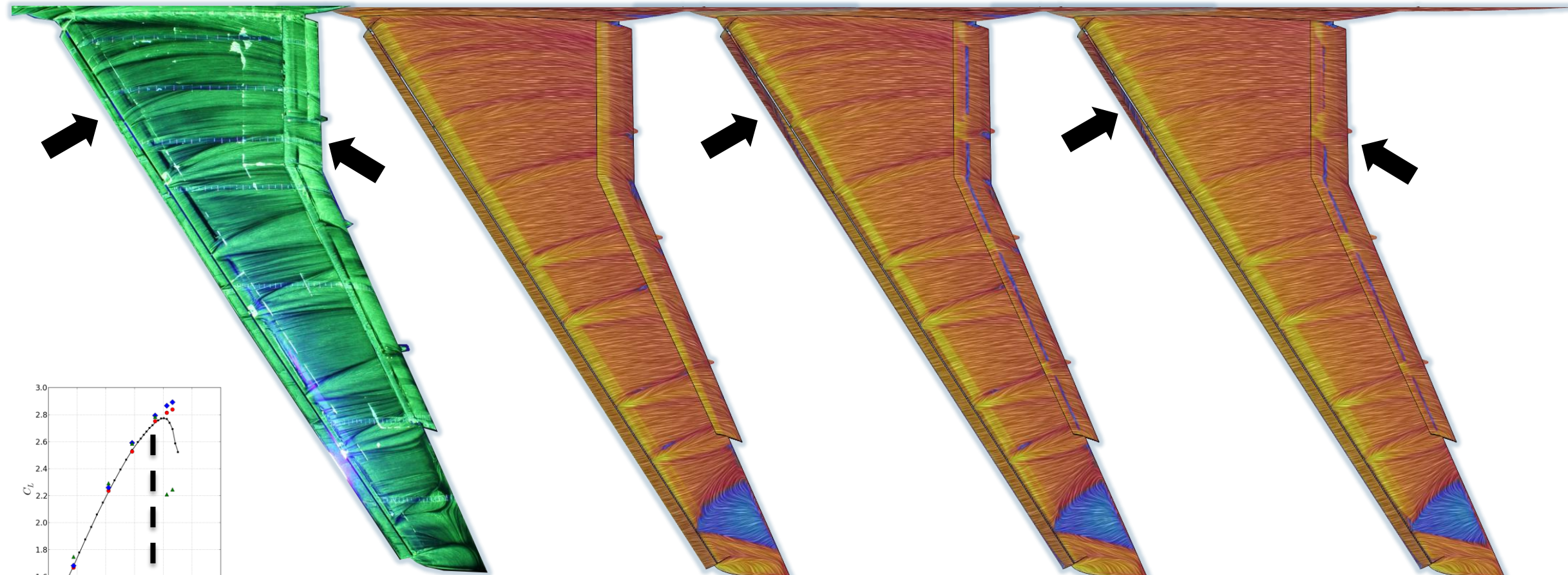
$\alpha = 18.59 \text{ deg}$

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*Ingenuity for life*

turbulent

$\gamma$ -transition

LagEB



*Skin Friction Coefficient*



# Flow visualisation on the upper wing surface

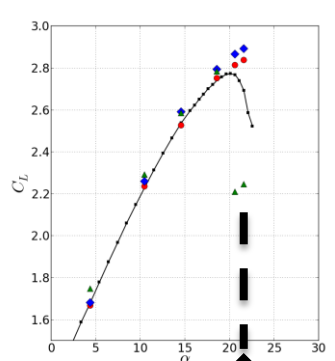
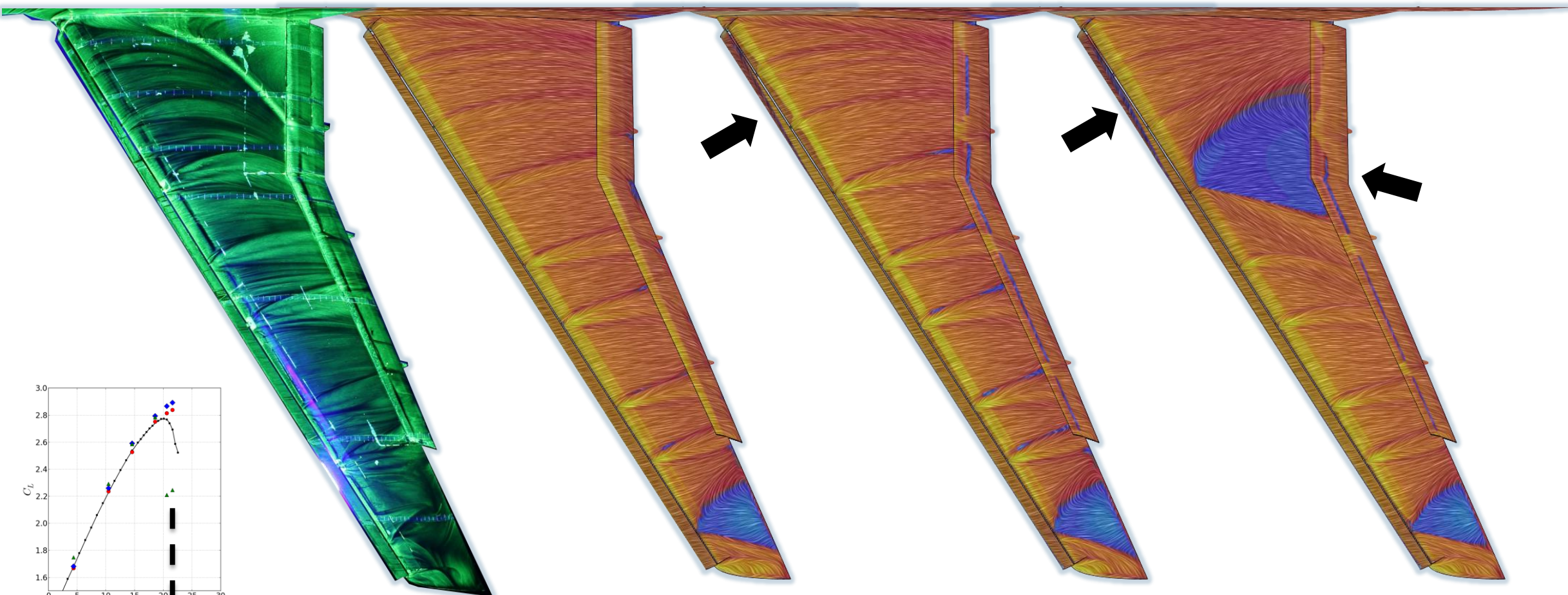
$\alpha = 21.57$  deg

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*Skin Friction Coefficient*



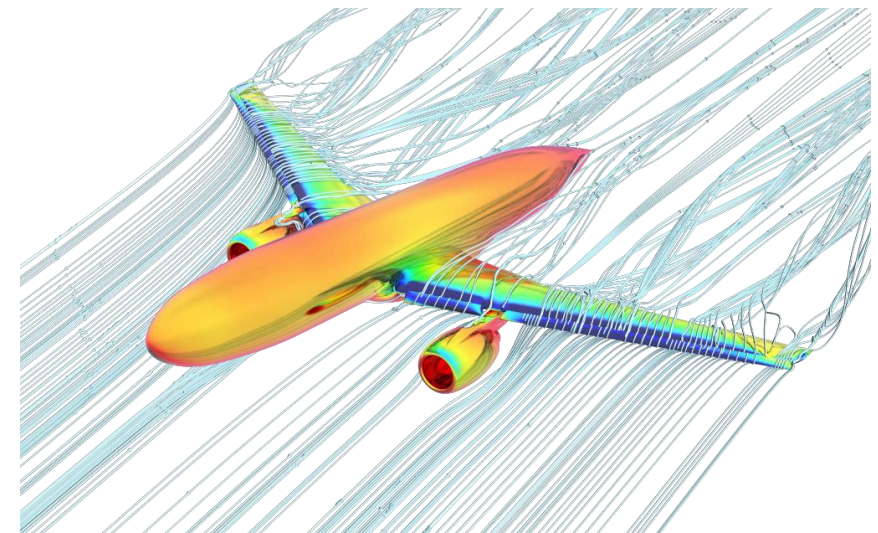
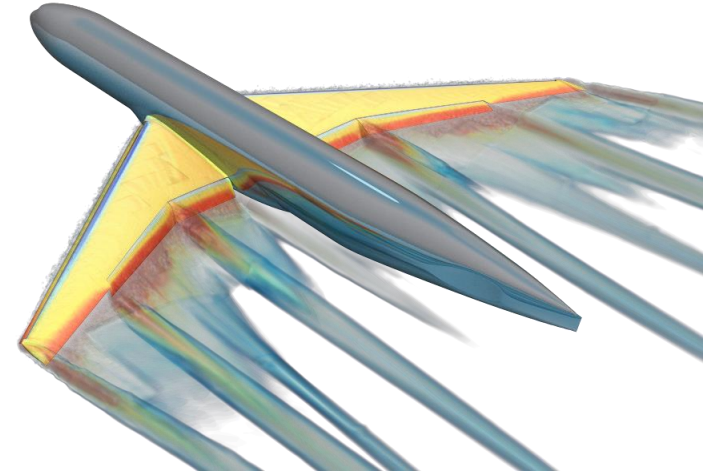
# Summary

## Case 1: HL-CRM

- Grid convergence study for 3 turbulent models:
  - SST
  - SA
  - LagEB
- Investigation of impact of gap sealing

## Case 2: JSM

- Nacelle installation study
- Investigated impact of transition modelling



## ACKNOWLEDGEMENTS

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