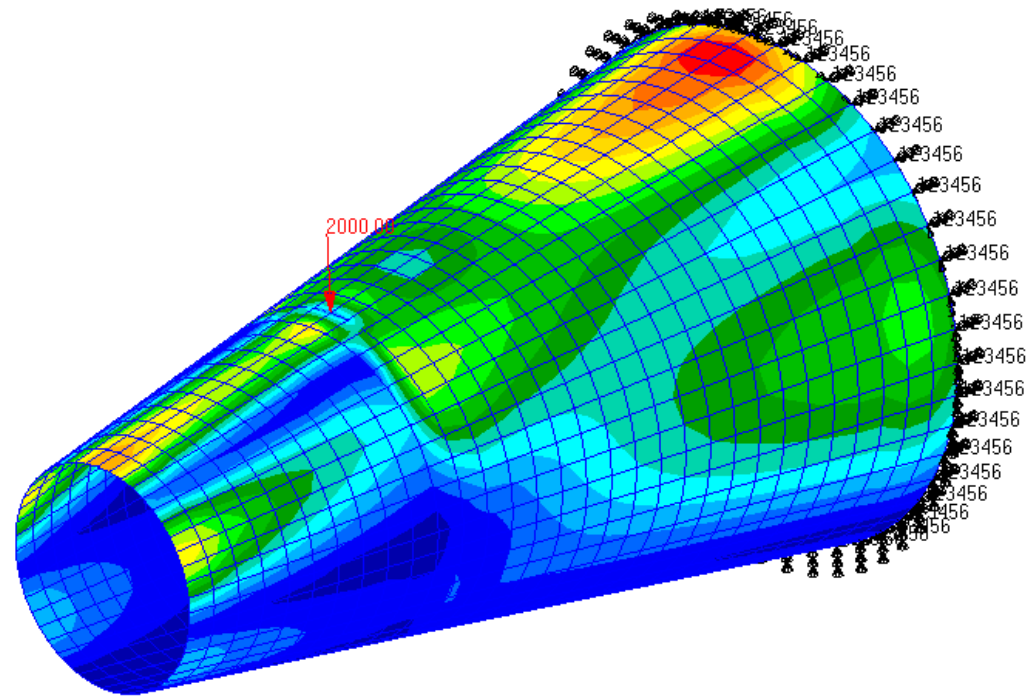
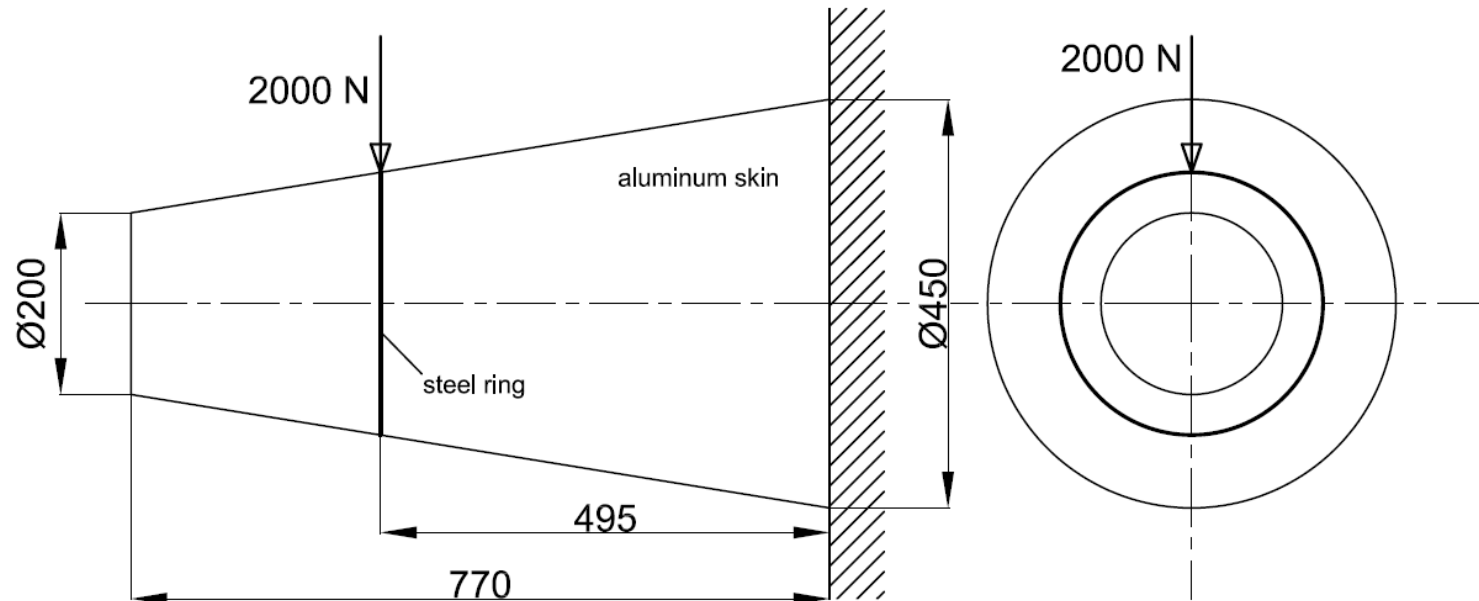


Mechanics of Thin-walled Structures

CONICAL



PROBLEM DESCRIPTION



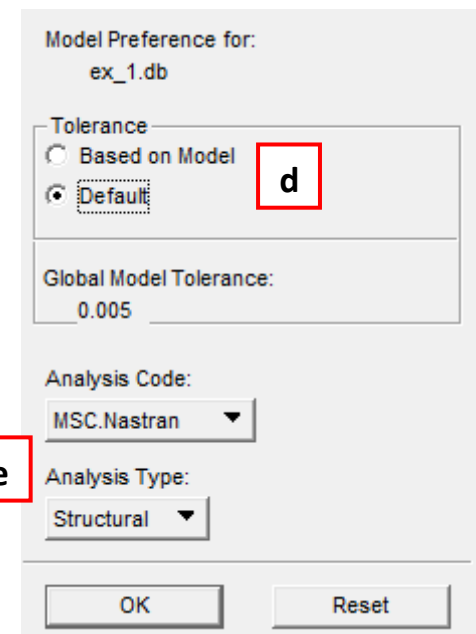
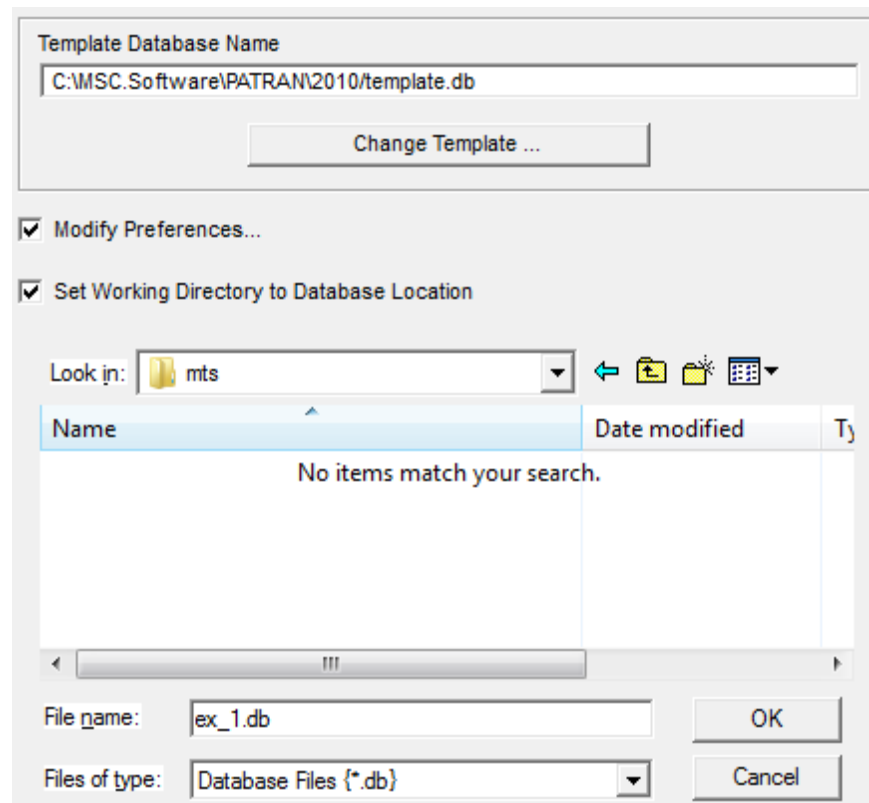
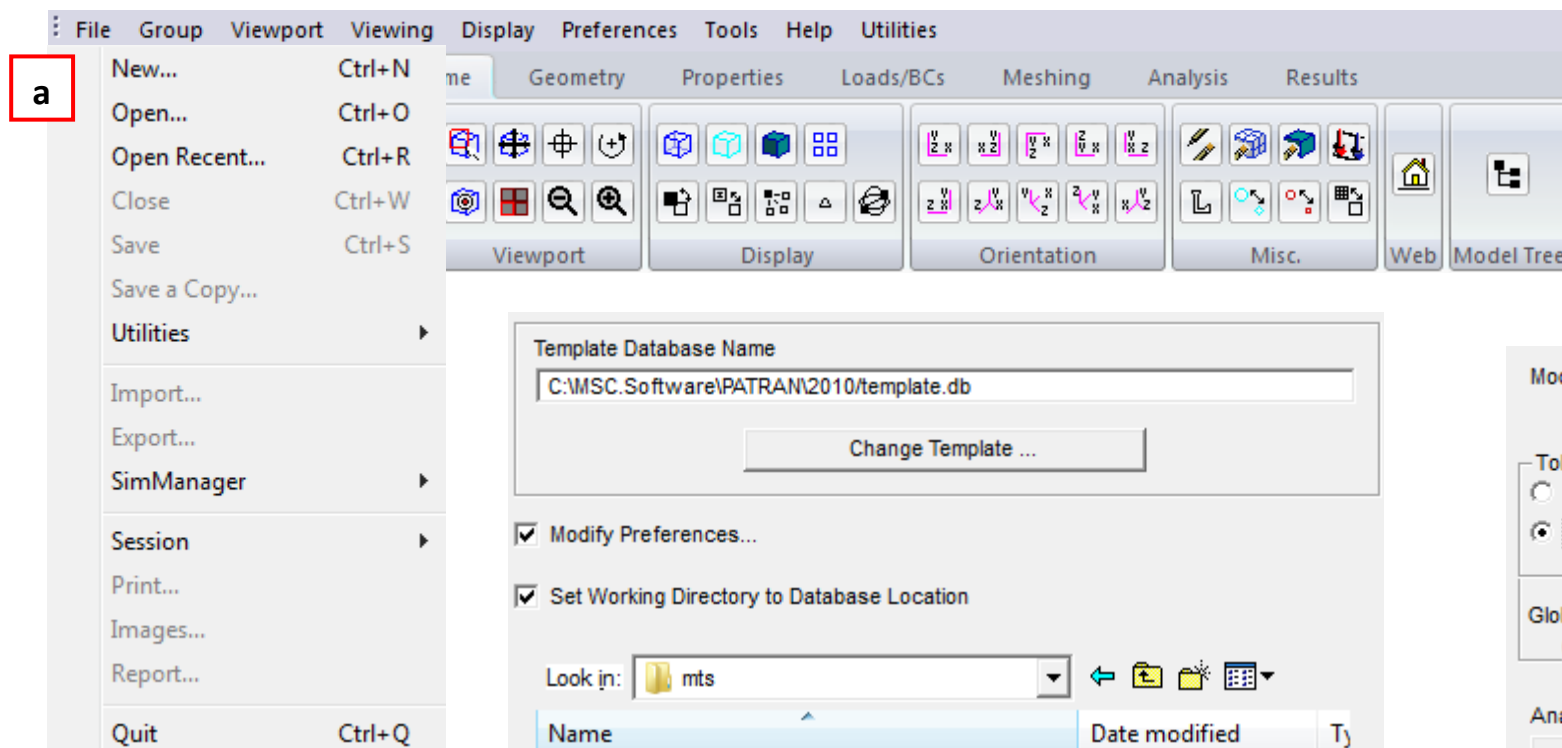
Static, linear analysis of thin-walled aluminum skin, reinforced by a stiffening ring, will be performed.

In the course of this exercise, students will gain basic knowledge about: **creating geometry and mesh; applying loads and boundary conditions; setting up static, linear analysis and results post-processing.**

Two cases will be considered:

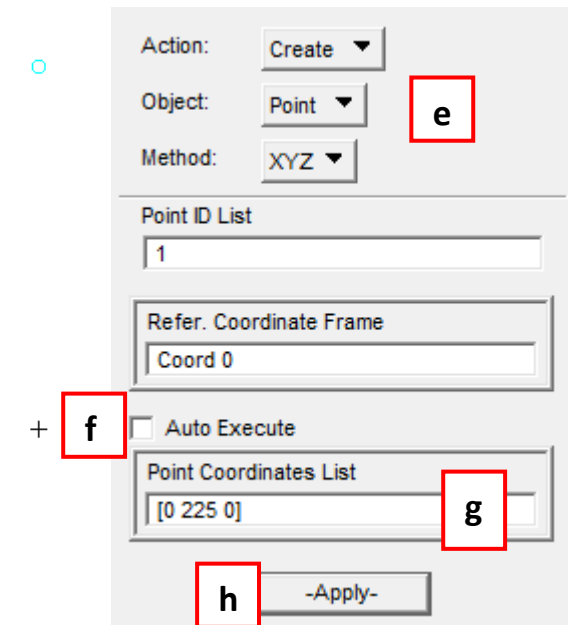
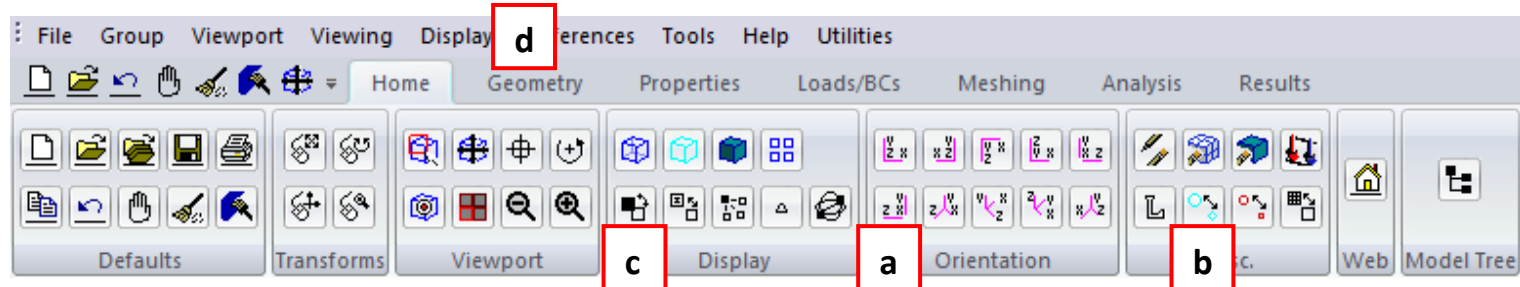
1. When stiffening ring is (relatively) **rigid**
2. When stiffening ring is made of **steel** (and its deformations influence the deformation of the aluminum skin)

Units: mm, N, MPa



Create a new database:

- File / New...**
- Enter **ex_1.db** as the File name
- Click **OK**
- Select **Default**
- Click **OK**

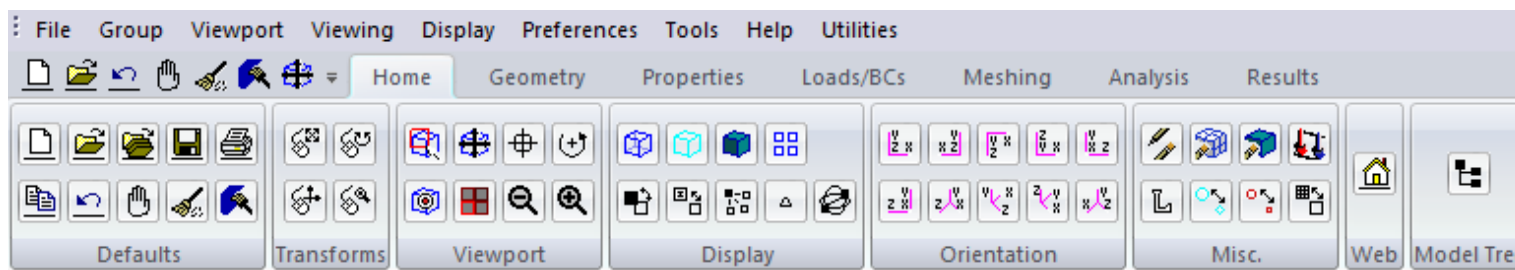


Create geometry points:

- Click on the **Right side view** icon
- Click on the **Point size** icon
- Click on the **Cycle Background** icon
- Click on the **Geometry** icon
- Geometry: **Create/Point/XYZ**
- Uncheck **Auto Execute**
- Enter **[0 225 0]** as the Point Coordinates List
- Click **Apply**
- Create two more points using coordinates: **[0 100 770]**, **[0 144 495]**

Auto-Execute Option

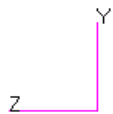
The Auto-Execute option will automatically apply the operation when the last field is filled in.



Point 2

Point 3

Point 1



Create curves:

- Geometry: **Create/Curve/Point**
- Option: **2 Point**
- Uncheck **Auto Execute**
- Select **Point 1** as the starting point and **Point 3** as the ending point
- Click **Apply**
- Create one more curve using **Point 3** and **Point 2**

Remark: Points numbers may be different.

Action: Create ▾

Object: Curve ▾

Method: Point ▾

Curve ID List

1

Option: 2 Point ▾

☐ Auto Execute

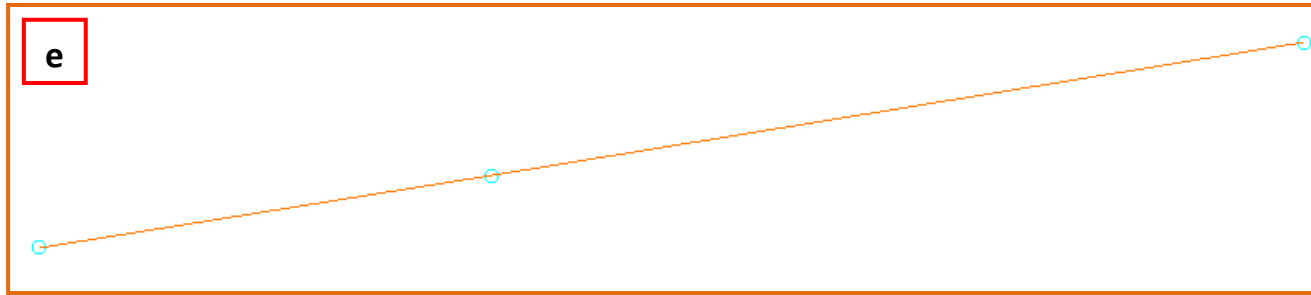
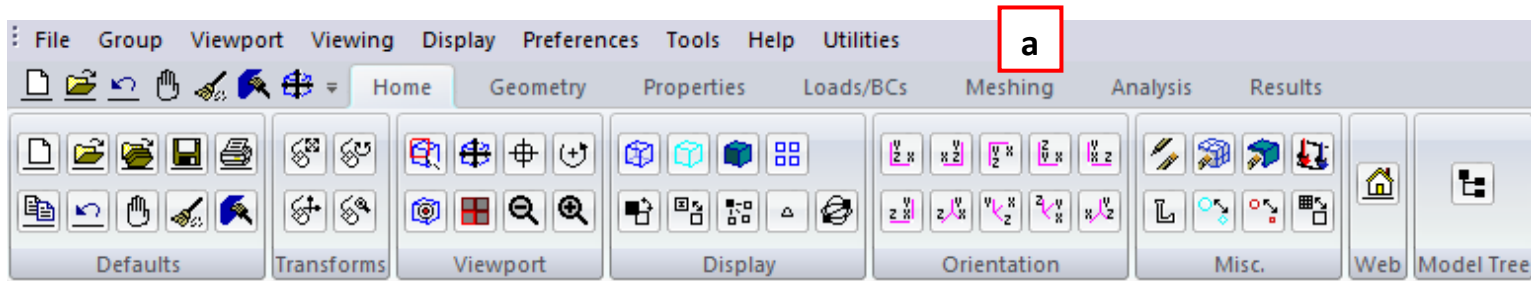
Starting Point List

Point 1

Ending Point List

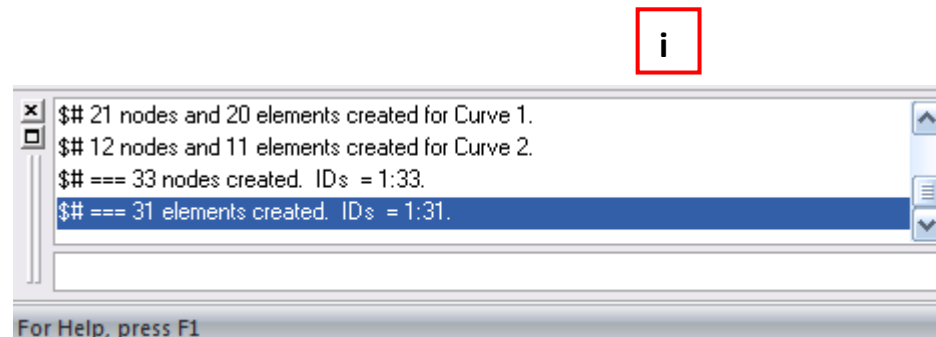
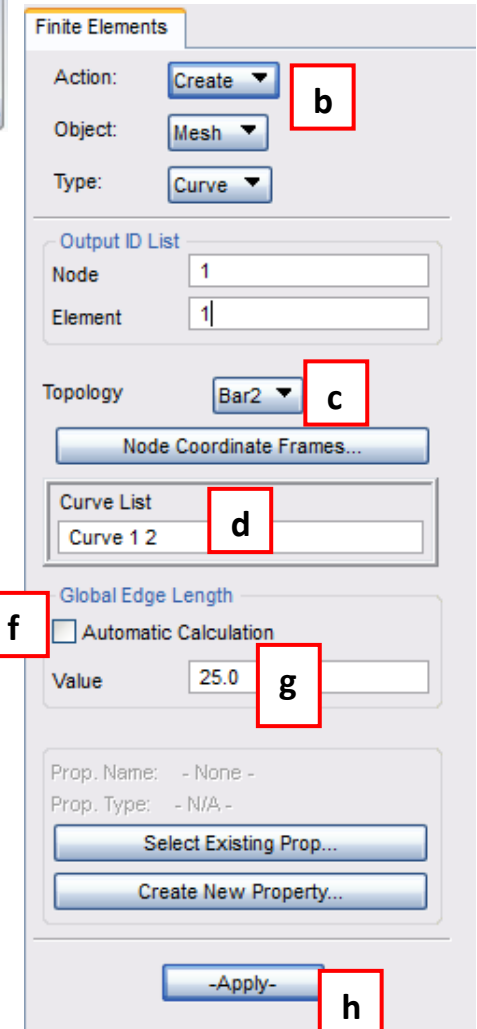
Point 3

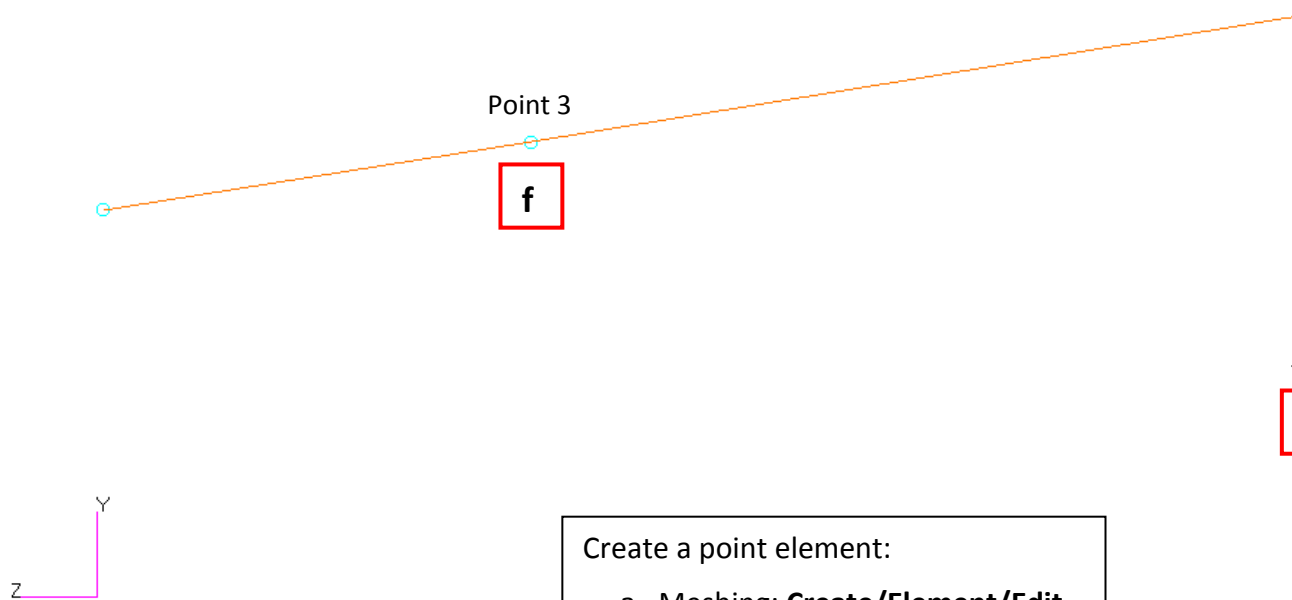
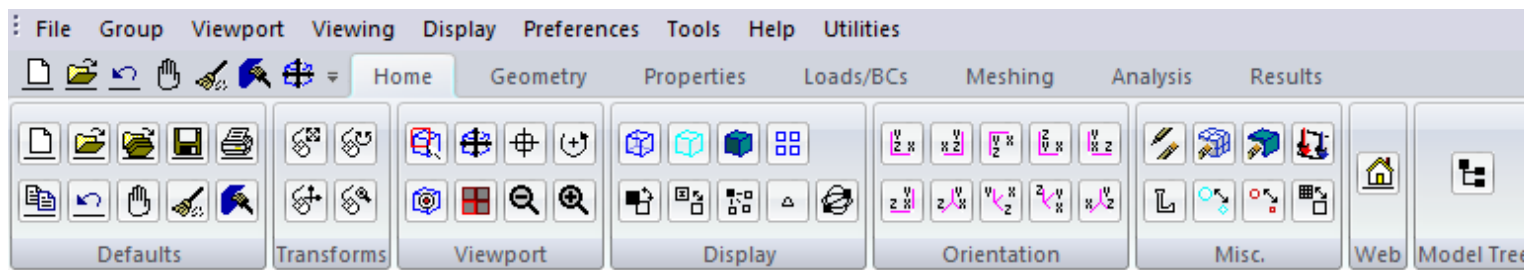
-Apply-



Mesh the curves:

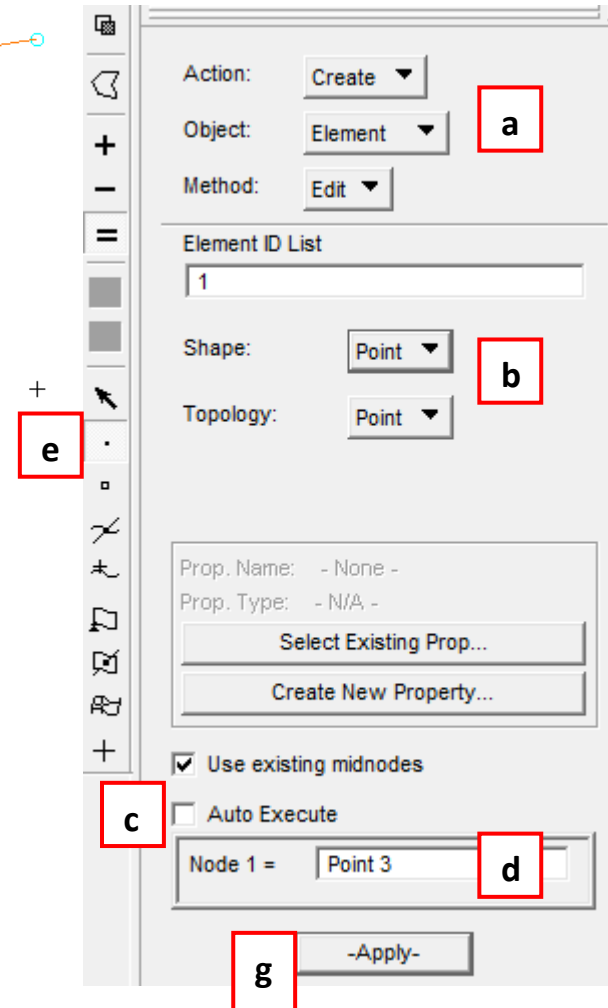
- Click on the **Meshing** tab
- Meshing: **Create/Mesh/Curve**
- Topology: **Bar2**
- Click on the **Curve List** panel
- Select all curves by clicking and dragging the mouse
- Uncheck **Automatic Calculation**
- Enter **25** as the Value of the Global Edge Length
- Click **Apply**
- Check message window whether elements were created

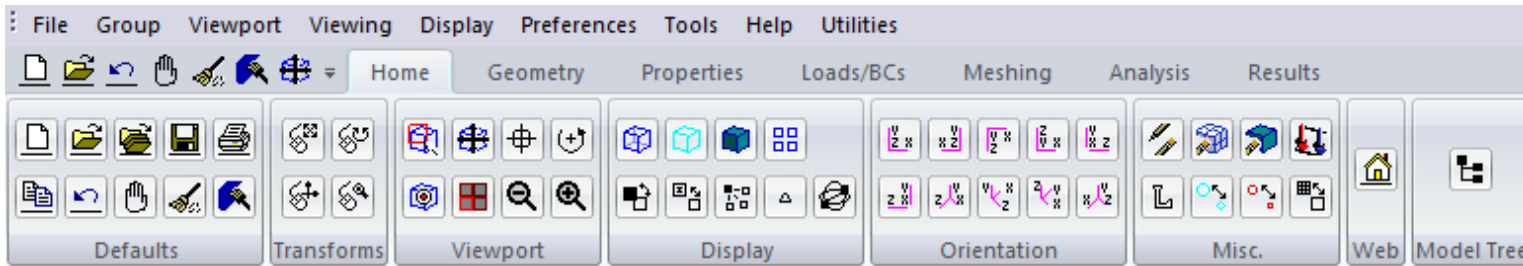




Create a point element:

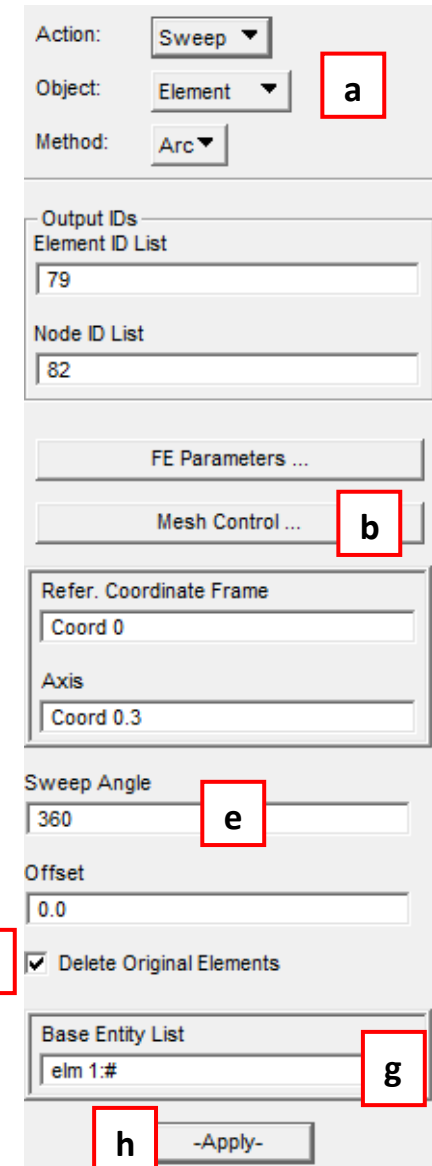
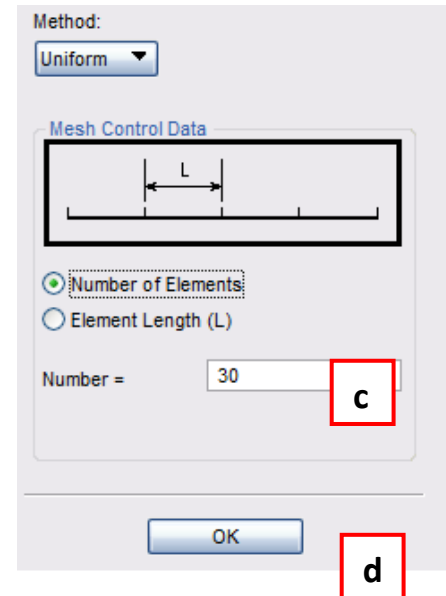
- Meshing: **Create/Element/Edit**
- Shape and Topology: **Point**
- Uncheck **Auto Execute**
- Click on the **Node 1** panel
- Select **Point** icon
- Select **Point 3**
- Click **Apply**

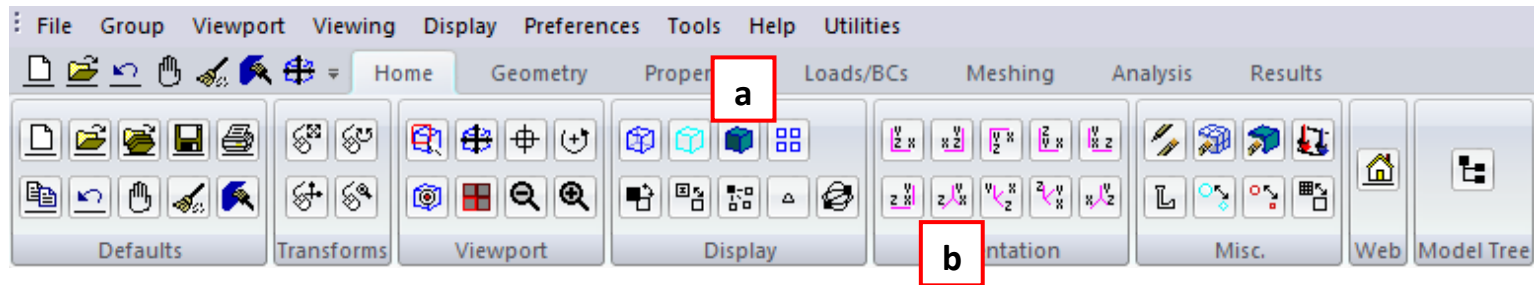




Sweep the elements to create the final mesh:

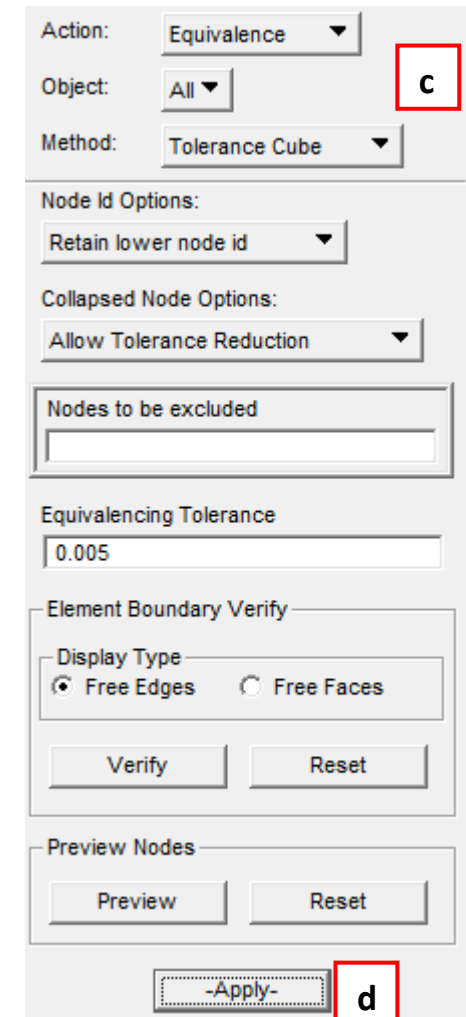
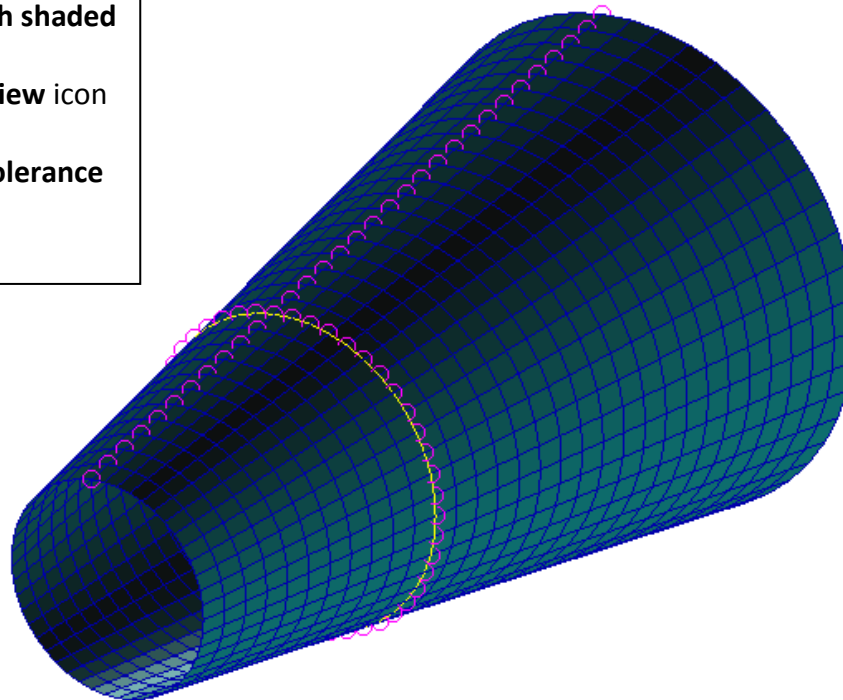
- Meshing: **Sweep/Element/Arc**
- Click **Mesh Control...**
- Enter **30** as the Number of Elements
- Click **OK**
- Enter **360** as the Sweep Angle
- Check **Delete Original Elements**
- In the **Base Entity List** enter **elm 1:#** to select all previously created elements
- Click **Apply**

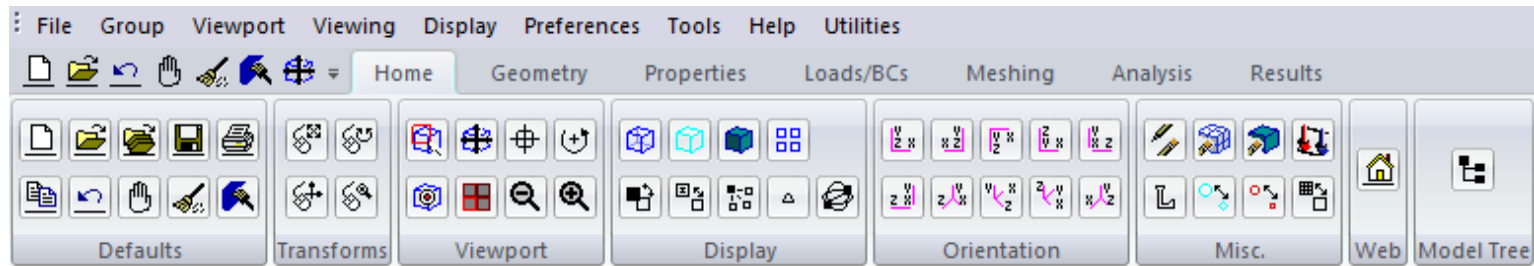




Delete the duplicate nodes:

- Click on the **Smooth shaded** icon
- Click on the **Iso 1 View** icon
- Elements:
Equivalence/All/Tolerance Cube
- Click **Apply**

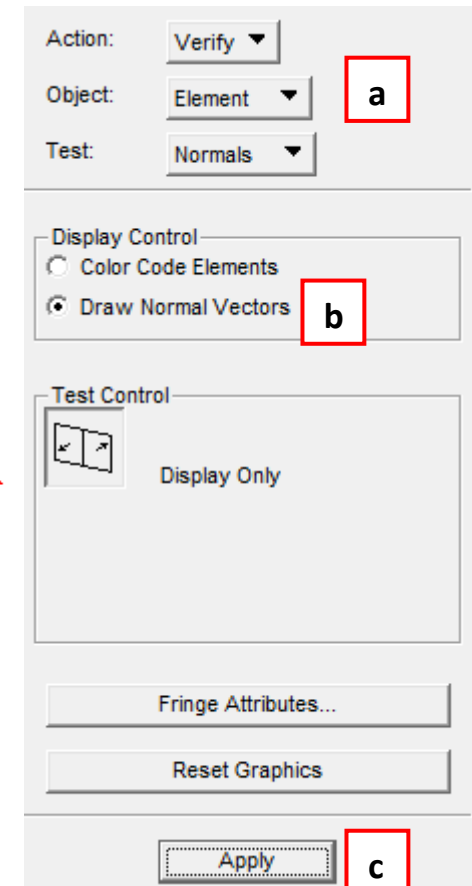
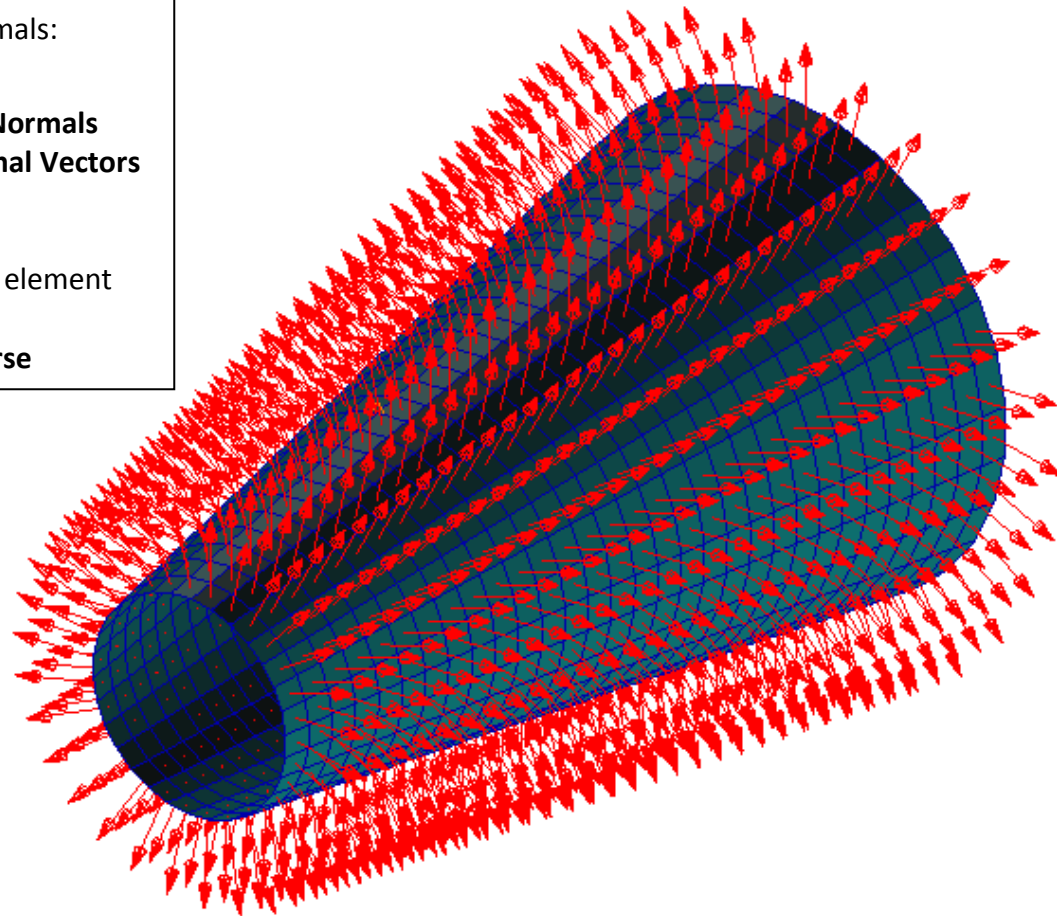


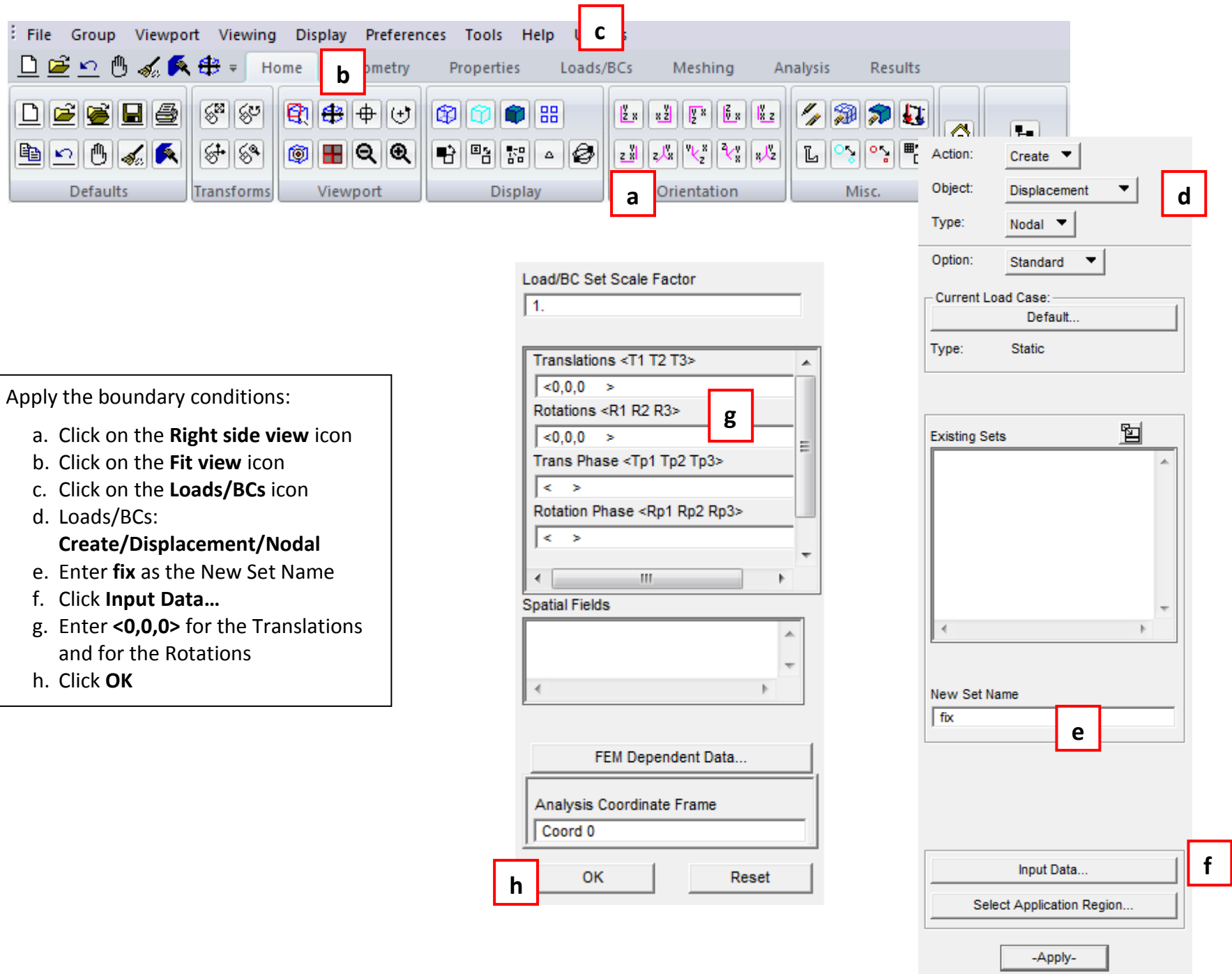


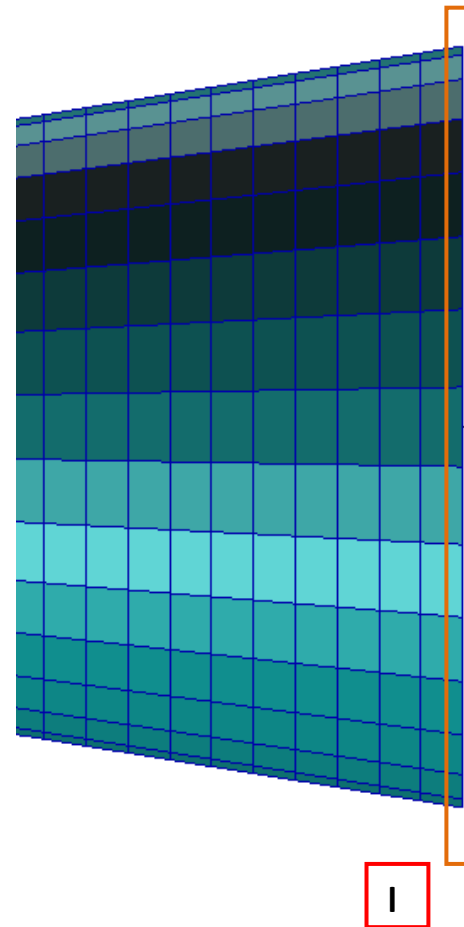
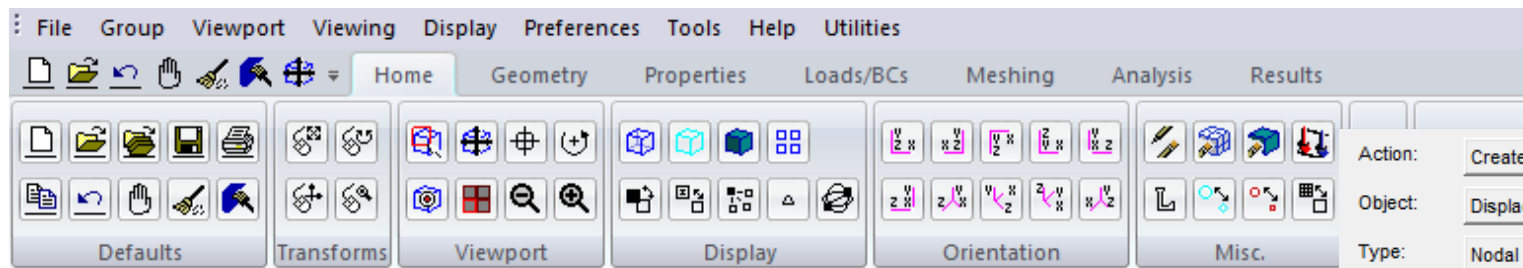
Verify the element normals:

- Meshing:
Verify/Element/Normals
- Select **Draw Normal Vectors**
- Click **Apply**

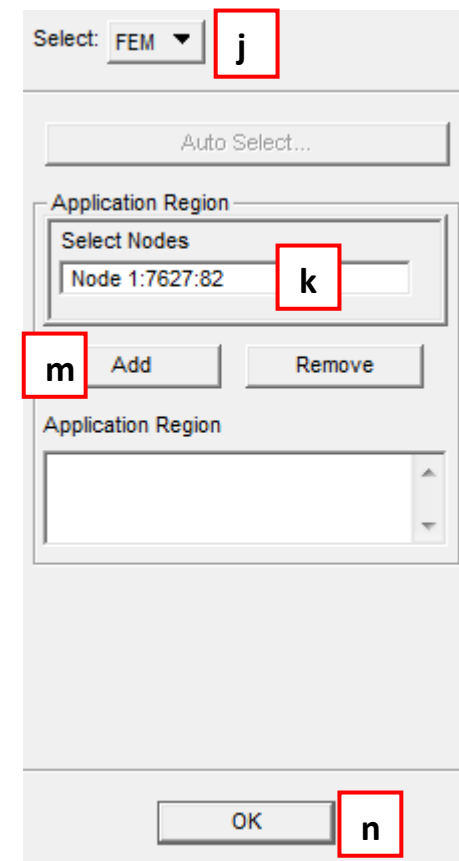
Remark: To reverse the element normals use **Elements: Modify/Element/Reverse**







- i. Click **Select Application Region...**
- j. Select **FEM**
- k. Click on the **Select Nodes** panel
- l. Select nodes by clicking and dragging the mouse
- m. Click **Add**
- n. Click **OK**
- o. Click **Apply**



File Group Viewport Viewing Display Preferences Tools Help Utilities

Home Geometry Properties Loads/BCs Meshing Analysis Results

Defaults Transforms Viewport Display **a** Animation Misc.

Action: Create
Object: Force **b**
Type: Nodal

Current Load Case: Default...
Type: Static

Existing Sets

New Set Name: load **c**

Input Data... **d**
Select Application Region...
-Apply-

Load/BC Set Scale Factor: 1.

Force <F1 F2 F3> **e**
<0,-2000,0>
Moment <M1 M2 M3>
Force Phase <Fp1 Fp2 Fp3>
Moment Phase <Mp1 Mp2 Mp3>

Spatial Fields

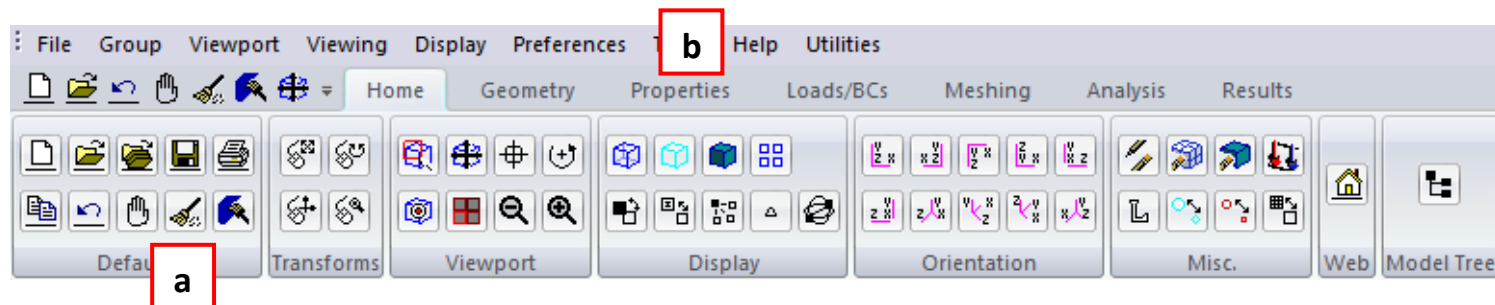
FEM Dependent Data...

Analysis Coordinate Frame: Coord 0

f OK Reset

Apply the force:

- Click on the **Iso 1 View** icon
- Loads/BCs: **Create/Force/Nodal**
- Enter **load** as the New Set Name
- Click **Input Data...**
- Enter **<0,-2000,0>** for the Force
- Click **OK**



a. Click on the **Reset graphics** icon
Define materials:

- b. Click on the **Properties** tab
- c. Click on **Isotropic** icon, then:
Create/Isotropic/Manual Input
- d. Enter **aluminum** as the Material Name
- e. Click **Input Properties...**
- f. Enter **73000** as Elastic Modulus and **0.33** as Poisson Ratio
- g. Click **OK**
- h. Click **Apply**
- i. Repeat steps f÷k to define the second, relatively stiff material e.g.: **ring_mat**, $E = 2e7$, $\nu = 0.29$

Constitutive Model: Linear Elastic

Property Name	Value
Elastic Modulus =	73000 (labeled 'f')
Poisson Ratio =	.33
Shear Modulus =	
Density =	
Thermal Expan. Coeff =	
Structural Damping Coeff =	
Reference Temperature =	

Temperature Dep/Model Variable Fields:

Current Constitutive Models:

OK (labeled 'g') Clear Cancel

Action: Create (labeled 'c')

Object: Isotropic

Method: Manual Input

Existing Materials

Filter *

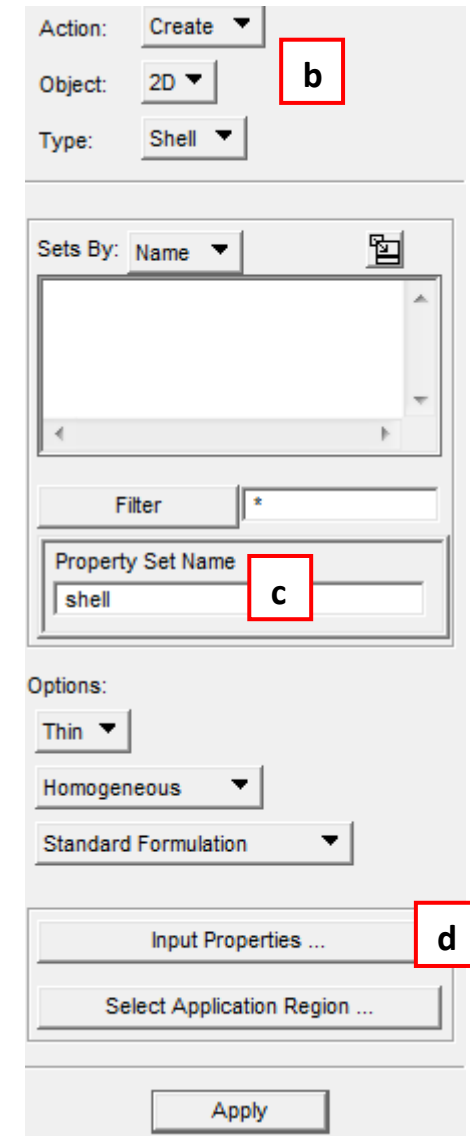
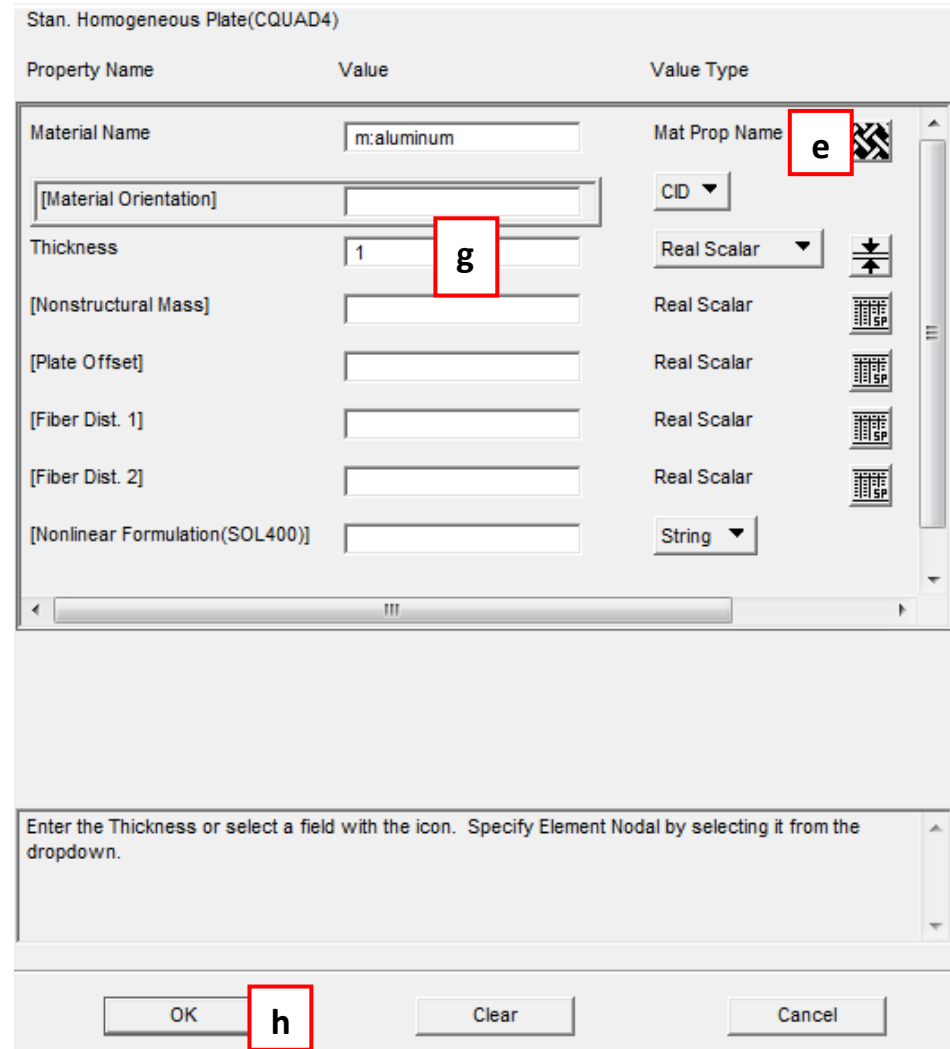
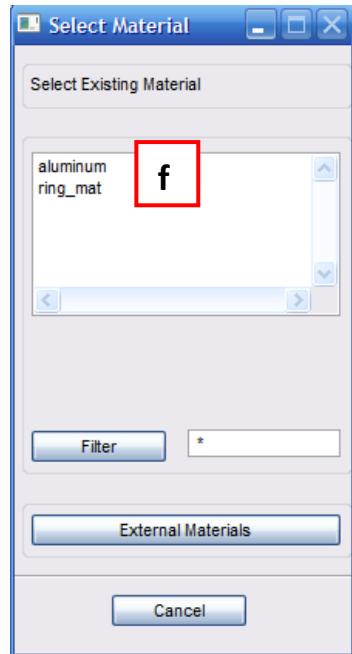
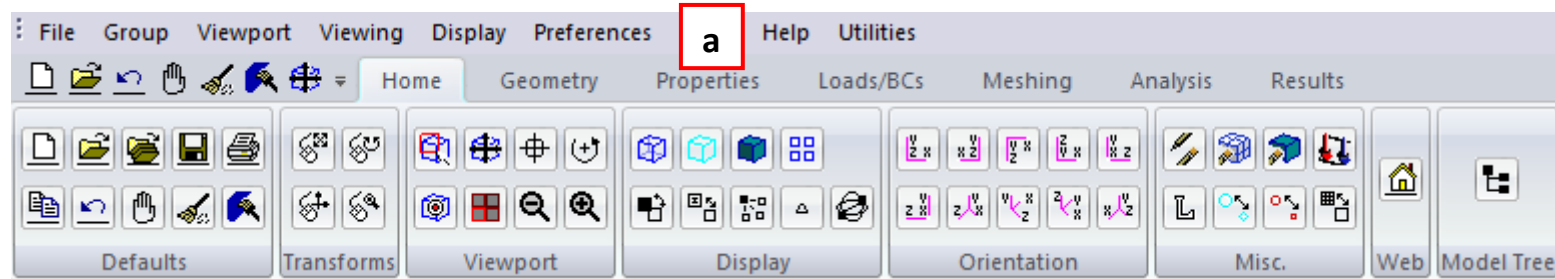
Material Name: aluminum (labeled 'd')

Description

Input Properties ... (labeled 'e')

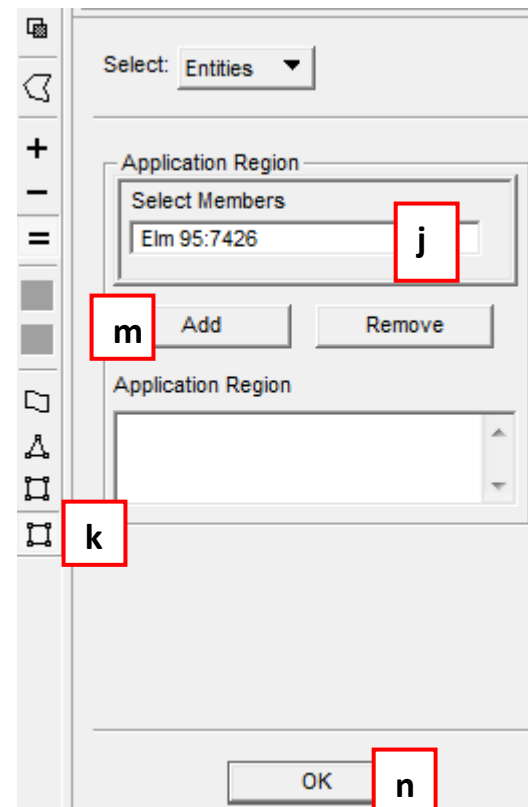
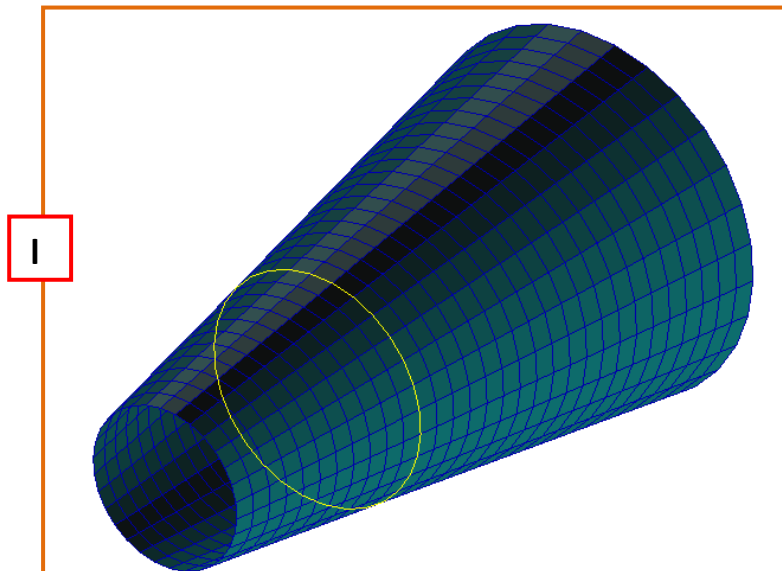
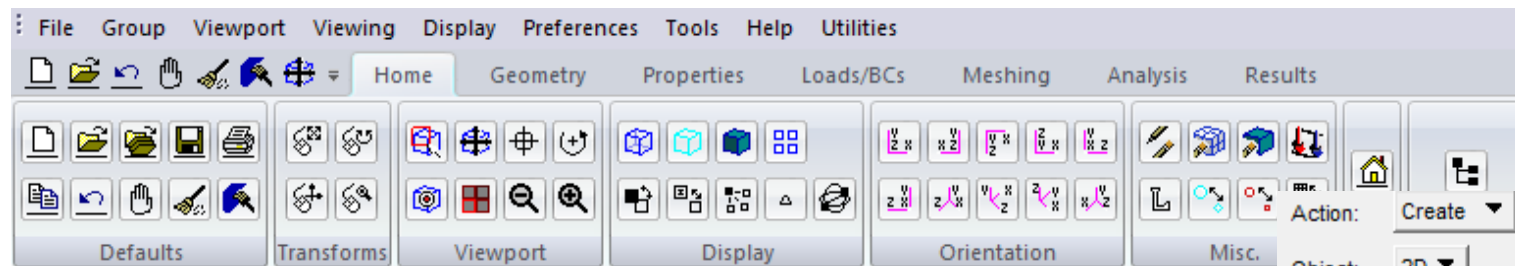
Change Material Status ...

Apply (labeled 'h')

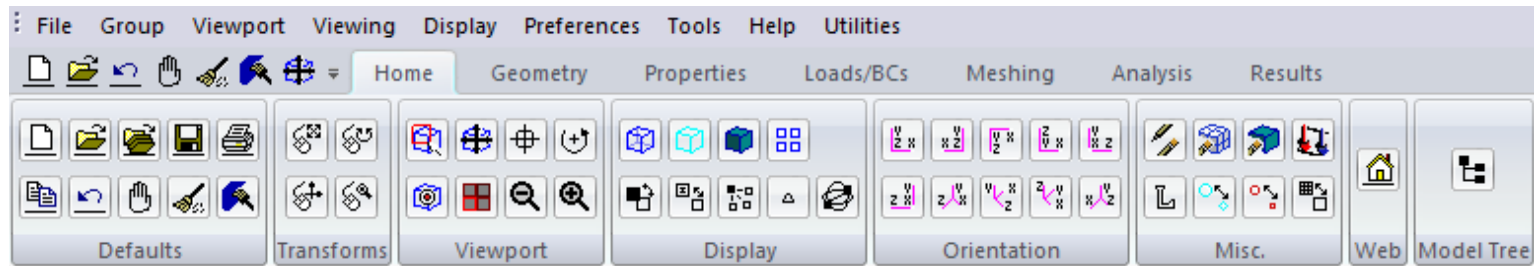


Assign the properties (shell):

- a. Double-click **Properties** tab
- b. Properties: **Create/2D/Shell**
- c. Enter **shell** as the New Set Name
- d. Click **Input Properties...**
- e. Click on the **Mat Prop Name** icon
- f. Select **aluminum**
- g. Enter **1** as the Thickness
- h. Click **OK**



- i. Click **Select Application Region...**
- j. Click on the **Select Members** panel
- k. Select **Shell element** icon
- l. Select all shell elements by clicking and dragging the mouse
- m. Click **Add**
- n. Click **OK**
- o. Click **Apply**



Assign the properties (ring):

- Properties:
Create/1D/Beam
- Enter **ring** as the New Set Name
- Click **Input Properties...**
- Click on the **Create Sections** icon

General Section Beam (CBAR)

Property Name	Value	Value Type
[Section Name]	na:	Properties
Material Name		Mat Prop Name
Bar Orientation		Vector
[Offset @ Node 1]		Vector
[Offset @ Node 2]		Vector
[Pinned DOFs @ Node 1]		String
[Pinned DOFs @ Node 2]		String
Area		Real Scalar

Create Sections

 Beam Library
☐ Assoc. Beam Section

Enter the Section Name, select existing section using the icon, or use the create sections icon below to create a new section.

OK Clear Cancel

Action: Create

Object: 1D

Type: Beam

Sets By: Name

shell

Filter *

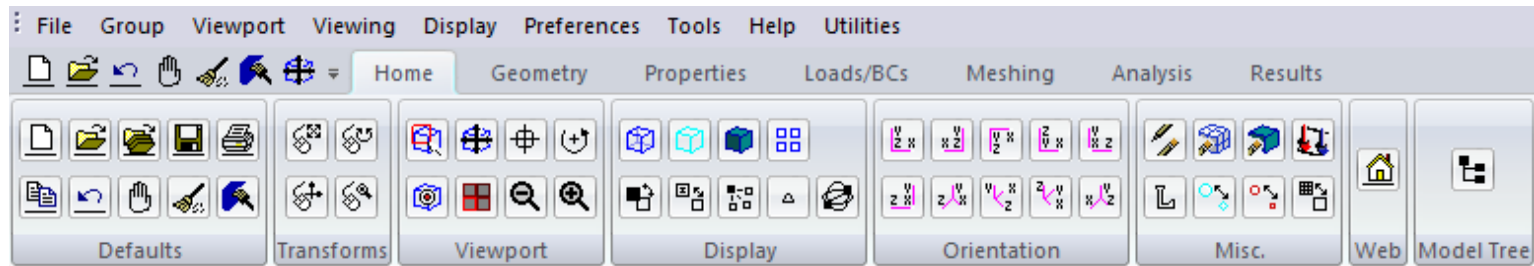
Property Set Name
ring

Options:
 General Section
 Standard Formulation

Input Properties ...

Select Application Region ...

Apply



- e. Create Sections: **Create/Standard Shape/NASTRAN Standard**
- f. Enter **section** as the New Section Name
- g. Click 
- h. Click 
- i. Enter **12** as the value of W and **2** as the value of H
- j. Click **OK**

Action: Create e

Object: Standard Shape

Method: NASTRAN Standard

Existing Sections

Filter

New Section Name f

section

h g

Calculate/Display

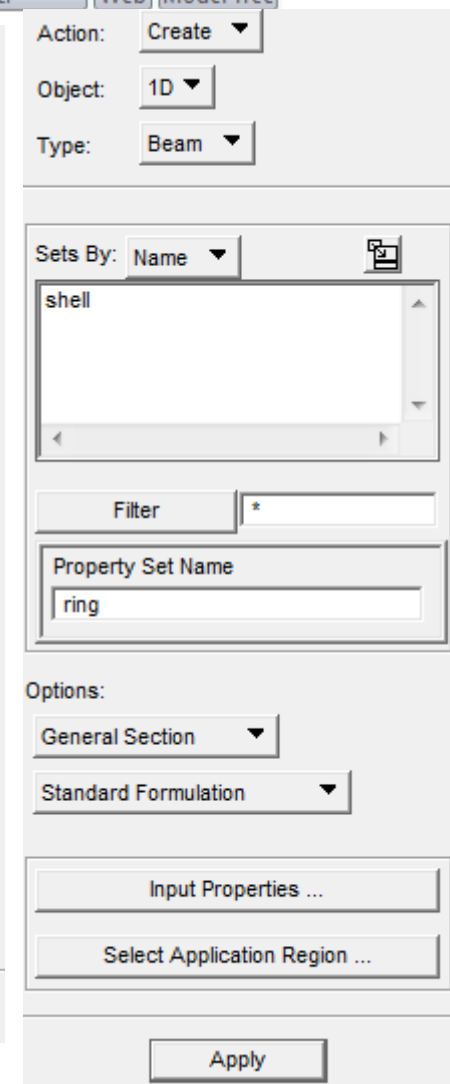
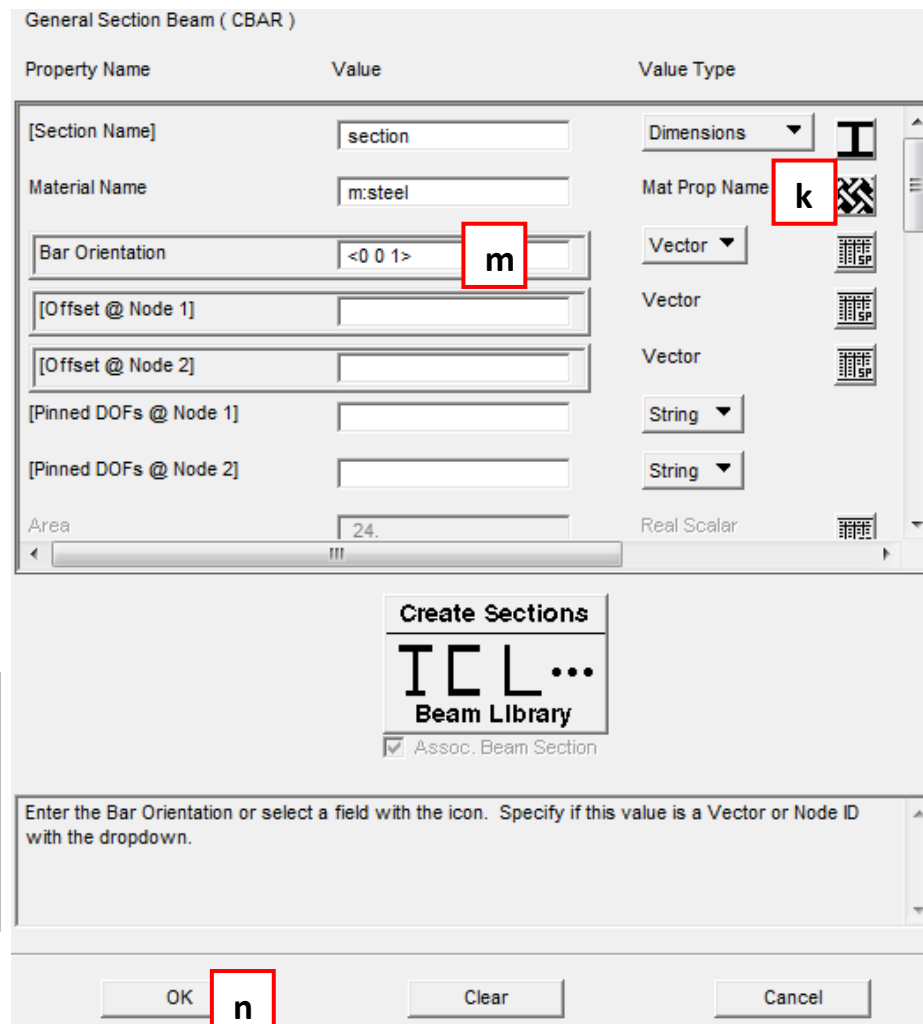
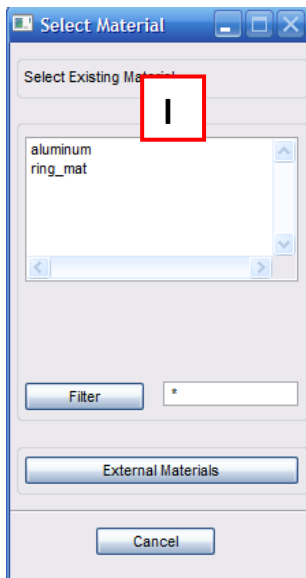
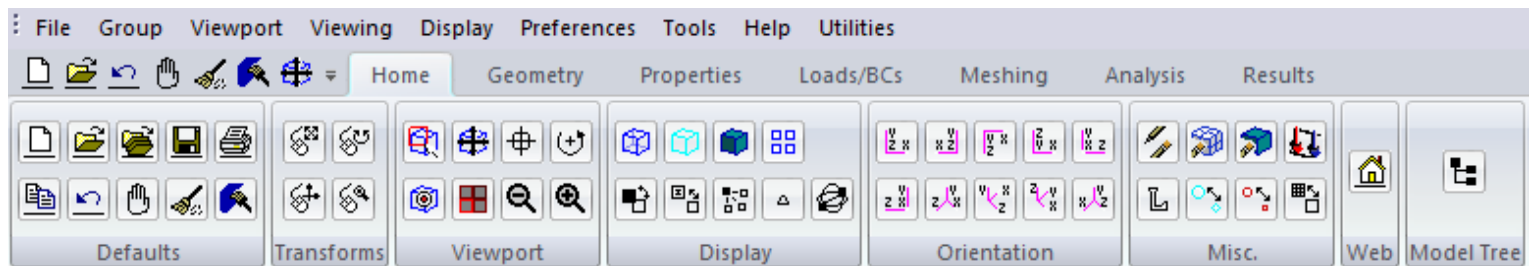
Write to Report File

j OK Apply Reset Cancel

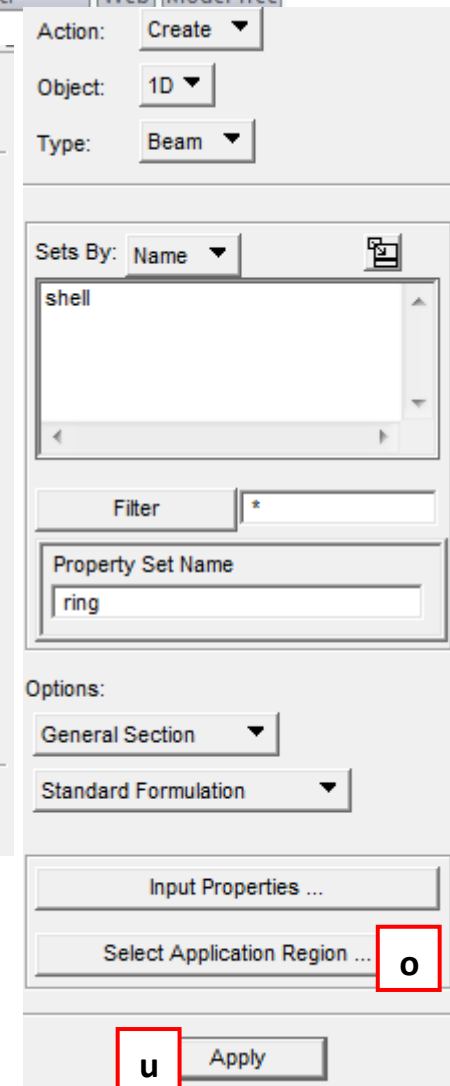
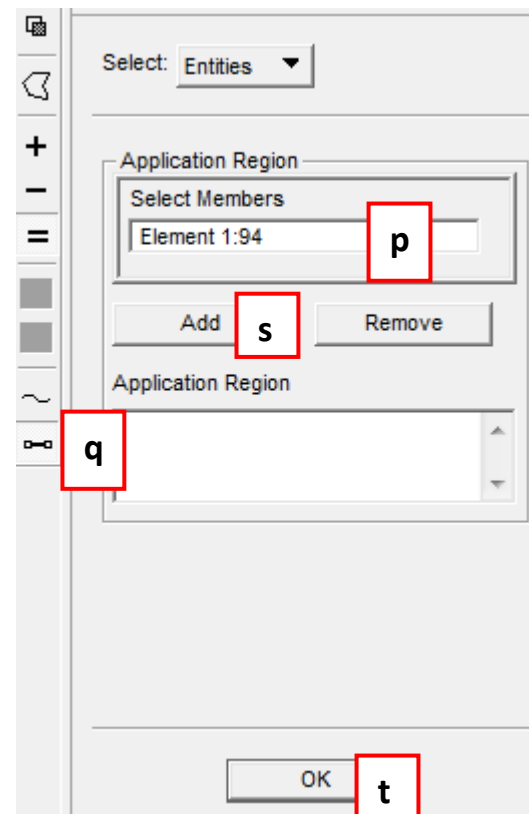
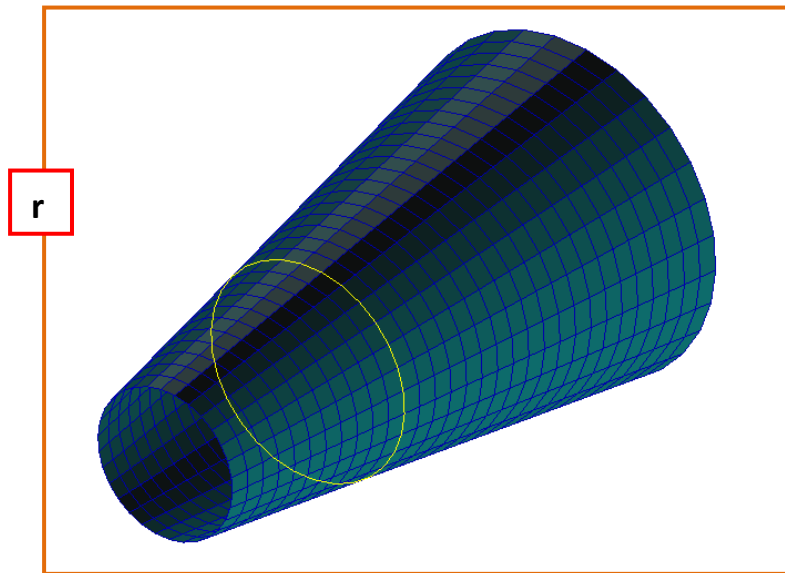
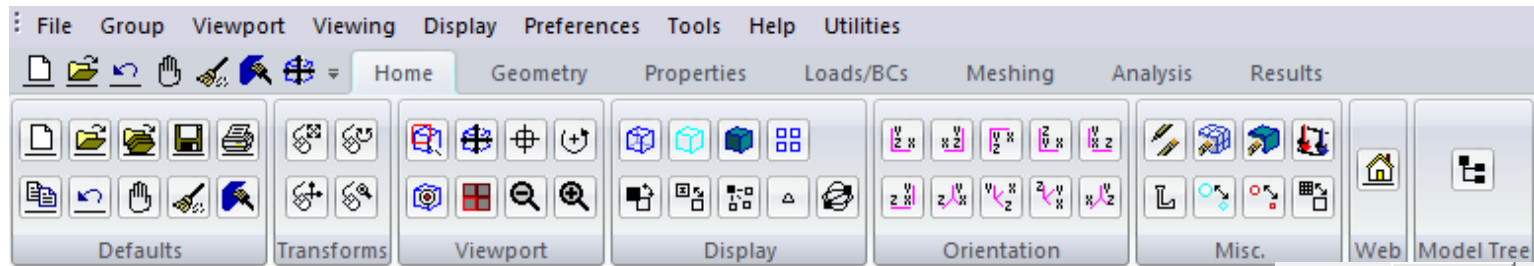
W 12 i

H 2

Spatial Scalar Fields



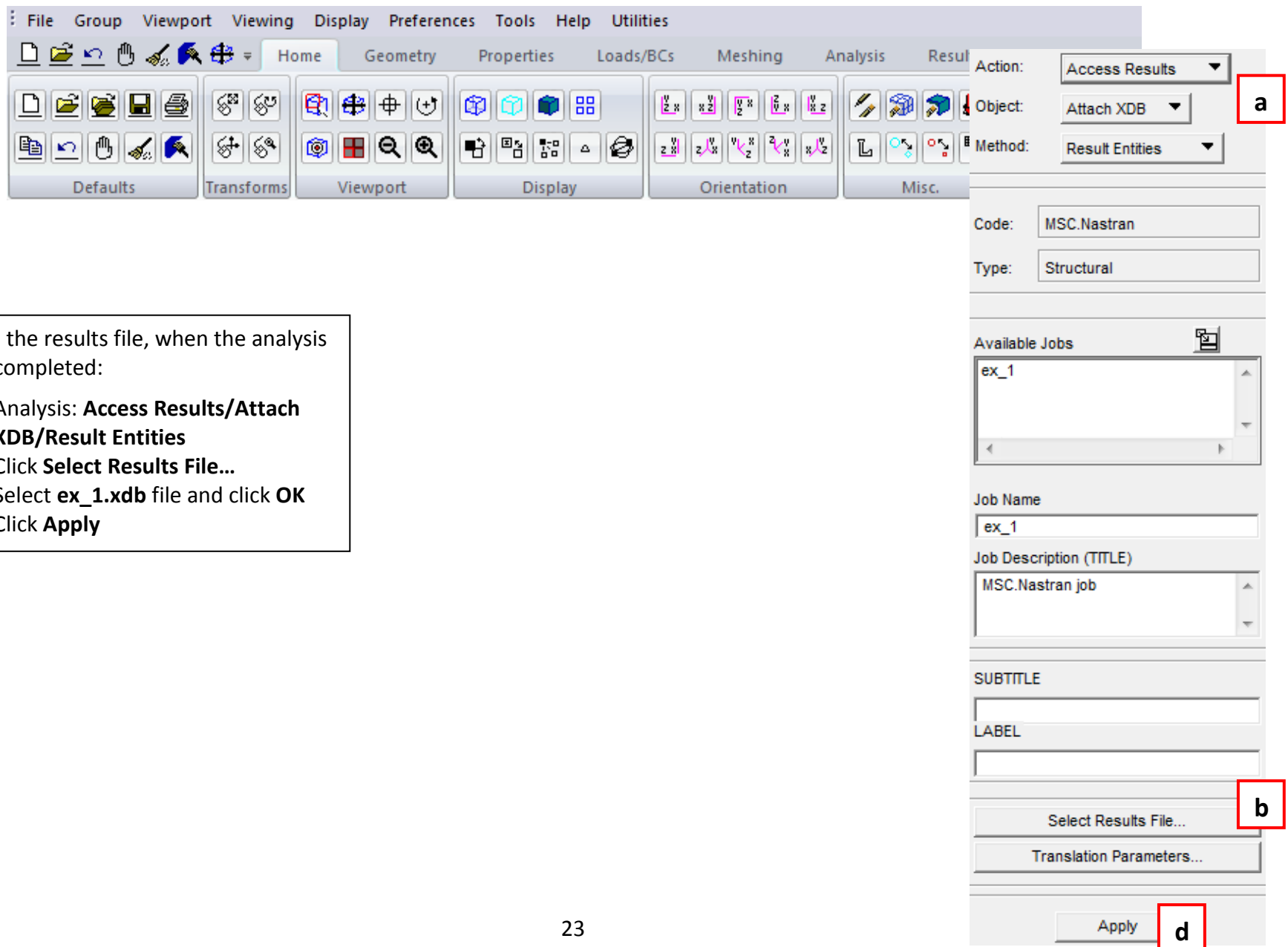
- k. Click on the **Mat Prop Name** icon
- l. Select **ring_mat**
- m. Enter **<0 0 1>** for the Bar Orientation
- n. Click **OK**



- o. Click **Select Application Region...**
- p. Click on the **Select Members** panel
- q. Select **Beam element** icon
- r. Select all beam elements by clicking and dragging the mouse
- s. Click **Add**
- t. Click **OK**
- u. Click **Apply**

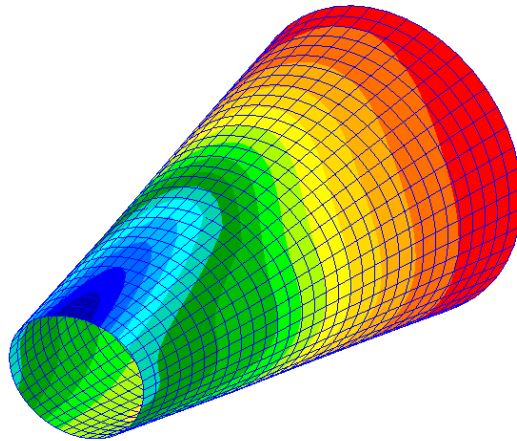
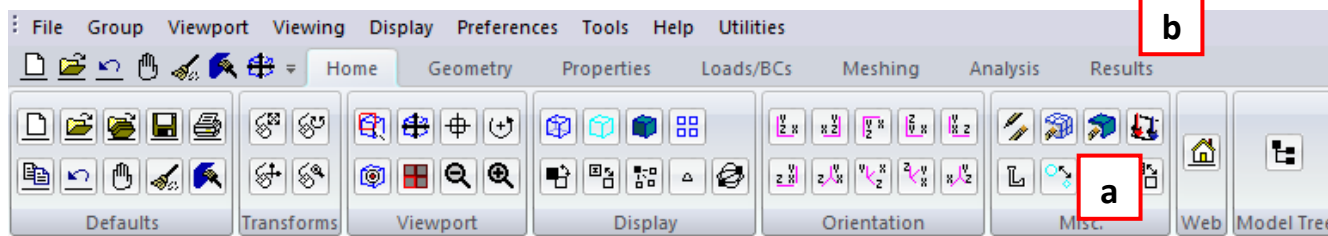
Run a linear analysis:

- Click on the **Analysis** tab
- Analysis: **Analyze/Entire Model/Full Run**
- Click **Solution Type...**
- Select **LINEAR STATIC** as the Solution Type
- Click **Solution Parameters...**
- Click **Results Output Format...**
- Uncheck **Print**
- Click **OK**
- Click **OK**
- Click **OK**
- Click **Apply**



Attach the results file, when the analysis job is completed:

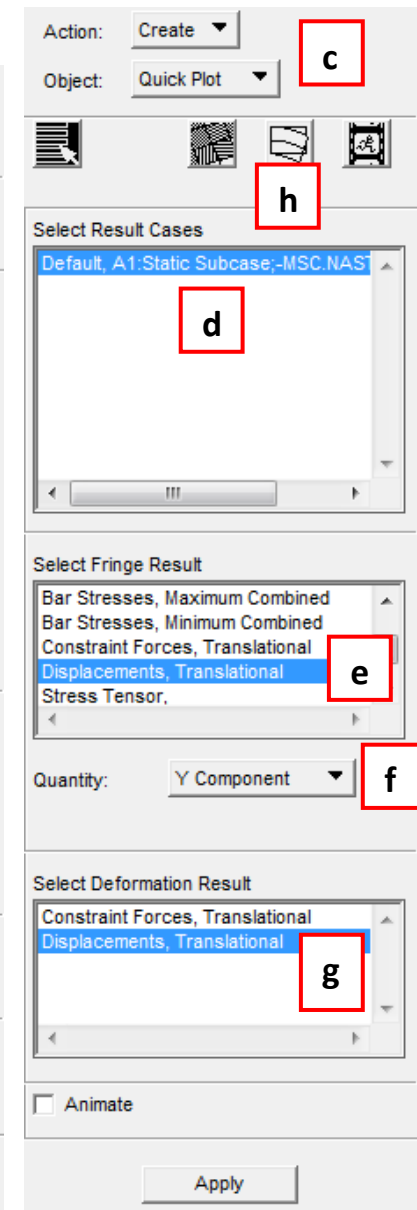
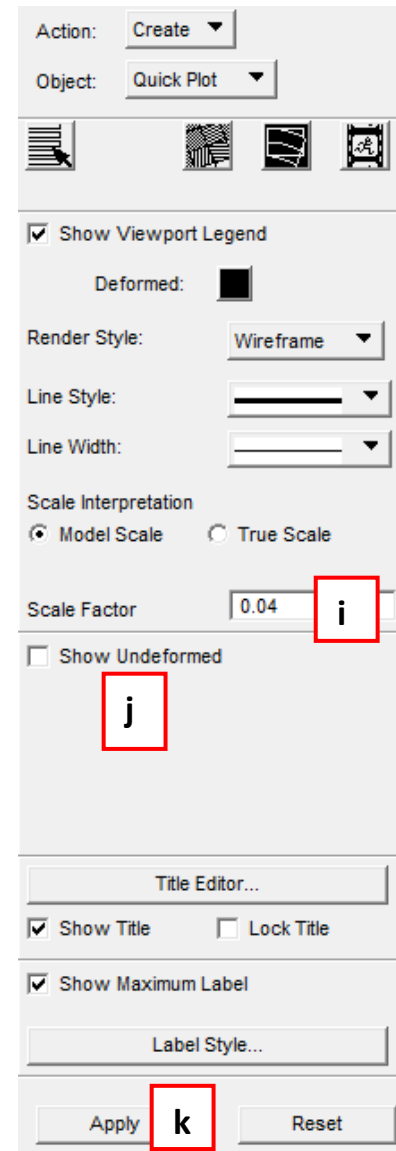
- Analysis: **Access Results/Attach XDB/Result Entities**
- Click **Select Results File...**
- Select **ex_1.xdb** file and click **OK**
- Click **Apply**

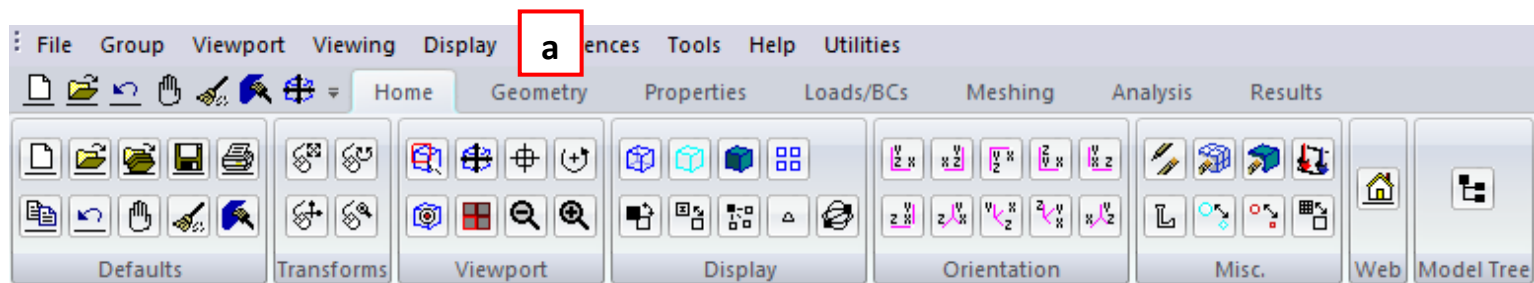


Post-process the results:

- Click on the **Plot/Erase Geometry** icon
- Click on the **Results** tab
- Results: **Create/Quick Plot**
- Select Result Cases: **Default, A1:Static Subcase**
- Select Fringe Result: **Displacements, Translational**
- Quantity: **Y Component**
- Select Deformation Result: **Displacements, Translational**
- Click on the **Deform Attributes** icon
- Enter **0.04** as the Scale Factor
- Uncheck **Show Undeformed**
- Click **Apply**

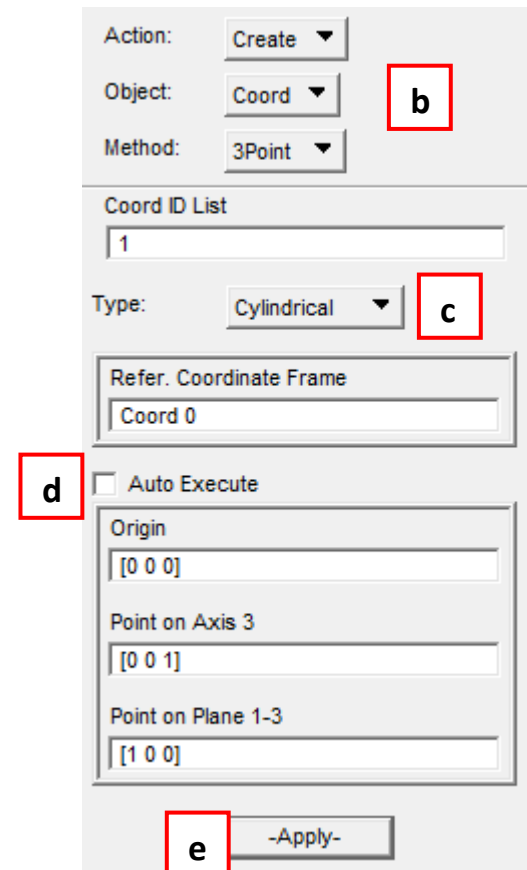
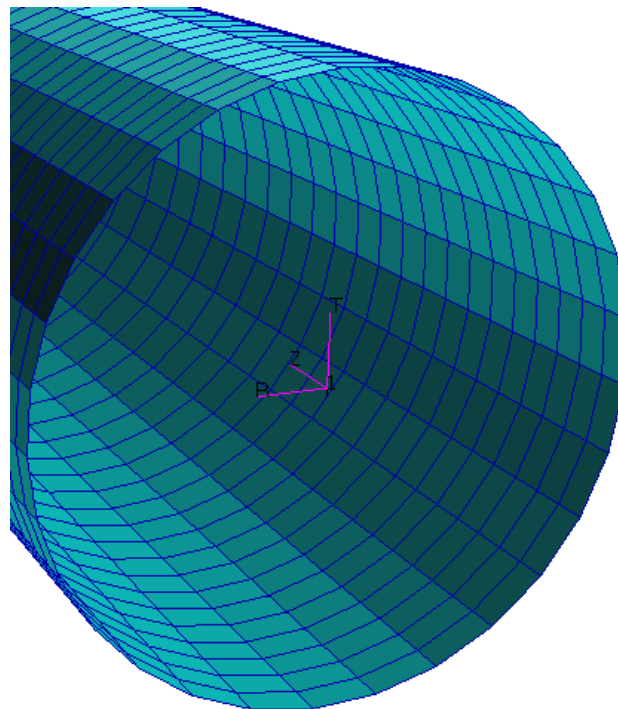
Remark: To capture the plot use **File / Images...**

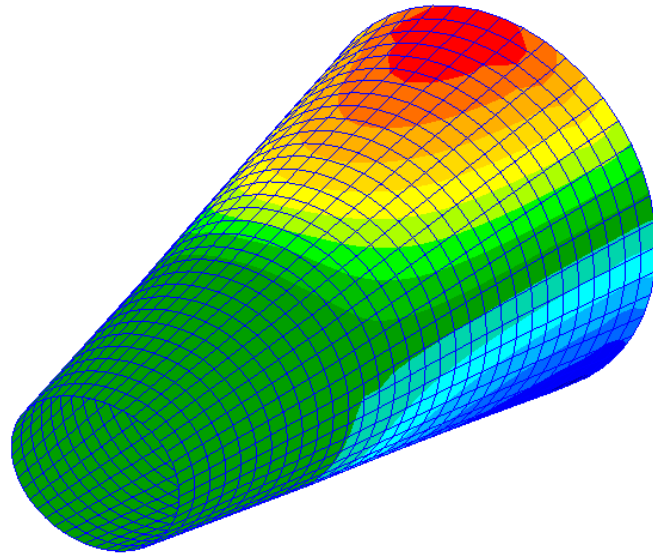
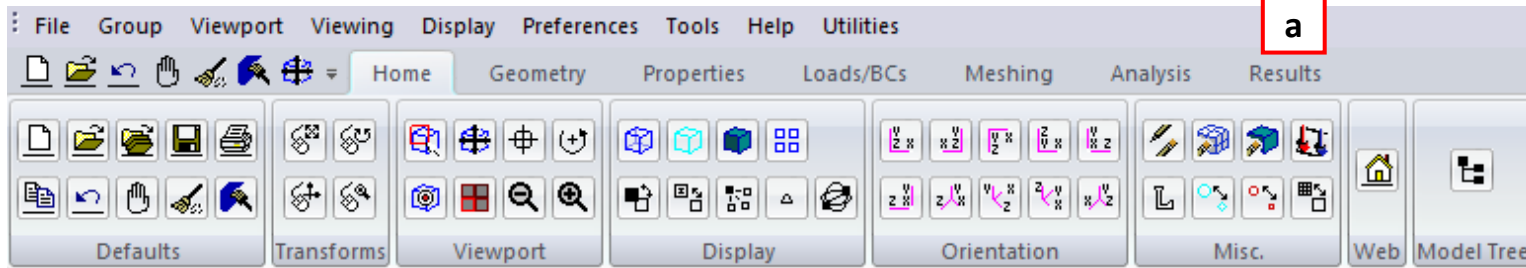




Create a cylindrical coordinate system:

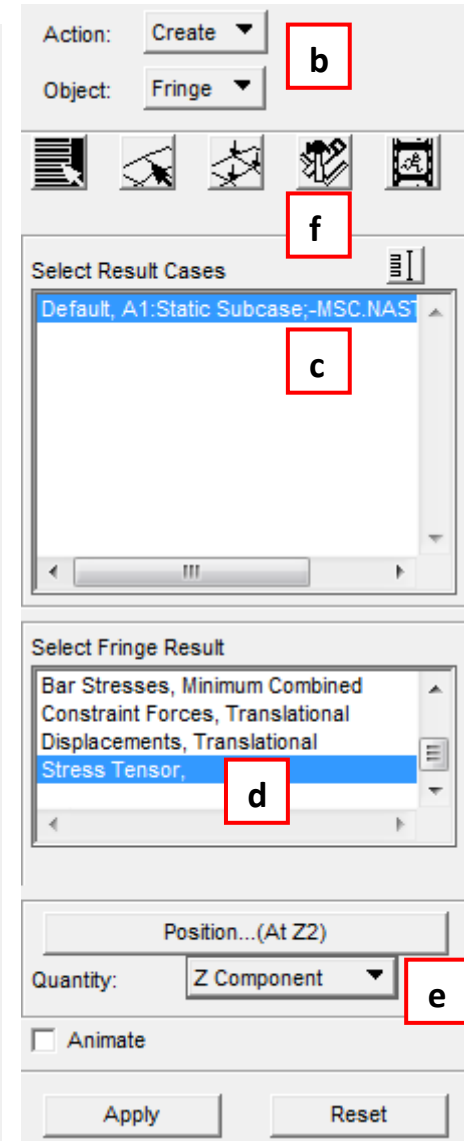
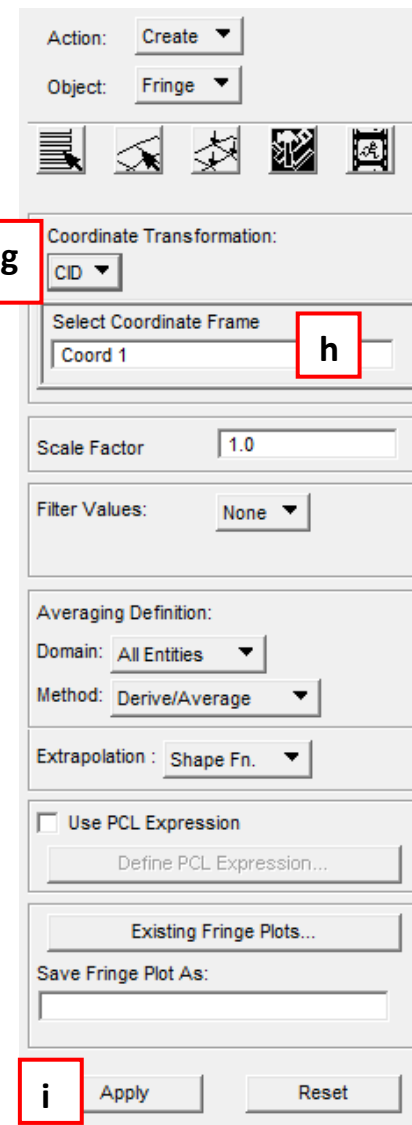
- Click on the **Geometry** tab
- Geometry: **Create/Coord/3Point**
- Type: **Cylindrical**
- Uncheck **Auto Execute**
- Click **Apply**

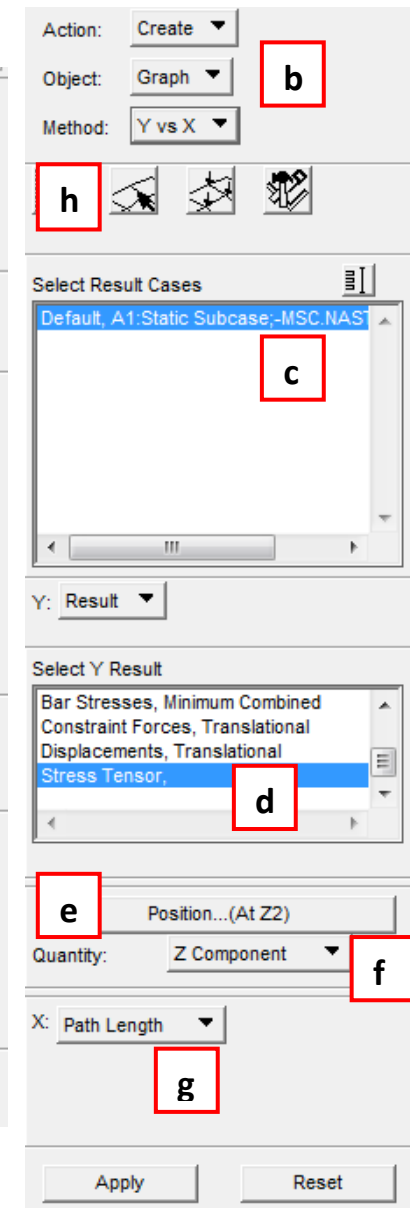
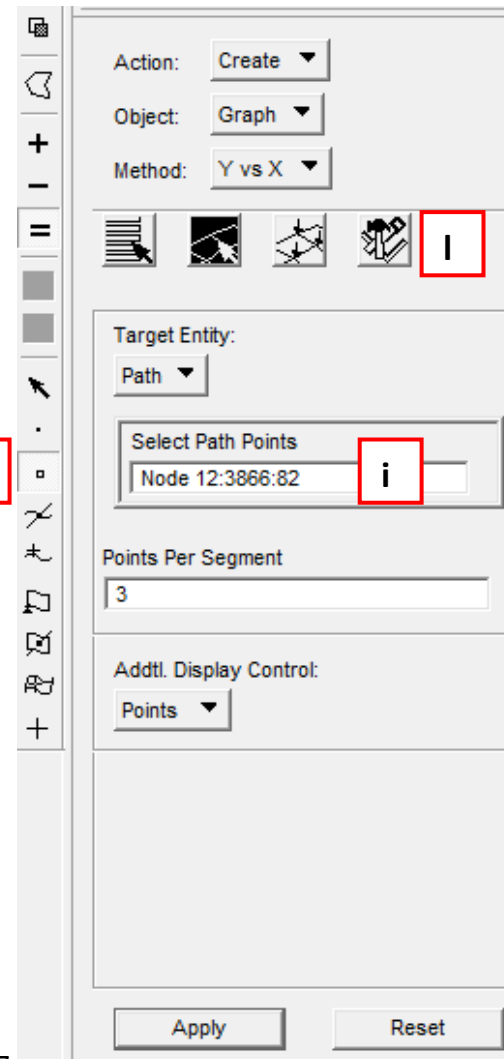
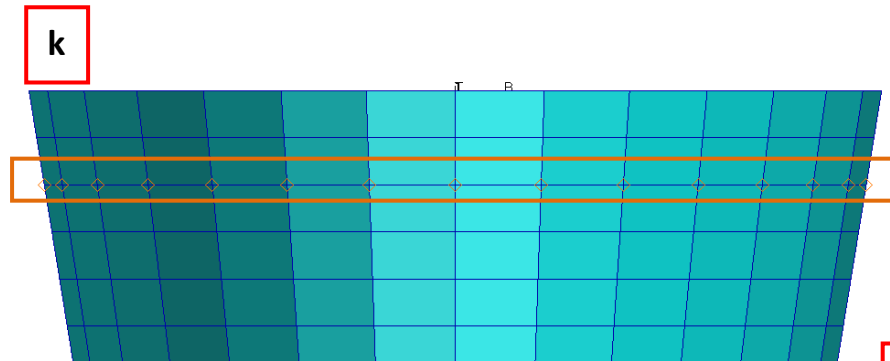
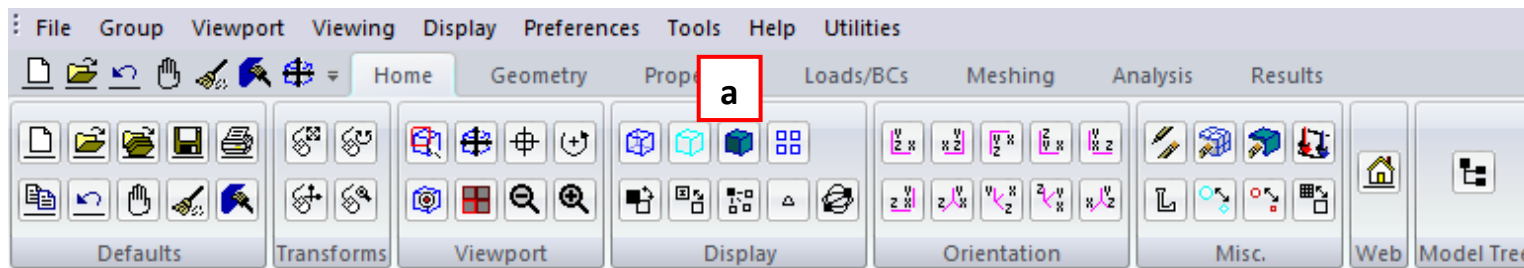




Post-process the results:

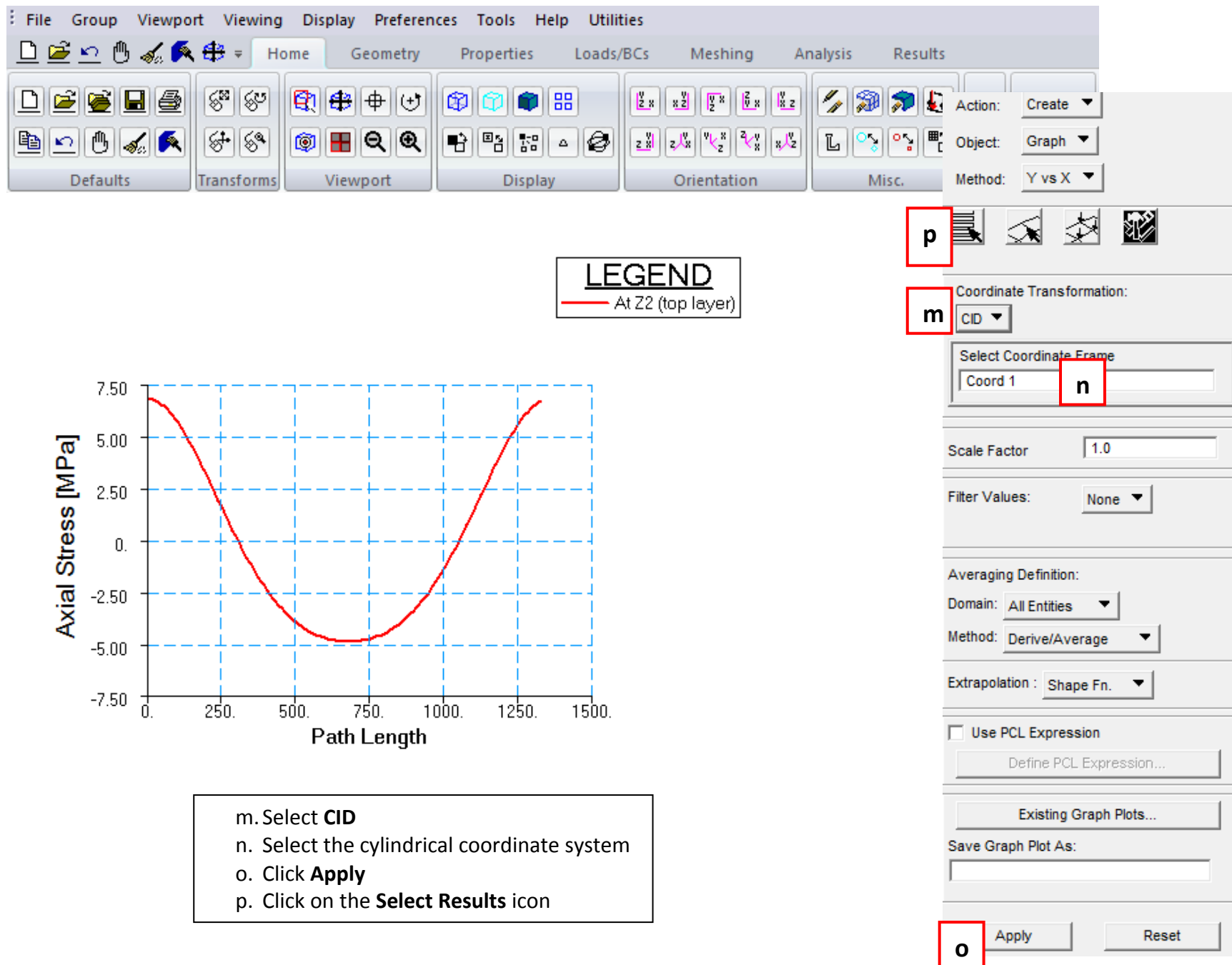
- a. Click on the **Results** tab
- b. Results: **Create/Fringe**
- c. Select Result Cases: **Default, A1:Static Subcase**
- d. Select Fringe Result: **Stress Tensor**
- e. Quantity: **Z Component** (axial stress)
- f. Click on the **Plot Options** icon
- g. Select **CID**
- h. Select the cylindrical coordinate system
- i. Click **Apply**

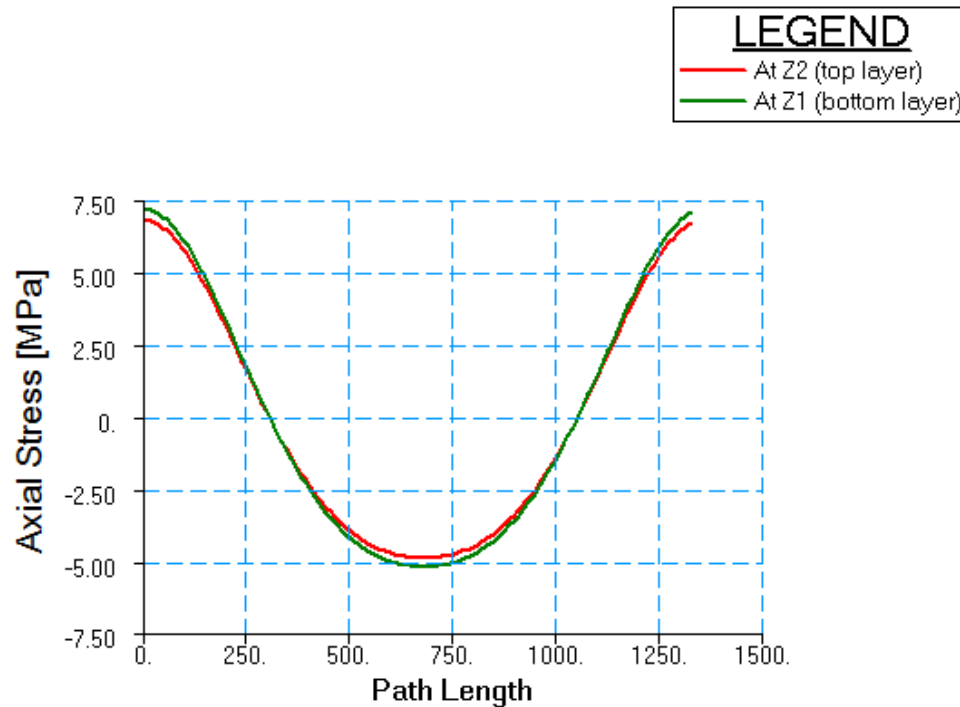
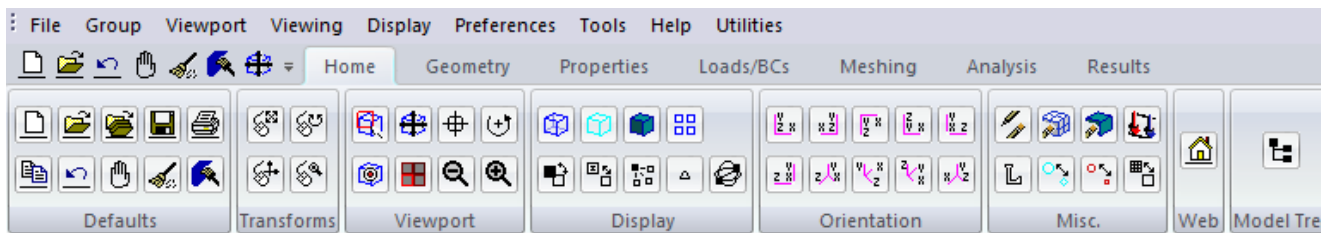




Create a graph:

- Click on **smooth shaded** display icon
- Results: **Create/Graph/Y vs X**
- Select Result Cases: **Default, A1:Static Subcase**
- Select Fringe Result: **Stress Tensor**
- Position: **At Z2** (top layer)
- Quantity: **Z Component** (axial stress)
- X: **Path Length**
- Click on the **Target Entities** icon
- Click on the **Select Path Points** panel
- Select **Node** icon
- Select nodes by clicking and dragging the mouse
- Click on the **Plot Options** icon





- q. Change *Position* to **At Z1** (bottom layer)
- r. Click on the **Display Attributes** icon
- s. Check **Append Curves in XY Window**
- t. Click **Apply**

Why do results for both layers differ?

Compare foregoing graphs with the constraint forces distribution from the next slide

Action: Create
Object: Graph
Method: Y vs X

Curve Fit: Linear
Curve Style:
☐ Show Symbol

☒ Show X Axis Label
X Axis Label: Path Length
X Axis Scale: ☒ Linear ☐ Log
X Axis Format...

☒ Show Y Axis Label
Y Axis Label: Axial stress [MPa]
Y Axis Scale: ☒ Linear ☐ Log
Y Axis Format...

XY Window Name: 3
☒ Append Curves in XY Window

t Apply Reset

Action: Create
Object: Graph
Method: Y vs **r**

Select Result Cases
Default, A1:Static Subcase;-MSC.NAS

Y: Result

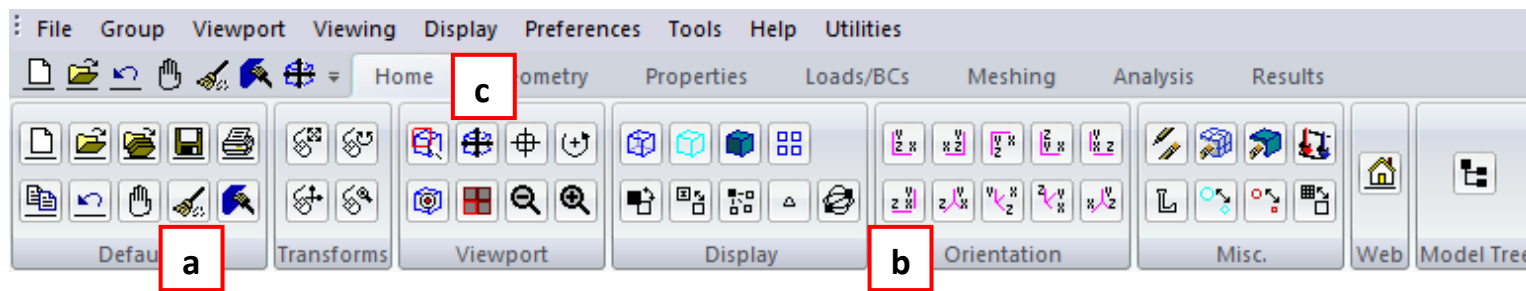
Select Y Result
Bar Stresses, Maximum Combined
Bar Stresses, Minimum Combined
Constraint Forces, Translational
Displacements, Translational
Stress Tensor

Position...(At Z1) **q**

Quantity: Z Component

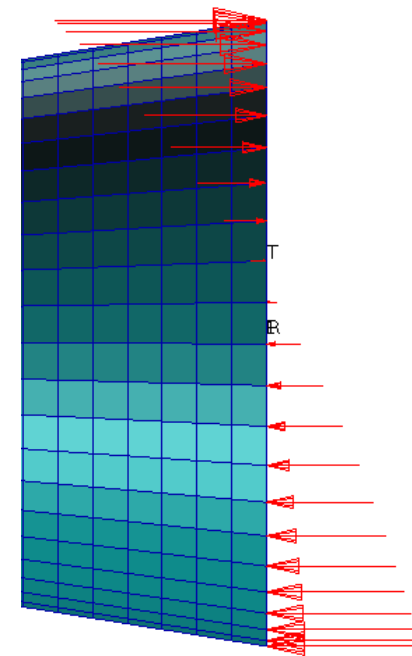
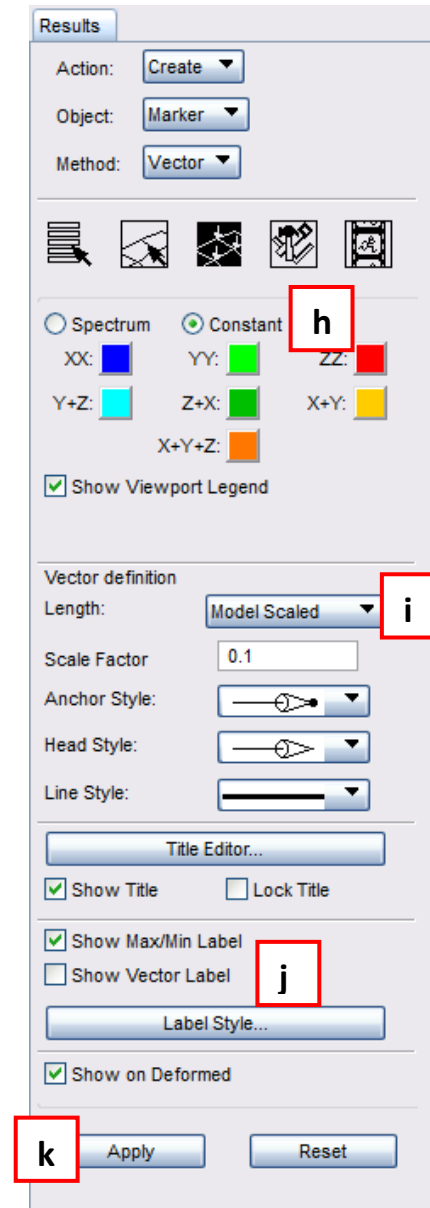
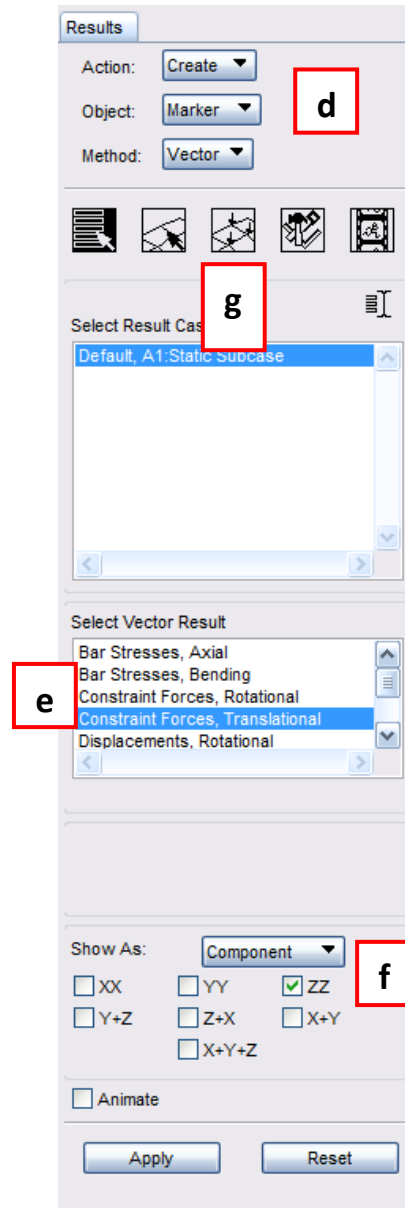
X: Path Length

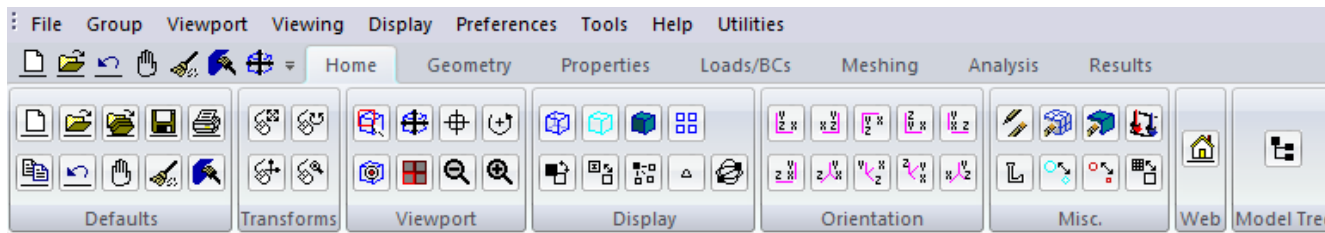
Apply Reset



Compare obtained graph with vector results:

- Click on the **Reset graphics** icon
- Select **Right side view**
- Fit** window
- Click on **Results/Create/Marker/Vector**
- Select **Constraint Forces, Translational** for the Vector Result
- Change option to **Component** and toggle on only **ZZ** option
- Click on **Display Attributes**
- Toggle **Constant** vector color option and choose color for **ZZ** component
- Change **Length** to **Model Scaled**
- Toggle off **Show Vector Label**
- Click **Apply**



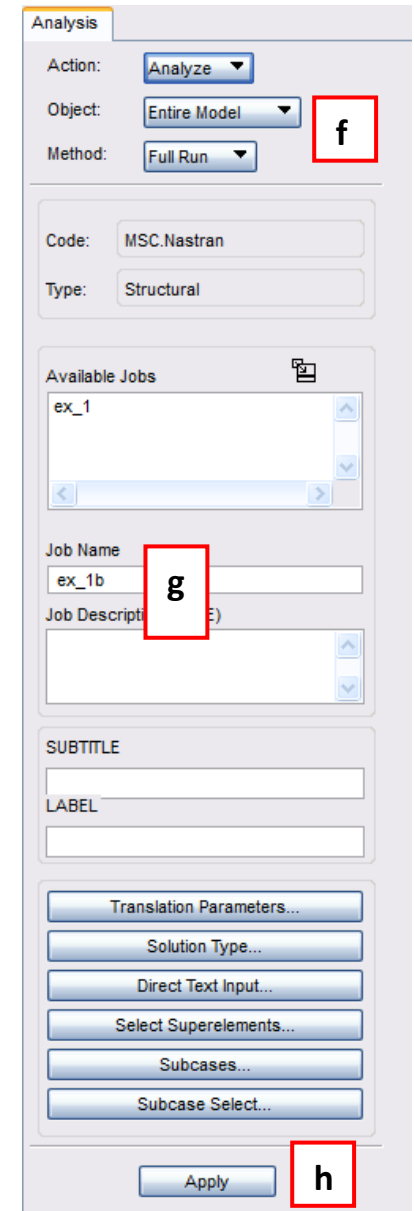
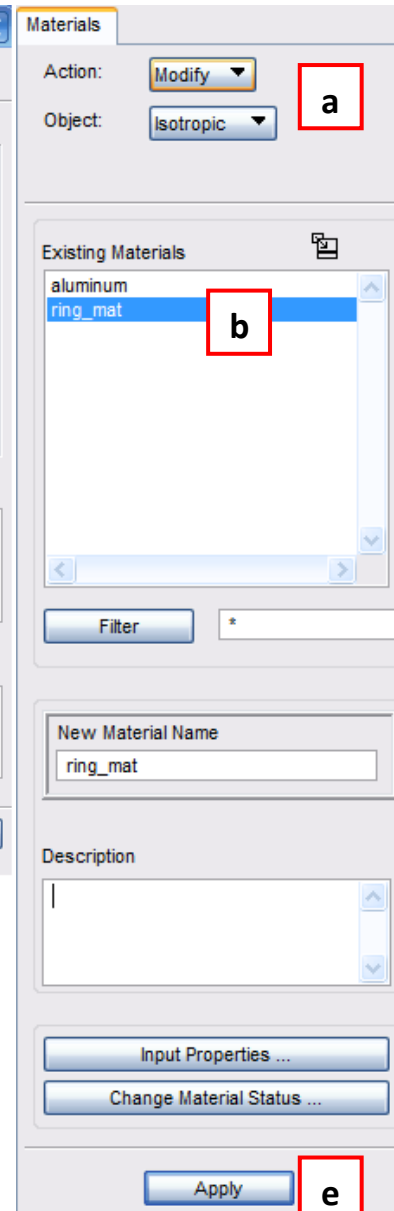
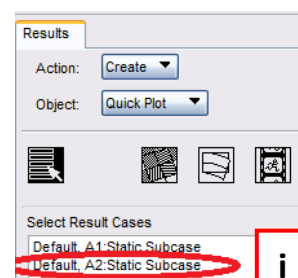
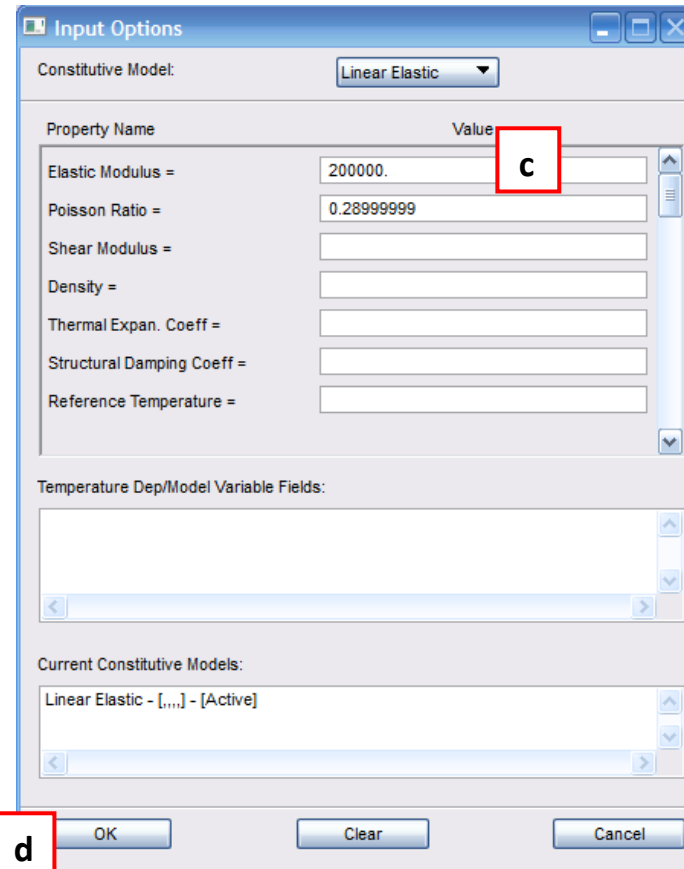


ANALYSIS NO. 2

- change of the material properties

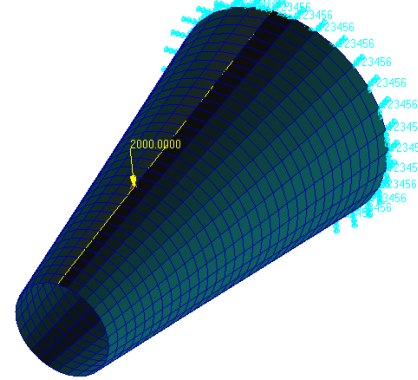
Change ring material properties, run second analysis and compare results with recently obtained:

- Go to **Properties** tab and select **Modify/Isotropic**
- Choose **ring_mat**
- Change Young's Modulus to **2e5** (note that previous material was 100 times stiffer!)
- Click **OK**
- Click **Apply**
- In the **Analysis** tab, select **Analyze/Entire Model/Full Run**
- Change **Job Name** to **ex_1b**
- Select **Apply**
- After analysis is done, **attach ex_1b.xdb** file and **plot results** in the same manner as previously. Results referring to the second analysis will be tagged as **A2: Static subcase**



LIST OF REQUIRED PLOTS (total number of plots = **10**):

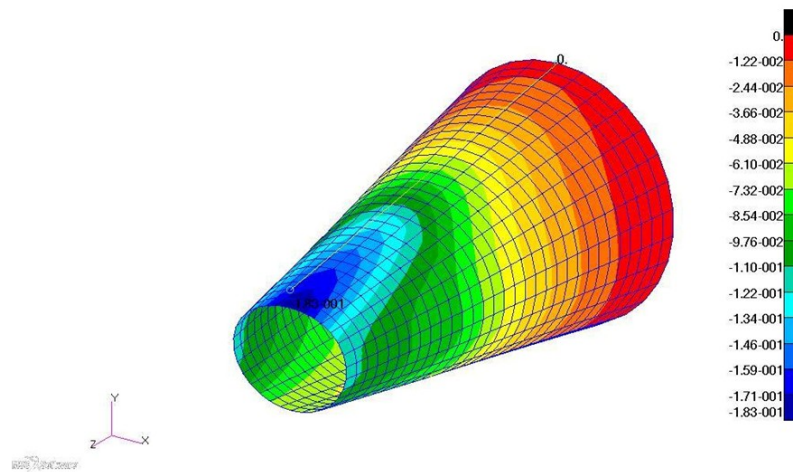
1) FE MODEL WITH LOADS AND BOUNDARY CONDITIONS (1 plot)



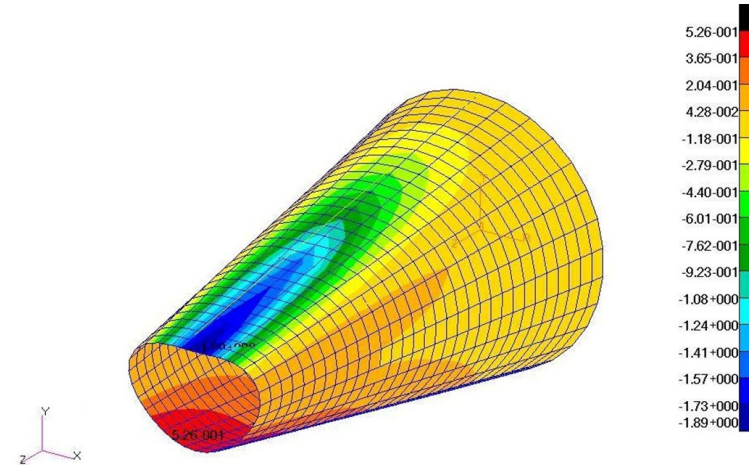
Use this reference to
verify the results of
both analyses.

2) VERTICAL TRANSLATIONAL DISPLACEMENT IN Y DIRECTION (2 plots)

- for 1st and 2nd analysis separately

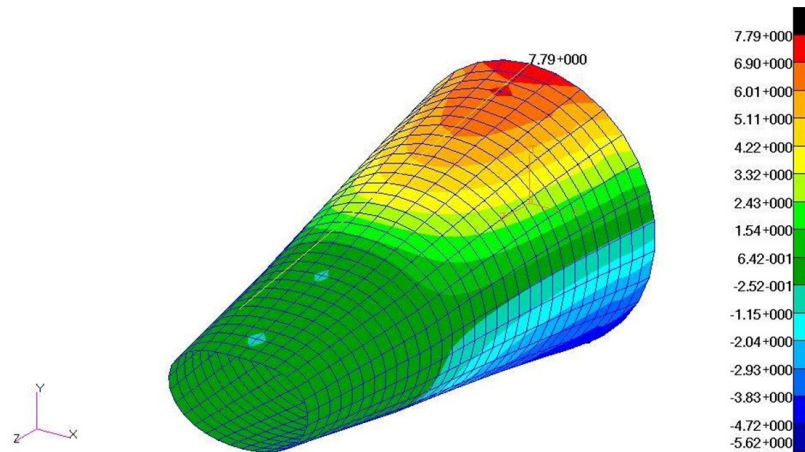


Analysis no. 1

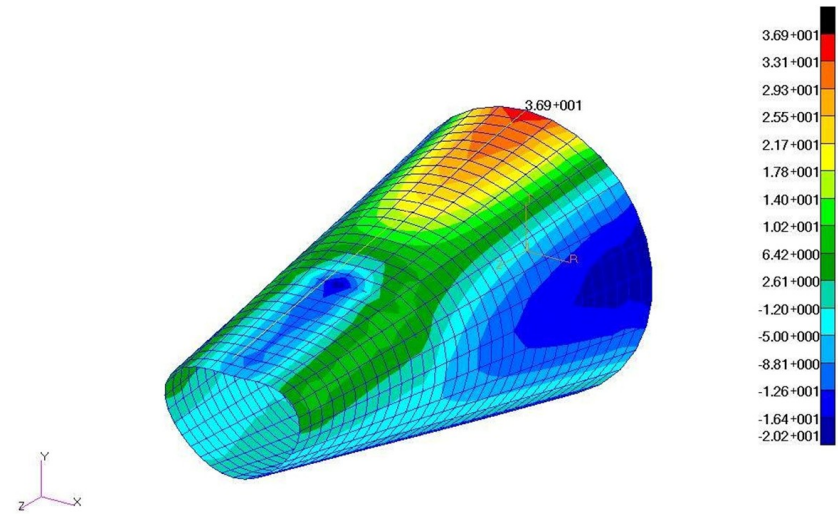


Analysis no. 2

3) AXIAL STRESS "sigma_z" in cylindrical CS (2 plots)
 - for 1st and 2nd analysis separately

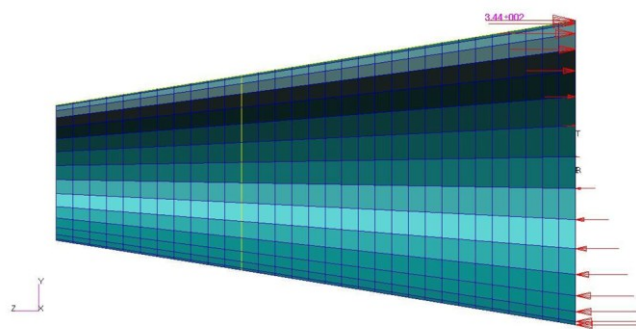


Analysis no. 1

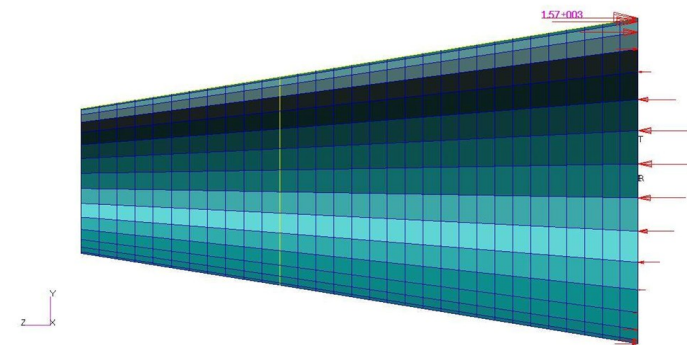


Analysis no. 2

4) CONSTRAINT FORCES, TRANSLATIONAL, COMPONENT ZZ PLOT (2 plots)
 - for 1st and 2nd analysis separately



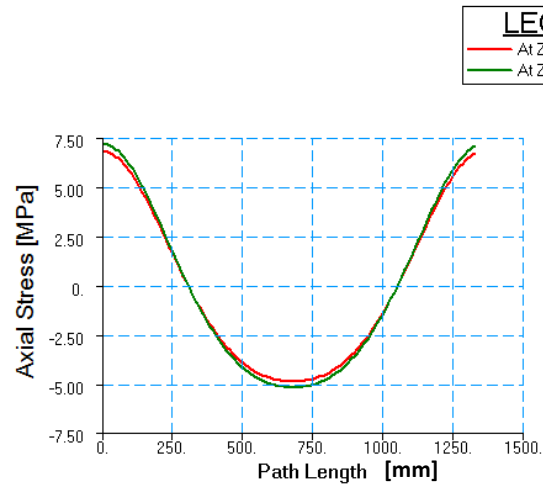
Analysis no. 1



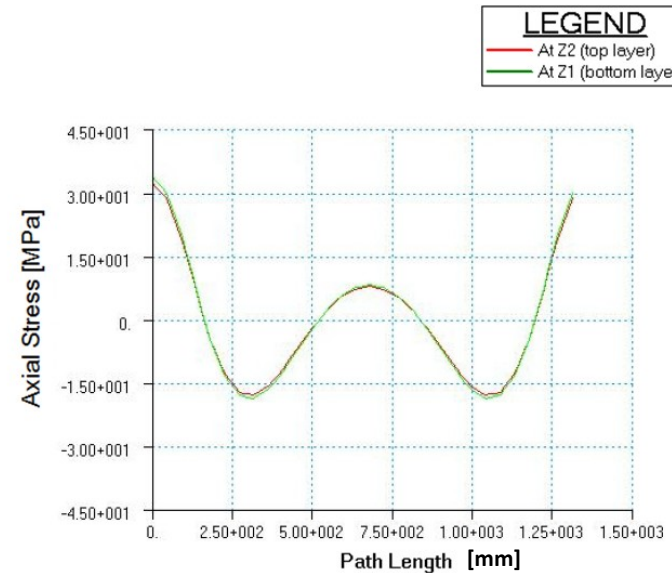
Analysis no. 2

5) YX GRAPH OF AXIAL STRESS " σ_z " on top and bottom layer in cylindrical CS (2 plots)

- for 1st and 2nd analysis separately



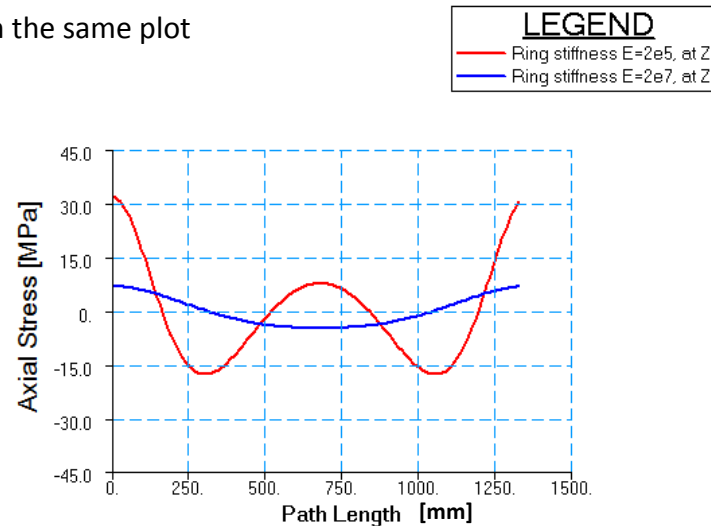
Analysis no. 1



Analysis no. 2

6) YX GRAPH OF AXIAL STRESS " σ_z " on top **OR** bottom layer in cylindrical CS (1 plot)

- curve from 1st and curve from 2nd analysis on the same plot



What qualitative changes can be noticed?

Which calculation yield results more similar to the theoretical solution? Why?